Correspondence History

History for Manuscript Number: RINENG-D-22-00804 oktarina heriyani (INDONESIA): "Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel"

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| Correspondence Date $\triangle \ \mathbb{V}$ | Letter △ ♥ | Recipient | Revision |
|--|--|--------------------------------|----------|
| Oct 14, 2022 | Editor Decision - Accept | oktarina heriyani, S.Si., M.T | 1 |
| Sep 23, 2022 | Author Submits Revision Confirmation | oktarina heriyani, S.Si., M.T | 1 |
| Sep 23, 2022 | PDF Built and Requires Approval | oktarina heriyani, S.Si., M.T | 1 |
| Sep 21, 2022 | Author Revision Reminder - Before Due Date | oktarina heriyani, S.Si., M.T | 1 |
| Sep 18, 2022 | Author Revision Reminder - Before Due Date | oktarina heriyani, S.Si., M.T | 1 |
| Sep 14, 2022 | Author Revision Reminder - Before Due Date | oktarina heriyani, S.Si., M.T | 1 |
| Sep 11, 2022 | Author Revision Reminder - Before Due Date | oktarina heriyani, S.Si., M.T | 1 |
| Sep 10, 2022 | Author Requests Deadline Extension on Revision | noreply_EMsupport@elsevier.com | 1 |
| Sep 07, 2022 | Author Revision Reminder - Before Due Date | oktarina heriyani, S.Si., M.T | 1 |
| Aug 24, 2022 | Editor Decision - Revise | oktarina heriyani, S.Si., M.T | 0 |
| Jul 21, 2022 | Author Notice of Manuscript Number | oktarina heriyani, S.Si., M.T | 0 |
| Jul 12, 2022 | Author Submits New Manuscript Confirmation | oktarina heriyani, S.Si., M.T | 0 |
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To: "oktarina heriyani" oktarina@uhamka.ac.id

"Mohammad Djaeni" moh.djaeni@live.undip.ac.id, "Aldila Kurnia Putri"

aldilakurniaputri@gmail.com, "Syaiful Syaiful" syaiful.undip2011@gmail.com

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Dear Miss heriyani,

Your submission entitled "Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel" has been been assigned the following manuscript number: RINENG-D-22-00804.

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Thank you for submitting your work to this journal.

Kind regards,

Antonio García, Ph.D Editor in Chief Results in Engineering

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Date: Aug 24, 2022

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Subject: Your Submission

Ref.: Ms. No. RINENG-D-22-00804

Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel Results in Engineering

Dear Miss heriyani,

Reviewers have now commented on your paper. You will see that they are advising that you revise your manuscript. If you are prepared to undertake the work required, I would be pleased to reconsider my decision.

For your guidance, reviewers' comments are appended below.

If you decide to revise the work, please submit a list of changes or a rebuttal against each point which is being raised when you submit the revised manuscript.

Please resubmit your revised manuscript by Sep 14, 2022.

To submit a revision, go to https://www.editorialmanager.com/rineng/ and log in as an Author. You will see a menu item call Submission Needing Revision. You will find your submission record there.

Yours sincerely

Ezio Mancaruso, Ph.D Associate Editor Results in Engineering

Comments from the Editors and Reviewers:

Your article would appear to be of interest to a wide engineering research community and in order to promote its visibility even more, may we recommend that you view the past published articles in Results in Engineering and if you find any relevant publications, CITE the article from this Journal.

Reviewer 1: Following are the few observations:

- 1. Author should add figure of location of thermocouple.
- 2. Heaters are placed after test section? How the heating of air takes place?
- 3. Details of perforations on rectangular and concave winglet is missing. pl add.
- 4. Provide details of validation of set up and heat loss analysis.
- 5. Mention pitch kept between two pins/winglet.
- 6. At lower Re both inline and staggered arrangement gives same result, while deviatio 3 after Re 8000. Author has to justify.
- 7. Provide more clarification about 1,2 & 3 pairs.
- 8. Compare the PRWP and PCRWP with without perforation.
- 9. Discuss how number of pairs contribute in improving TEF



Reviewer 2: This paper presented experimental results of the heat transfer, pressure drop and thermal performance characteristics of the perforated concave rectangular winglet pair vortex generators on plates in rectangular ducts to increase the heat transfer through the six heated tubes to the air stream. Perforated concave rectangular winglets were compared with perforated rectangular winglet pairs vortex generator mounted on rectangular plates. The subject of the article falls within the scope of the Results in Engineering. In my view, unless the paper is rewritten in a proper way, I think it is inadequate to be published in a scientific journal in its present state. I would like to provide the following comment:

- 1. There are many spelling and grammar mistakes in this paper, and many sentences are not easy to follow. The grammar and structure of sentences in this paper need to be modified carefully, such as the title of the article. The language of this paper need to be improved by a native English speaker.
- 2. The velocity ranges studied are given in the Abstract, whereas the Reynolds number ranges studied should be given.
- 3. Please review the keywords and add a few, for instance heat transfer, thermal performance.
- 4. The method section can be expanded.
- 5. The arrangement of vortex generators is given in Figure 2, but Figure 2 is not sufficient for a clear understanding of the construction of vortex generators. Additional figures should be drawn which clearly show the construction of vortex generators and the in-line and staggered arrangement.
- 6. Thermal characteristics are given in terms of the convective heat transfer coefficient (h), friction characteristics are given in terms of pressure drop. Why are thermal characteristics not given in terms of Nusselt number (Nu) and friction characteristics in terms of coefficient of friction (f)?
- 7. It is not specified how the hydraulic diameter (Dh) is calculated. How the Reynolds number (Re) was calculated is not specified.
- 8. No correlation (Nu-Re), (f-Re) is given.
- 9. Error analysis is given, but uncertainty analysis is not done.

Reviewer 3: In connection with climate change and an increase in the average annual temperature on Earth, there is a new danger of the negative impact of high temperatures on human life. In this case, the improvement of air conditioning systems, including the search for the best thermal enhancement factor, cost-benefit ratio, etc., takes on a new sense, which is one of the main targets of this article. The topic is timely and of great practical significance to environmental protection, enhancing safety, and people's life comfort.

The manuscript is well-structured and includes all necessary parts.

Two key strengths of the paper are a good introduction section and an analysis and discussion of the results. Both research objectives and content are clear. The key scientific issues to be solved are moderate. The research experimental method is reasonable.

There are also several shortages worthy to be mentioned:

- 1. Seriously revise the formulas
- a) If you use an italic font in formulas, use an italic font in their descriptions. For example, in formula (1), the Nusselt number (Nu) and friction factor (f); in formula (3), heat transfer coefficient (h). It may confuse the reader.
- b) The Nusselt numbers in formula (1) and formula (2) have different designations. It may confuse the reader.
- c) What are Nusselt number and friction factor with subscript 0 in formula (1)?
- d) In formula (5), a pressure drop is the lowercase letter Δp , but in formula (6), a pressure drop is uppercase ΔP . Are these different pressures?
- 2. What is the error of pressure drop measurement with the Fluke 922 Airflow Micromanometer described in section "3.2 Effect of perforated vortex generators on pressure drop"? Did " over the necessary range of pressures to be investigated? Could micromanometer error have aff of the conclusions of the section? Because the pressure drop values of 4.58 Pa, 5 Pa and 5.4 Pa are close to each other.
- 3. The measurement error in the section "3.3 Effect of perforated VGs on thermal enhancement factor" and "Effects of perforated VGs on the cost-benefit ratio" is not clear. Can you show the error bar or describe it in the description?

4. In my humble opinion, the section "3.5 Flow visualisation" is better presented first in section "3. Results and Discussion".

As a result, the article appears to be a qualified research paper. The results presented are consistent with the aims and scope of the journal. But now the article needs a major revision with correction of formulas and clarification of measurement errors.

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Ref.: Ms. No. RINENG-D-22-00804R1

Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel

Dear Miss heriyani,

Results in Engineering has received your revised submission.

You may check the status of your manuscript by logging onto Editorial Manager at (https://www.editorialmanager.com/rineng/).

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Date: Oct 14, 2022

To: "oktarina heriyani" oktarina@uhamka.ac.id **From:** "Ezio Mancaruso" ezio.mancaruso@stems.cnr.it

Subject: Your Submission

Ref.: Ms. No. RINENG-D-22-00804R1

Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel

Results in Engineering

Dear Miss heriyani,

I am pleased to tell you that your work has now been accepted for publication in Results in Engineering.

It was accepted on Oct 14, 2022

Comments from the Editor and Reviewers can be found below.

Thank you for submitting your work to this journal.

With kind regards

Ezio Mancaruso, Ph.D Associate Editor Results in Engineering

Comments from the Editors and Reviewers:

Reviewer 2: Dear Editor,

I reviewed the revised manuscript.

The authors have made most of the corrections I have asked of them. Therefore, the manuscript can be accepted for publication.

Kind regards,

Reviewer 3: The authors responded to all questions of interest, provided missing data, and corrected deficiencies. After the review, I recommend the article for publication in the journal.

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Author's Response To Reviewer Comments

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Dear Editors and Reviewer

Thank you for your letter and for the reviewers' comments concerning our manuscript entitled "Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel" (Manuscript Number: Rineng-D-22-00804R1). The comments are all valuable and very helpful to revise and improve our paper, as well as significant guidelines for our research. We have learned comments carefully and have made the correction that we hope you meet with your approval. We have included the parts requested to be revised in the manuscript. Revised portions are marked in red in the revised paper. The main correction in papers and responses to reviewing comments is flowing.

Reviewer 1: Following are the few observations:

Author should add figure of location of thermocouple.

Thank you very much for the proposal. I've revised Figure 1 because I made a mistake in captioning the figure. One thermocouple is placed in the air inlet area, six thermocouples on the back surface of the tubes, and 15 on the outlet side of the wire, as observed in Figure 1.

Heaters are placed after test section? How the heating of air takes place?

Thanks for the question. The heater is connected to six tubes with each tube getting the same power. The total heating power of 40 W is induced in the six tubes by a regulator. Heating air flowing through the tubes occurs by convection. So that the air at the outlet side becomes hotter than that from the inlet side.

Details of perforations on rectangular and concave winglet is missing, pl add.

Thank you for the correction. I have added in the paper the detailed geometry of the perforated rectangular winglet (PRW) and perforated concave rectangular winglet (PCRW) vortex generators (VGs), as shown in Figure 2.

Perforated RWP Perforated CRWP

VGs have dimensions of the same length and width of 30 mm and have 36 holes. The bore diameter on the VGs is 2.5 mm. The distance between the holes is 5 mm from the center of the holes.

Provide details of validation of set up and heat loss analysis. Set up validation

Thank you for your suggestion. The current study is a follow-up investigation of the w 3 and fid et al [1]. The experimental set-up of this study is similar to that of Yafid et al. experiment. The difference between the current study and the experiment of Yafid et al. is a test object where the current study uses concave rectangular winglet (CRW) VGs, while the work of Yafid et al. uses concave delta winglet (CDW) VGs. Whitaker et al. [2] studied the heat transfer characteristics of airflow through a single cylinder in a rectangular duct. To confirm the results of the experiment

Yafid et al. are valid, the same experimental set-up is determined. The Nu value from the experiment of Yafid et al. compared with Nu values from the experiments of Whitaker et al. in the Reynolds number range of 2,143 to 11,763, as shown in the figure below.

From the figure, it can be observed that Nu from the experimental results of Yafid et al. have the same trend as the experiments of Whitaker et al.

b. Heat Loss analysis

Heat loss analysis is carried out by taking into account the convection heat transfer from the six tubes to the surrounding fluid flow. Calculation of the heat transfer rate is carried out for two types of flow, namely laminar flow and turbulent flow.

The calculation of heat loss in this experiment is determined by calculating the difference between the induced electric power and the total heat through convection from the surface of the tubes to the fluid. In this experiment, six tubes in a wind tunnel are heated by a heater with a power of 40 W. In this work, the velocity of the inlet fluid is varied from 0.4 to 2 m/s at intervals of 0.2 m/s or in the Reynolds number range from 2,143 to 11,763. Based on the Reynolds number range, two types of flow are determined, namely laminar and turbulent. Therefore, the heat loss was determined from the correlation between laminar at 0.4 m/s and turbulent for other velocities. The described formulas Nu, h, and q were used to determine the heat loss in the conduit of the six tubes.

```
Nu=(q D_h)/(A_tube [\Delta T] LMTD k) (2)
```

 $h=q/(A_tube [\Delta T]_LMTD)(3)$

Where D_h, A_tube, $[\Delta T]$ _LMTD, m, c_p, T_out, T_in, are hydraulic diameter, tube surface area, log mean temperature difference, mass flow rate, specific heat, outlet temperature, and inlet temperature.

$$D h = [4A] c/P(5)$$

$$[\![\Delta T]\!]$$
 LMTD= ((T tube-T out)-(T tube-T in))/In[(T tube-T out)-(T tube-T in)] (6)

Where A_c dan T_tube are channel surface area and tube temperature, respectively. The experimental data for hydraulic diameter D_h,, tube surface area A_tube, channel surface area A_c, and air specific heat c_p are 0.09223 m, 0.02338908 m2, 0.01056 m2, and 1.007 J/kgK, respectively. The following is a table for calculating the heat loss baseline.

Table 1 Heat Loss Baseline

baseline v (m/s) Re Mass flow rate (kg/s) Density (kg/m3) Dynamics viscous (kg/ms) k Pr T inlet (C) T outlet (C) T tube (C) Δ T LMTD Δ T (T tube - T inlet) Nu h (W/mK) q conv (W) q input (W) q loss (W)

```
0.4 2165 0.004757 1.13 1.9.E-05 0.03 0.73 29 33 50 19 21 155 45 19.48 40 20.52
```

0.6 3291 0.00719 1.13 1.9.E-05 0.03 0.73 28 31 46 16 18 174 50 18.98 40 21.02

0.8 4413 0.009618 1.14 1.9.E-05 0.03 0.03 28 30 44 15 16 192 50 19.19 40 20.81

1 5545 0.012056 1.14 1.9.E-05 0.03 0.73 28 30 43 14 15 214 55 19.84 40 20.16

1.2 6661 0.014477 1.14 1.9.E-05 0.03 0.73 28 29 43 14 15 228 61 21.15 40 18.85

1.4 7826 0.016958 1.15 1.9.E-05 0.03 0.73 28 29 41 12 13 247 70 20.30 40 19.70

1.6 8965 0.019407 1.15 1.9.E-05 0.03 0.73 27 29 40 12 13 263 75 21.03 40 18.97

1.8 10110 0.021863 1.15 1.9.E-05 0.03 0.73 27 28 39 12 12 296 84 22.54 40 17.46 2 11272 0.024341 1.15 1.9.E-05 0.03 0.73 27 28 38 11 11 342 97 24.17 40 15.83

In the table 1, it can be seen that the greater the velocity with the increase in Re number, the

lower the heat loss. It can be seen that the heat flow from the heater does not only spread into the tube, but convection occurs to the outside of the tube. Heat output increases with increasing Re. That is, the higher the flow velocity, the greater the turbulence through the silinder and the higher the turbulence intensity. An increase in turbulence intensity between a cold airflow and a hot cylinder with constant surface temperature is caused by the airflow velocity [3]. In row-tube arrays, this recirculation area increases for the second and subsequent columns. A lower air velocity in the circulation region indicates less airflow in that region participating in the local heating process [4]. The heat loss of all conditions in this experiment is shown in table 2 below.

Table 2 Calculation of heat loss for the whole case

type VGs q conv (W) q input (W) q loss (W)

Baseline 20.74 40 19.26

PCRWPI1 25.15 40 14.85

PCRWPI2 27.55 40 12.45

PCRWPI3 27.61 40 12.39

PCRWPS1 26.43 40 13.57

PCRWPS2 26.43 40 13.57

PCRWPS3 27.94 40 12.06

PRWPI1 24.09 40 15.91

PRWPI2 27.25 40 12.75

PRWPI3 28.82 40 11.18

PRWPS1 23.94 40 16.06

PRWPS2 26.37 40 13.63

PRWPS3 28.12 40 11.88

Table 2 shows that the lowest heat loss occurs when three sets of PCRWPs are staggered. The placement of the VGs can increase heat transfer in square ducts as the VGs create longitudinal vortices that increase vortex strength in the wake region downstream of the tube. Longitudinal vortices make the overall temperature field more uniform, improve heat mixing and boundary layer modification, and improve heat transfer performance. A higher number of vortex generators creates more longitudinal vortices and greatly increases heat transfer [5], [3].

Mention pitch kept between two pins/winglet.

Thank you for the question. I've added Figure 3 (in the paper) showing the pitch between VGs for both inline and staggered configurations.

CRWP in-line CRWP staggered

RWP in-line RWP staggered

CRWP in-line CRWP staggered

RWP in-line RWP staggered

At lower Re both inline and staggered arrangement gives the same result, while deviation is observed after Re 8000. The author has to justify.

Thank you for your suggestion. In Figure 7(a) (in the paper), the convection heat transfer coefficient for the case of PRW VGs in-line has the same value as that of PCRW VGs ir 3 or staggered in a pair of VGs. In one pair of VGs, the longitudinal vortex is generated afte. The flow hits the VGs and weakens (He et al., 2013). This is in contrast to two and three pairs of VGs where the longitudinal vortex produced after striking the first VGs has amplified again when the flow strikes the second VGs and so on. Therefore, the value of the heat transfer coefficient in the case of a pair of PRW VGs has the same value as that of PCRW VGs at Reynolds numbers above 8,000.

Provide more clarification about 1,2 & 3 pairs.

Thank you for your suggestion. This study describes the cases of PCRW and PRW VGs for one, two, and three pairs. I have added explanations for cases one, two, and three pairs of perforated VGs to the paper.

For PRW and PCRW VGs in-line configurations with one, two, and three pairs are shown in Figures 4. For one pair, the VGs are placed on the left and right sides of the first row of tubes. VGs are placed on the first and third row tubes for two pairs. As for the three pairs, VGs are placed on the first, third, and fifth row tubes.

one pair PCRW inlline two pairs PCRW inline

three pairs PCRW inline

one pair PRW inline two pairs PRW inline

three pairs PRW inline

Figure 4. VGs pairs in-line configurations

For PRW and PCRW VGs staggered configurations with one, two, and three pairs are shown in Figures 5. For one pair, the VGs are placed on the right side of the first row tube and the left side of the second row tube. VGs are placed on the right side of the first, third row tubes and on the left side of the second and fourth tubes for two pairs. As for the three pairs, the VGs are placed on the right side of the first, third, fifth row of tubes and on the left side of the second, fourth and sixth tubes.

one pair PCRW staggered two pairs PCRW staggered

three pairs PCRW staggered

one pair PRW staggered two pairs PRW staggered

three pairs PRW staggered

Figure 5. VGs pairs staggered configurations

Compare the PRWP and PCRWP with without perforation.

Thank you for your request. In this study, an experiment was conducted to compare PRW VGs and PCRW VGs in improving heat transfer in a rectangular channel, as shown in Figures 7 to 9.

Comparison of convection heat transfer coefficient values.

Figure 7 provides a comparison of the convection heat transfer coefficients for PRW and PCRW. It can be seen that there was an increase in the convection heat transfer coefficient with a rise in Re due to an increase in flow vortices and high turbulence intensity in the channel [6], a' 3 side a reduction in the wage region and stagnation area for each increase in flow velocity [7]. The increase in heat transfer for staggered VGs was better than in-line for PCRW VGs with any number of pairs at the highest Re (11,000). The results in Figure 7 show that the PCRWP VGs worked better than the PRWP VGs, and the staggered arrangement of the former with three pairs gave the highest yield, of 153.5 W/m2.K, as shown in Figure 7(c). Meanwhile, two PCRW pairs (137.33)

W/m2.K, Figure 7(b)) were better than one (132.25 W/m2.K, Figure 7(a)) because VGs with a concave surface destabilise the centrifugal force of the fluid flow, which strengthens the flow vortices. This makes the mixing of the hot fluid near the wall with the cold fluid of the main flow more robust [8].

Pressure drop comparison

In general, the highest pressure drop was observed using PCRWP VGs with staggered configuration for all Reynolds numbers except for one pair of VGs, as shown in Figure 8. The highest pressure drop was found in PRWP VGs with in-line configuration at Reynolds numbers greater than 8,000. The pressure drop on the staggered VGs was found to be higher than that of the in-line due to the shorter distance between the VGs of the staggered configuration than that of the in-line [5].

TEF comparison

TEF is the thermal-hydraulic performance which is the ratio of the increase in heat transfer to the pressure drop ratio. In general, the highest TEF was observed in the use of PCRWP VGs with staggered configuration, as depicted in Figure 9. PCRW creates wider flow vortices that can reduce the wake area behind the cylinder. Reducing the wake area can reduce the recirculation zone. This affects the increased heat transfer from the back of the cylinder to the stream [3]. A large radius, high intensity antrior-posterior vortex can reduce the wake area. A reduction in the wake area increases the flow velocity behind the tube and reduces the recirculation area, resulting in an increased heat transfer in this area [9], [10].

CBR comparison

A low value of CBR means a more economical value from the use of VGs. In general, CBR on the use of PCRWP VGs with staggered configuration is the best, as informed in Figure 10. The lowest CBR value results were obtained with three pairs of staggered type VGs PCRW. Three pairs of VGs lower CBR than one and two VG pairs. This is because the installation of three pairs VGs results in a higher Nusselt number increase than one and two pairs VGs, resulting in a lower pressure drop increase and therefore a lower CBR. These results show that lower CBR improves thermal performance relative to resistivity [11].

Discuss how number of pairs contribute in improving TEF

Thank you for your question. Figure 9 shows the effect of the number of pairs and configuration of VGs on TEF, this is also found in Ref. (He et al., 2013), (Sun et al., 2020), and (Ranjan et al., 2022).

One pair

Two pairs

Three pairs

From the experimental results, as shown in Figure 9, the TEF with three pairs of VGs for both inline and staggered was the highest. The TEF for the case of 3 pairs of PCRWs was 5.02% greater than that of one and two pairs of PRWP VGs. The main reason is because the concave surface of the PCR causes the flow to be thrown away due to the centrifugal force which results in a stronger flow vorticity [12]. Larger and stronger flow vortices can reduce the recirculation zone which has an impact on increasing heat transfer from the rear surface of the tube to the flow [10]. The presence of flow that is formed in each gap between the VGA and the tube causes the TEF in the staggered configuration to be greater than that of the in-line [13]. The increase in TEF for the three-pair case with the staggered configuration was 1.50% and 4.91% greater than that of the inline PCRWP and PRWP, respectively.

Reference

[1] Y. Effendi, A. Prayogo, Syaiful, M. Djaeni, and E. Yohana, "Effect of perforated concave delta winglet vortex generators on heat transfer and flow resistance through the heated tubes in the channel," Experimental Heat Transfer, vol. 35, no. 5, pp. 553–576, 2022, doi:

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- [5] Y. L. He, P. Chu, W. Q. Tao, Y. W. Zhang, and T. Xie, "Analysis of heat transfer and pressure drop for fin-and-tube heat exchangers with rectangular winglet-type vortex generators," Appl Therm Eng, vol. 61, no. 2, pp. 770–783, Nov. 2013, doi: 10.1016/J.APPLTHERMALENG.2012.02.040.
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Reviewer 2: This paper presented experimental results of the heat transfer, pressure drop and thermal performance characteristics of the perforated concave rectangular winglet pair vortex generators on plates in rectangular ducts to increase the heat transfer through the six heated tubes to the air stream. Perforated concave rectangular winglets were compared with perforated rectangular winglet pairs vortex generator mounted on rectangular plates. The subject of the article falls within the scope of the Results in Engineering. In my view, unless the paper is region in a proper way, I think it is inadequate to be published in a scientific journal in its present attended. I would like to provide the following comment:

There are many spelling and grammar mistakes in this paper, and many sentences are not easy to follow. The grammar and structure of sentences in this paper need to be modified carefully, such as the title of the article. The language of this paper need to be improved by a native English speaker.

Thank you for your suggestion. To improve grammar and structure, I have sent this paper to elsevier service for proof reading.

The velocity ranges studied are given in the Abstract, whereas the Reynolds number ranges studied should be given.

Thank you for your correction. In this experiment, the intake air velocity is in the range of 0.4 to 2 m/s at intervals of 0.2 m/s or in the Reynolds number range of 2.143 to 11.763. I have included the Reynolds number range in the abstract.

Please review the keywords and add a few, for instance heat transfer, thermal performance. Thank you for your suggestion. The keywords for this study are vortex generators, heat transfer, thermal-hydraulic performance, economic benefit. I've added keywords in the paper.

The method section can be expanded.

Thank you for your suggestion. I have improved this research method in the paper. Based on Fig. 1, the rectangular channel is equipped with a blower (50 Hz, Wipro with a rated voltage of 220V), an inverter (Mitsubishi Electric type FR-D700 with an accuracy of 0.01), straightener, hot wire anemometer (Lutron type AM-4204 with an accuracy of 0.1), wattmeter (Lutron DW-6060 with an accuracy \pm 1.0), central processing unit (CPU), micromanometer, thermocouple (K type with a temperature interval of $-200 - 1250^{\circ}$ C and an accuracy ± 0.5) where one thermocouple was placed in the air inlet area, six thermocouples on the back surface of the tubes and 15 on the outlet side of the wire, data acquisition (Advantech USB-4718 type with an accuracy of 0.001) and heater regulator. The heater was connected to six tubes with a diameter of 19.05 mm and height of 65.8 mm, with each tube having the same power. Total heating power of 40 W was applied to the six tubes using a regulator. The heating air flowing through the tubes occurs via convection. Thus, the air at the outlet side becomes hotter than that at the inlet side. A pressure micromanometer (Fluke type 922, with an accuracy of \pm 0.05) was used to monitor the flow pressure drop. Two pitot tubes, each set 26 cm ahead of the inlet of the test specimen and 2.5 cm behind it, were connected to a micromanometer to measure the pressure drop. The pressure drop measurements were recorded 30 times for 5 sekon at each speed variation. Furthermore, flow visualisation was performed by directing the smoke from vaporised fluid in the fluid vaporator into the mainflow.

The arrangement of vortex generators is given in Figure 2, but Figure 2 is not sufficient for a clear understanding of the construction of vortex generators. Additional figures should be drawn which clearly show the construction of vortex generators and the in-line and staggered arrangement. Thank you for your suggestion. The following shows the construction of vortex generators inline and staggered (figure 2, 4 and 5 in the paper).

Figure 2. Geometry of the VGs

For PRW and PCRW VGs in-line configurations with one, two, and three pairs are shown in Figures 4. For one pair, the VGs are placed on the left and right sides of the first row of tubes. VGs are placed on the first and third row tubes for two pairs. As for the three pairs, VGs are placed on the first, third, and fifth row tubes.

one pair PCRW inlline two pairs PCRW inline

three pairs PCRW inline

3

one pair PRW inline two pairs PRW inline

three pairs PRW inline

Figure 4. VGs pairs in-line configurations

For PRW and PCRW VGs staggered configurations with one, two, and three pairs are shown in Figures 5. For one pair, the VGs are placed on the right side of the first row tube and the left side of the second row tube. VGs are placed on the right side of the first, third row tubes and on the left side of the second and fourth tubes for two pairs. As for the three pairs, the VGs are placed on the right side of the first, third, fifth row of tubes and on the left side of the second, fourth and sixth tubes.

one pair PCRW staggered two pairs PCRW staggered

three pairs PCRW staggered

one pair PRW staggered two pairs PRW staggered

three pairs PRW staggered

Figure 5. VGs pairs staggered configurations

Thermal characteristics are given in terms of the convective heat transfer coefficient (h), friction characteristics are given in terms of pressure drop. Why are thermal characteristics not given in terms of Nusselt number (Nu) and friction characteristics in terms of coefficient of friction (f)?

Thank you for your correction. h versus Re and P vs Re are used instead of Nu - Re and f - Re in this experiment because, in the TEF calculation, the value of Nu represents the value of h resulting from the equation in formula 3 (in the paper), thus

 $h=(N_u k)/D_h (3)$

Figure 7 (in the paper) that h rises as Re rises shows that the Nu increases as Re rises, where h rises as Re rises [1]. While f in this experiment represents the ΔP as shown in the following formula

 $f=(2 \Delta P D_h)/(\rho V^2 (L+6D)) (8)$

Formula 8 states that the friction factor (f) in the flow rate is determined by using the pressure drop (ΔP) characteristic where increasing the Reynolds number in figure 8 will decrease the friction factor [2].

The following are examples of several studies that use h as a representative of Nu and (ΔP) as a representative of f.

The experimental results of Yafid et al indicate that perforated VGs can increase the heat transfer rate and decrease the pressure drop using the parameters h, ΔP , and TEF as shown in the following graph[3].

Al Asadi et al represents heat transfer coefficient and pressure drop to show the results of their investigation that the addition of span wise gap variations can increase heat transfer performance and reduce pressure drop [4], as shown in the figure below.

3

Increased heat transfer in heat sinks, Zhang et al described the micro gap by pairing more VGs which resulted in a larger heat transfer coefficient and a reduced pressure drop value [5], as shown in the figure below.

Hosseinirad et al showed that the increase in heat transfer coefficient and pressure drop vs. Reynold number had a tendency to increase with the increase in Re to evaluate heat transfer [6]. There is an increase in heat transfer with an increase in Re which is indicated by an increase in the heat transfer coefficient and an increase in pressure drop along with an increase in Re.

It is not specified how the hydraulic diameter (Dh) is calculated. How the Reynolds number (Re) was calculated is not specified.

Thank you for your corrections. The calculation of the hydraulic diameter in this experiment uses a rectangular air duct with a side length of $a=0.165\,\mathrm{m}$ and a side width of $b=0.064\,\mathrm{m}$ with the resulting Dh of 0.0106 from the following formula.

 $D_h=(4A_c)/p=4ab/2(a+b)=2ab/(a+b)$ (5)

The result of Dh is used to calculate Re with the formula $Re=(\rho u_i n D_h)/\mu$ (7)

With u_i of 0.4 – 2 m/s with an interval of 0.2 m/s, on the physical properties of air at a pressure of 1 atm and is the viscosity of the fluid so that Re used in this experiment ranges from 2,143 – 11,763.

No correlation (Nu-Re), (f-Re) is given.

Thank you for the correction. In this experiment, Nu – Re and f – Re are not shown but use h vs Re and ΔP vs Re because the Nu value represents the value of h that arises in this experiment as in equation 4 (in the manuscript)

 $h=(N_u k)/D_h (3)$

While f in this experiment represents the ΔP as shown in the following formula

 $f=(2 \Delta P D_h)/(\rho V^2 (L+6D)) (8)$

The formula states that the friction factor (f) in the flow rate is determined by using the pressure drop (ΔP) characteristic where increasing the Reynolds number in figure 8 will decrease the friction factor [2].

Error analysis is given, but uncertainty analysis is not done.

Thank you for the correction. In the following, uncertainty analysis calculation data will be shown for the temperature at base line conditions with a velocity of 0.4 m/s as shown in Table 3 below.

Table 3 Base-line test temperature data at a speed of 0.4 m/s

T (Tube1) T (Tube2) T (Tube3) T (Tube4) T (Tube5) T (Tube6)

49.19093 51.21368 48.32313 49.76915 47.80219 51.27142

49.1834 51.17728 48.3156 49.79053 47.7657 51.2639

49.14545 51.16826 48.30655 49.7526 47.7856 51.25489

49.12105 51.17277 48.28214 49.72821 47.76118 51.2594

49.15297 51.20465 48.28515 49.73122 47.73524 51.2624

49.09966 51.15141 48.28967 49.73573 47.76871 51.26691

49.09815 51.14991 48.23029 49.73423 47.73826 51.29428

49.08912 51.14089 48.25019 49.66739 47.72922 51.22751

From these data, it is found that T_{tobe} can be calculated by the equation as $T_{tobe} = (T_{tobe} + T_{tobe} + T_{$

s_tube= $\sqrt{((\Sigma_{i=1})^N_{i=1}^T tube)^2)/N(N-1)}=0.029$ (12) Therefore, the average T_tube can be written as 49.5 \pm 0.029°C. T_out calculation results

```
obtained 32.95°C. The average standard deviation was calculated using the following equation:
s_{\text{Tout}} = \sqrt{((\Sigma_{(i=1)}^N) (T_{\text{outi}} - T_{\text{out}})^2)/N(N-1))} = 0.051 (13)
Furthermore, the average value of T out can be written as 32.95 ± 0.051°C. Using the same
equation, the standard deviation of T in was found to be 0.033. Thus, the average T in value was
29.75 ± 0.016°C.
The value of q at a speed of 0.4 m/s was found to be 19.48 W. To determine of the standard
deviation of q, the following equation was used:
 [RSS]] q=\sqrt{([s(\Delta T)] \text{ out})\partial q/(\partial T \text{ out}))^2+([s(\Delta T)] \text{ in})\partial q/(\partial T \text{ in}))^2)(14)}
\partial q/(\partial T_{out}) = (\partial (m \cdot c_p \cdot T_{out} - m \cdot c_p \cdot T_{in}))/(\partial T_{out}) = m \cdot c \cdot p
\partial q/(\partial T_{in}) = (\partial (m \cdot c_p \cdot T_{out} - m \cdot c_p \cdot T_{in}))/(\partial T_{in}) = -(m \cdot c \cdot p)
where [s(\Delta T)] out)=0.051°C and [s(\Delta T)] in)=0.033°C, ensuring, that [RSS] q= \pm 0.290 W.
Therefore, the heat transfer rate q becomes 19.48 \pm 0.290 W. The value of [\Delta T] Imtd at a speed
of 0.4 m/s was found to be 18.56°C. To determine the value of the standard deviation of [\![\Delta T]\!]
 Imtd we used the following equation:
 [RSS] _( [\DeltaT] _lmtd ) = \sqrt{([s(\Delta T)] _tube)} \partial([\Delta T] _lmtd )/(\partial T_tube ))^2 + ([s(\Delta T)] _tube)} \partial([\Delta T] _tube) \partial([\Delta T] _tube)
 \cot \theta ( \Delta T ) = 1  \cot \theta ( \Delta T ) = 1 
∂(〖ΔΤ〗_lmtd )/(∂T_tube )=∂(((T_tube-T_out )-(T_tube-T_in ))/ln〖(T_tube-T_out)/(T_tube-T_in
)] )/(∂T tube )
\partial( \Delta T) = \frac{1}{\Delta T}  in \frac{1}{\Delta T} = \frac{1}{\Delta T} = \frac{1}{\Delta T}  in \frac{1}{\Delta T} = \frac{1}{\Delta T} 
\partial( \Delta T) = \operatorname{Imtd} /(\partial T_{in}) = \partial(((T_{tube}-T_{out}) - (T_{tube}-T_{in})) / (T_{tube}-T_{out}) / (T_{tube}-T_{in}))
)/(\partial T \text{ in })
where [s(\Delta T)] tube = 0,029°C, [s(\Delta T)] out = 0,051°C and [s(\Delta T)] in = 0,033°C; we get
 [RSS] ([\DeltaT]] Imtd ) of \pm 0.043, ensuring, that the obtained [\DeltaT]] Imtd is 8.56 \pm 0.043.
The value of Nu at a speed of 0.4 m/s was found to be 155.31. The standard deviation of Nu was
obtained using following equation
 [RSS] _Nu=\sqrt{(s(q)\partial Nu/\partial Q)^2+[s(\Delta T)] - Imtd)\partial Nu/(\partial [\Delta T]] - Imtd))} (16)
\partial Nu/\partial q = \partial (q \cdot D_h. [At] ^{-1} \cdot [[\Delta T]] \lim d[AT] ^{-1} \cdot k^{-1})/\partial q = D_h/((At)([\Delta T]] \lim d(k))
\partial \text{Nu}/(\partial \ [\![\Delta T]\!] \ \_\text{Imtd}\ ) = \partial (q \cdot \ [\![D_h \cdot \text{At}]\!] \ ^{(-1)} \cdot \ [\![\Delta T]\!] \ \_\text{Imtd}\ ] \ ^{(-1)}.k^{(-1)})/(\partial \ [\![\Delta T]\!] \ \_\text{Imtd}\ ) =
(g.D h)/((At)( \Delta T) Imtd )^2 (k)
With the values of s(q)=0.290 W and [s(\Delta T)]_{mtd}=0.043, the obtained [RSS]_{mu} Wu was \pm
2.889 W/(m2°C). Therefore, the value of [RSS] Nu is 155.31 \pm 2.889 W/m2°C.
The value of h at a speed of 0.4 m/s was found to be 44.86. To determine the standard deviation of
Nu the following equation is used
 [RSS]] h=\sqrt{((s(Nu)\partial h/Nu)^2)(17)}
\partial h/\partial Nu = \partial (h.D h.k^{(-1)})/\partial h = k/D h
Furthermore, the value of D h is 0.092 m and k at T f= 40.24 is 0.026. So the value of h at a
speed of 0.4 m/s is:
 [RSS]] h=\sqrt{((s(Nu)\partial h/Nu)^2)} = 0.83
Thus, the number h at a speed of 0.4 m/s is 44.86 \pm 0.83. So, the error h for the baseline at a
speed of 0.4 m/s is
Error= [RSS] _h/h \times 100 (18)
Error=0.83/44.86×100=1.51%
From the test in the baseline case with a speed of 2.0 m/s, the results of the pressure drop are
listed in Table 4, which show that the average P can be calculated as follows:
(\Delta P)^{-} = ( [\![ \Delta P ]\!] _1 + [\![ \Delta P ]\!] _2 + [\![ \Delta P ]\!] _3 + \dots + [\![ \Delta P ]\!] _3 0)/30 = 3.51 Pa (19)
The average standard deviation of the pressure drop can then be calculated using the equation
s = \sqrt{((\Sigma_{i=1})^{N} ([\Delta P]_{i-(\Delta P)^{-}})^{2})/N(N-1)} = 8.9 \times 10^{(-5)} (20)
Baseline case for the pressure drop value at a speed of 2.0 m/s is 3.51±8.9×10^(-5) Pa. Then, the
error in the form of percentage can be calculated using the following equation:
(8.9 \times 10^{(-5)})/3.51 \times 100 = 0.71
Table 4 Baseline pressure drop data at a speed of 2.0 m/s
                                                                                                                                                                                              3
\Delta P (Pa)
Data to 2.0 m/s Data to 2.0 m/s
1 0.013 16 0.012
2 0.013 17 0.013
3 0.013 18 0.012
```

```
4 0.013 19 0.012
5 0.012 20 0.013
6 0.013 21 0.013
7 0.013 22 0.012
8 0.012 23 0.013
9 0.013 24 0.012
10 0.013 25 0.013
11 0.013 26 0.013
12 0.013 27 0.013
13 0.012 28 0.013
14 0.012 29 0.012
15 0.013 30 0.012
```

The equal calculation approach changed into used for all data. Therefore, the overall error outputs for the pressure-drop vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are listed in Table 5.

Table 5. Overall Pressure Drop (ΔP)

Vortex Generator Variations Overall Error P

(perforated)

```
1 PRWP in-line 2.94%
```

2 PRWP in-line 2.87%

3 PRWP in-line 1.98%

1 PRWP staggered 2.88%

2 PRWP staggered 2.34%

3 PRWP staggered 1.36%

1 PCRWP in-line 2.72%

2 PCRWP in-line 1.80%

3 PCRWP in-line 1.80%

1PCRWPstaggered 2.43%

2PCRWPstaggered 1.91%

3 PCRWP staggered 0.97%

The average TEF results from the experimental results can be calculated as follows.

 $(TEF) = ([TEF] _1 + [TEF] _2 + [TEF] _3 + \dots + [TEF] _12)/12 = 1.12 (21)$

Then, the average standard deviation of the TEF can be calculated with the equation $s=\sqrt{([\Sigma_{i=1}^{n}]^{-i-(TEF)^{-i})^2})/N(N-1)}=1.07$ (22)

Therefore, the TEF value was 1.12 ± 1.07 . Then, the error in the form of percentage can be calculated using the following equation:

1.07/1.12×100=0.94%

The overall error results for the TEF vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are are listed in Table 6.

Table 6. Overall error TEF

Variasi Vortex Generator Overall Error TEF

(Berlubang)

```
1 RWP in-line 0.47 %
```

2 RWP in-line 0.47%

3 RWP in-line 0.43%

1 RWP staggered 0.47%

2 RWP staggered 0.47%

3 RWP staggered 0.43%

1 CRWP in-line 0.45%

2 CRWP in-line 0.45%

3 CRWP in-line 0.42%

1 CRWP staggered 0.45%

2 CRWP staggered 0.45%

3 CRWP staggered 0.41%

3

First, find the average CBR of the experimental results with the following formula. $(CBR) = ((CBR) _1 + (CBR) _2 + (CBR) _3 + \cdots + (CBR) _1 + (CBR) _2 + (CBR) _3 + \cdots + ($

The average standard deviation of the pressure drop CBR can then be calculated using the following equation:

 $s=\sqrt{((\Sigma_{i=1})^N[(CBR)_i-(CBR)^-)^2)/N(N-1)}=1.60 (24)$

The CBR value is 2.14 ± 1.60 . Then the error in the form of percentage can be calculated using the following equation:

1.60/2.14×100=0.63%

The overall error results for the CBR vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are listed in Table 7 Table 7. Overall error CBR

Variasi Vortex Generator Overall Error CBR

(Berlubang)

- 1 RWP in-line 0.32%
- 2 RWP in-line 0.29%
- 3 RWP in-line 0.45%
- 1 RWP staggered 0.32%
- 2 RWP staggered 0.31%
- 3 RWP staggered 0.45%
- 1 CRWP in-line 0.4%
- 2 CRWP in-line 0.42%
- 3 CRWP in-line 0.56%
- 1 CRWP staggered 0.43%
- 2 CRWP staggered 0.42%
- 3 CRWP staggered 0.66%

Reference

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3

Reviewer 3: In connection with climate change and an increase in the average annual temperature on Earth, there is a new danger of the negative impact of high temperatures on human life. In this case, the improvement of air conditioning systems, including the search for the best thermal enhancement factor, cost-benefit ratio, etc., takes on a new sense, which is one of the main targets of this article. The topic is timely and of great practical significance to environmental protection, enhancing safety, and people's life comfort.

The manuscript is well-structured and includes all necessary parts.

Two key strengths of the paper are a good introduction section and an analysis and discussion of the results. Both research objectives and content are clear. The key scientific issues to be solved are moderate. The research experimental method is reasonable.

There are also several shortages worthy to be mentioned:

Seriously revise the formulas

If you use an italic font in formulas, use an italic font in their descriptions. For example, in formula (1), the Nusselt number (Nu) and friction factor (f); in formula (3), heat transfer coefficient (h). It may confuse the reader.

Thank you for the corrections. For formula and description fonts, improvements have been made where all formula and description fonts are italicized consistently.

The Nusselt numbers in formula (1) and formula (2) have different designations. It may confuse the reader.

Thanks for the correction. Consistent improvements have been made to writing the Nu symbol on paper.

What are Nusselt number and friction factor with subscript 0 in formula (1)?

Thanks for the corrections. The subscript 0 for Nusselt number and friction factor is meant for the baseline condition. This additional explanation has been added to the paper. The following is an explanation of formula 1.

TEF= $((Nu/[Nu]_0))/(f/f_0)^(1/3)(1)$

Di mana: [Nu] = 0 Nusselt number pada kondisi baseline $f_0 = f$ friction factor pada kondisi baseline

In formula (5), a pressure drop is the lowercase letter Δp , but in formula (6), a pressure drop is uppercase ΔP . Are these different pressures?

Thank you for the corrections. I'm so sorry for the error in writing the pressure drop symbol which is inconsistent. Improvements in writing pressure drop have been made with upperca $_3$ for the formula on the paper.

What is the error of pressure drop measurement with the Fluke 922 Airflow Micromanometer described in section "3.2 Effect of perforated vortex generators on pressure drop"? Did it cover the

necessary range of pressures to be investigated? Could micromanometer error have affected the conclusions of the section? Because the pressure drop values of 4.58 Pa, 5 Pa and 5.4 Pa are close to each other.

Thanks for the question. Pressure measurement errors with the Fluke 922 Airflow Micromanometer are explained in the uncertainty analysis section with the calculation results as below.

From the test in the baseline case with a speed of 2.0 m/s, the results of the pressure drop are listed in Table 4, which show that the average P can be calculated as follows:

 $(\Delta P)^{-} = ([\Delta P] _1 + [\Delta P] _2 + [\Delta P] _3 + \dots + [\Delta P] _30)/30 = 3.51 Pa (19)$

The average standard deviation of the pressure drop can then be calculated using the equation $s=\sqrt{((\Sigma_{(i=1)}^N)[(\Delta P)]_{i-(\Delta P)}^-)^2)/N(N-1)}=8.9\times10^{-5}$ (20)

Baseline case for the pressure drop value at a speed of 2.0 m/s is $3.51\pm8.9\times10^{-5}$ Pa. Then, the error in the form of percentage can be calculated using the following equation: $(8.9\times10^{-5})/3.51\times100=0.71$

Table 4 Baseline pressure drop data at a speed of 2.0 m/s ΔP (Pa)

Data to 2.0 m/s Data to 2.0 m/s

1 0.013 16 0.012

2 0.013 17 0.013

3 0.013 18 0.012

4 0.013 19 0.012

5 0.012 20 0.013

6 0.013 21 0.013

7 0.013 22 0.012

8 0.012 23 0.013

9 0.013 24 0.012

10 0.013 25 0.013

11 0.013 26 0.013

12 0.013 27 0.013

13 0.012 28 0.013

14 0.012 29 0.012

15 0.013 30 0.012

The equal calculation approach changed into used for all data. Therefore, the overall error outputs for the pressure-drop vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are listed in Table 5.

Table 5. Overall Pressure Drop (ΔP)

Vortex Generator Variations Overall Error P

(perforated)

1 PRWP in-line 2.94%

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2 PRWP staggered 2.34%

3 PRWP staggered 1.36%

1 PCRWP in-line 2.72%

2 PCRWP in-line 1.80%

3 PCRWP in-line 1.80%

1 PCRWP staggered 2.43%

2 PCRWP staggered 1.91%

3 PCRWP staggered 0.97%

3

The results of the measurement error calculation are still below the maximum accuracy limit of the tool by 5% so that it does not affect the conclusion section.

The measurement error in the section "3.3 Effect of perforated VGs on thermal enhancement

factor" and "Effects of perforated VGs on the cost-benefit ratio" is not clear. Can you show the error bar or describe it in the description?

Thanks for the question. The measurement error for the thermal increase factor and the cost benefit ratio has been added to the explanation in the paper in the data uncertainty section, as follows.

error bar TEF

The average TEF results from the experimental results can be calculated as follows.

 $(TEF)^{-}=((TEF) 1+ (TEF) 2+ (TEF) 3+\cdots + (TEF) 12)/12=1.12(21)$

Then, the average standard deviation of the TEF can be calculated with the equation $s=\sqrt{((\Sigma_{i=1})^N)[(TEF)_{i=1}^N)/(N-1)}=1.07$ (22)

Therefore, the TEF value was 1.12 ± 1.07 . Then, the error in the form of percentage can be calculated using the following equation:

1.07/1.12×100=0.94%

The overall error results for the TEF vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are are listed in Table 6.

Table 6. Overall error TEF

Variasi Vortex Generator Overall Error TEF

(Berlubang)

- 1 RWP in-line 0.47 %
- 2 RWP in-line 0.47%
- 3 RWP in-line 0.43%
- 1 RWP staggered 0.47%
- 2 RWP staggered 0.47%
- 3 RWP staggered 0.43%
- 1 CRWP in-line 0.45%
- 2 CRWP in-line 0.45%
- 3 CRWP in-line 0.42%
- 1 CRWP staggered 0.45%
- 2 CRWP staggered 0.45%
- 3 CRWP staggered 0.41%

b error bar CBR

First, find the average CBR of the experimental results with the following formula. $(CBR)^-=((CBR))^-=(($

The average standard deviation of the pressure drop CBR can then be calculated using the following equation:

 $s=\sqrt{((\Sigma_{i=1})^N ([CBR]_{i-(CBR)}^-)^2)/N(N-1)} = 1.60 (24)$

The CBR value is 2.14 ± 1.60 . Then the error in the form of percentage can be calculated using the following equation:

1.60/2.14×100=0.63%

The overall error results for the CBR vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are listed in Table 7.

Table 7. Overall error CBR

Variasi Vortex Generator Overall Error CBR

(Berlubang)

- 1 RWP in-line 0.32%
- 2 RWP in-line 0.29%
- 3 RWP in-line 0.45%
- 1 RWP staggered 0.32%
- 2 RWP staggered 0.31%
- 3 RWP staggered 0.45%
- 1 CRWP in-line 0.4%
- 2 CRWP in-line 0.42%
- 3 CRWP in-line 0.56%
- 1 CRWP staggered 0.43%

3

2 CRWP staggered 0.42% 3 CRWP staggered 0.66%

In my humble opinion, the section "3.5 Flow visualisation" is better presented first in section "3. Results and Discussion".

Thanks for the suggestions. The discussion of section 3.5 on visualization has been moved to the earlier section to 3.1 in the paper.

We tried our best to improve the manuscript and made some changes in the revised paper, and here we did not list the specific changes but marked in red in revised paper. We appreciate for Editors and Reviewrs' warm work earnestly, and hope that the correction will meet with approval. Once again, thank you very much for your comments and suggestions.

Yours Sincerely Oktarina Heriyani

Close

Results in Engineering

Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel --Manuscript Draft--

| Manuscript Number: | RINENG-D-22-00804R1 | |
|---|--|--|
| Full Title: | Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel | |
| Short Title: | PCRWP VGs enhance the heat transfer of air flowing through heated tubes inside a channel | |
| Article Type: | Research paper | |
| Section/Category: | Mechanical Engineering | |
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| Abstract: | A significant increase in the rate heat transfer in a heat exchanger system is made possible by increasing the convection heat-transfer coefficient using a passive method. The addition of vortex generators (VGs) to the fins and tubes of a heat exchanger is currently the most effective passive method. However, the increase in heat was accompanied by an increase in pressure drop. Therefore, in this study, we installed perforated concave rectangular winglet pair vortex generators (PCRWP VGs) on plates in rectangular ducts to increase the heat transfer through the six heated tubes to the air stream by lowering the enhancement in the pressure drop. We attempted to determine the best cost-benefit ratio (CBR) with a fluid flow velocity difference of 0.4 –2 m/s at intervals of 0.2 m/s (Reynolds number (Re) of 2,143 to 11,763) in the channel. The PCRWP VGs were composed of in-line and staggered configurations. The results showed a lower CBR (3.56) for the in-line configuration than for the staggered configuration. Moreover, the lowest CBR was accompanied by an increase in thermal performance (TEF) of 1.29. | |
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his research is focused on vortex generators

Response to Reviewers:

Dear Editors and Reviewer

Thank you for your letter and for the reviewers' comments concerning our manuscript entitled "Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel" (Manuscript Number: Rineng-D-22-00804R1). The comments are all valuable and very helpful to revise and improve our paper, as well as significant guidelines for our research. We have learned comments carefully and have made the correction that we hope you meet with your approval. We have included the parts requested to be revised in the manuscript. Revised portions are marked in red in the revised paper. The main correction in papers and responses to reviewing comments is flowing. Reviewer 1: Following are the few observations:

Authorist and add figure of the ethics of the great and th

Author should add figure of location of thermocouple.

Thank you very much for the proposal. I've revised Figure 1 because I made a mistake in captioning the figure. One thermocouple is placed in the air inlet area, six thermocouples on the back surface of the tubes, and 15 on the outlet side of the wire, as observed in Figure 1.

Heaters are placed after test section? How the heating of air takes place?

Thanks for the question. The heater is connected to six tubes with each tube getting the same power. The total heating power of 40 W is induced in the six tubes by a regulator. Heating air flowing through the tubes occurs by convection. So that the air at the outlet side becomes hotter than that from the inlet side.

Details of perforations on rectangular and concave winglet is missing. pl add. Thank you for the correction. I have added in the paper the detailed geometry of the perforated rectangular winglet (PRW) and perforated concave rectangular winglet (PCRW) vortex generators (VGs), as shown in Figure 2.

Perforated RWPPerforated CRWP

VGs have dimensions of the same length and width of 30 mm and have 36 holes. The bore diameter on the VGs is 2.5 mm. The distance between the holes is 5 mm from the center of the holes.

Provide details of validation of set up and heat loss analysis. Set up validation

Thank you for your suggestion. The current study is a follow-up investigation of the work of Yafid et al. [1]. The experimental set-up of this study is similar to that of Yafid et al. experiment. The difference between the current study and the experiment of Yafid et al. is a test object where the current study uses concave rectangular winglet (CRW) VGs, while the work of Yafid et al. uses concave delta winglet (CDW) VGs. Whitaker et al. [2] studied the heat transfer characteristics of airflow through a single cylinder in a rectangular duct. To confirm the results of the experiment Yafid et al. are valid, the same experimental set-up is determined. The Nu value from the experiment of Yafid et al. compared with Nu values from the experiments of Whitaker et al. in the Reynolds number range of 2,143 to 11,763, as shown in the figure below.

From the figure, it can be observed that Nu from the experimental results of Yafid et al. have the same trend as the experiments of Whitaker et al.

b. Heat Loss analysis

Heat loss analysis is carried out by taking into account the convection heat transfer

from the six tubes to the surrounding fluid flow. Calculation of the heat transfer rate is carried out for two types of flow, namely laminar flow and turbulent flow. The calculation of heat loss in this experiment is determined by calculating the difference between the induced electric power and the total heat through convection from the surface of the tubes to the fluid. In this experiment, six tubes in a wind tunnel are heated by a heater with a power of 40 W. In this work, the velocity of the inlet fluid is varied from 0.4 to 2 m/s at intervals of 0.2 m/s or in the Reynolds number range from 2,143 to 11,763. Based on the Reynolds number range, two types of flow are determined, namely laminar and turbulent. Therefore, the heat loss was determined from the correlation between laminar at 0.4 m/s and turbulent for other velocities. The described formulas Nu, h, and q were used to determine the heat loss in the conduit of the six tubes.

```
\label{eq:nu} \begin{split} &\text{Nu=(q D_h)/(A_tube} \quad \text{$\llbracket \Delta T \rrbracket$ \_LMTD k)(2)$} \\ &\text{h=q/(A_tube} \quad \text{$\llbracket \Delta T \rrbracket$ \_LMTD )(3)$} \\ &\text{q=m'} \quad \text{c_p} \quad \text{(T_out-T_in )(4)} \end{split}
```

Where D_h, A_tube, $[\![\Delta T]\!]$ _LMTD, m', c_p, T_out, T_in, are hydraulic diameter, tube surface area, log mean temperature difference, mass flow rate, specific heat, outlet temperature, and inlet temperature.

Where A_c dan T_tube are channel surface area and tube temperature, respectively. The experimental data for hydraulic diameter D_h,, tube surface area A_tube, channel surface area A_c, and air specific heat c_p are 0.09223 m, 0.02338908 m2, 0.01056 m2, and 1.007 J/kgK, respectively. The following is a table for calculating the heat loss baseline.

```
Table 1 Heat Loss Baseline
```

baselinev (m/s)ReMass flow rate (kg/s)Density (kg/m3)Dynamics viscous (kg/ms)kPrT inlet (C)T outlet (C)T tube (C)T LMTDT (T tube - T inlet)Nuh (W/mK)q conv (W)q input (W)q loss (W)

0.421650.0047571.131.9.E-050.030.7329335019211554519.484020.52 0.632910.007191.131.9.E-050.030.7328314616181745018.984021.02 0.844130.0096181.141.9.E-050.030.0328304415161925019.194020.81 155450.0120561.141.9.E-050.030.7328304314152145519.844020.16 1.266610.0144771.141.9.E-050.030.7328294314152286121.154018.85 1.478260.0169581.151.9.E-050.030.7328294112132477020.304019.70 1.689650.0194071.151.9.E-050.030.7327283912122968422.544017.46 2112720.0243411.151.9.E-050.030.7327283811113429724.174015.83

In the table 1, it can be seen that the greater the velocity with the increase in Re number, the lower the heat loss. It can be seen that the heat flow from the heater does not only spread into the tube, but convection occurs to the outside of the tube. Heat output increases with increasing Re. That is, the higher the flow velocity, the greater the turbulence through the silinder and the higher the turbulence intensity. An increase in turbulence intensity between a cold airflow and a hot cylinder with constant surface temperature is caused by the airflow velocity [3]. In row-tube arrays, this recirculation area increases for the second and subsequent columns. A lower air velocity in the circulation region indicates less airflow in that region participating in the local heating process [4]. The heat loss of all conditions in this experiment is shown in table 2 below.

Table 2 Calculation of heat loss for the whole case type VGsq conv (W)q input (W)q loss (W)
Baseline20.744019.26
PCRWPI125.154014.85
PCRWPI227.554012.45

PCRWPI327.614012.39 PCRWPS126.434013.57 PCRWPS226.434013.57 PCRWPS327.944012.06 PRWPI124.094015.91 PRWPI227.254012.75 PRWPI328.824011.18 PRWPS123.944016.06 PRWPS226.374013.63 PRWPS328.124011.88

Table 2 shows that the lowest heat loss occurs when three sets of PCRWPs are staggered. The placement of the VGs can increase heat transfer in square ducts as the VGs create longitudinal vortices that increase vortex strength in the wake region downstream of the tube. Longitudinal vortices make the overall temperature field more uniform, improve heat mixing and boundary layer modification, and improve heat transfer performance. A higher number of vortex generators creates more longitudinal vortices and greatly increases heat transfer [5], [3].

Mention pitch kept between two pins/winglet.

Thank you for the question. I've added Figure 3 (in the paper) showing the pitch between VGs for both inline and staggered configurations.

CRWP in-lineCRWP staggered

RWP in-lineRWP staggered

CRWP in-lineCRWP staggered

RWP in-lineRWP staggered

At lower Re both inline and staggered arrangement gives the same result, while deviation is observed after Re 8000. The author has to justify.

Thank you for your suggestion. In Figure 7(a) (in the paper), the convection heat transfer coefficient for the case of PRW VGs in-line has the same value as that of PCRW VGs in-line or staggered in a pair of VGs. In one pair of VGs, the longitudinal vortex is generated after the flow hits the VGs and weakens (He et al., 2013). This is in contrast to two and three pairs of VGs where the longitudinal vortex produced after striking the first VGs has amplified again when the flow strikes the second VGs and so on. Therefore, the value of the heat transfer coefficient in the case of a pair of PRW VGs has the same value as that of PCRW VGs at Reynolds numbers above 8,000.

Provide more clarification about 1,2 & 3 pairs.

Thank you for your suggestion. This study describes the cases of PCRW and PRW VGs for one, two, and three pairs. I have added explanations for cases one, two, and three pairs of perforated VGs to the paper.

For PRW and PCRW VGs in-line configurations with one, two, and three pairs are shown in Figures 4. For one pair, the VGs are placed on the left and right sides of the first row of tubes. VGs are placed on the first and third row tubes for two pairs. As for the three pairs, VGs are placed on the first, third, and fifth row tubes.

one pair PCRW inllinetwo pairs PCRW inline

three pairs PCRW inline

one pair PRW inlinetwo pairs PRW inline

three pairs PRW inline Figure 4. VGs pairs in-line configurations

For PRW and PCRW VGs staggered configurations with one, two, and three pairs are shown in Figures 5. For one pair, the VGs are placed on the right side of the first row tube and the left side of the second row tube. VGs are placed on the right side of the first, third row tubes and on the left side of the second and fourth tubes for two pairs. As for the three pairs, the VGs are placed on the right side of the first, third, fifth row of tubes and on the left side of the second, fourth and sixth tubes.

one pair PCRW staggeredtwo pairs PCRW staggered

three pairs PCRW staggered

one pair PRW staggeredtwo pairs PRW staggered

three pairs PRW staggered

Figure 5. VGs pairs staggered configurations

Compare the PRWP and PCRWP with without perforation.

Thank you for your request. In this study, an experiment was conducted to compare PRW VGs and PCRW VGs in improving heat transfer in a rectangular channel, as shown in Figures 7 to 9.

Comparison of convection heat transfer coefficient values.

Figure 7 provides a comparison of the convection heat transfer coefficients for PRW and PCRW. It can be seen that there was an increase in the convection heat transfer coeeficient with a rise in Re due to an increase in flow vortices and high turbulence intensity in the channel [6], alongside a reduction in the wage region and stagnation area for each increase in flow velocity [7]. The increase in heat transfer for staggered VGs was better than in-line for PCRW VGs with any number of pairs at the highest Re (11,000). The results in Figure 7 show that the PCRWP VGs worked better than the PRWP VGs, and the staggered arrangement of the former with three pairs gave the highest yield, of 153.5 W/m2.K, as shown in Figure 7(c). Meanwhile, two PCRW pairs (137.33 W/m2.K, Figure 7(b)) were better than one (132.25 W/m2.K, Figure 7(a)) because VGs with a concave surface destabilise the centrifugal force of the fluid flow, which strengthens the flow vortices. This makes the mixing of the hot fluid near the wall with the cold fluid of the main flow more robust [8].

Pressure drop comparison

In general, the highest pressure drop was observed using PCRWP VGs with staggered configuration for all Reynolds numbers except for one pair of VGs, as shown in Figure 8. The highest pressure drop was found in PRWP VGs with in-line configuration at Reynolds numbers greater than 8,000. The pressure drop on the staggered VGs was found to be higher than that of the in-line due to the shorter distance between the VGs of the staggered configuration than that of the in-line [5].

TEF comparison

TEF is the thermal-hydraulic performance which is the ratio of the increase in heat transfer to the pressure drop ratio. In general, the highest TEF was observed in the use of PCRWP VGs with staggered configuration, as depicted in Figure 9. PCRW creates wider flow vortices that can reduce the wake area behind the cylinder. Reducing the wake area can reduce the recirculation zone. This affects the increased heat transfer from the back of the cylinder to the stream [3]. A large radius, high intensity antrior-posterior vortex can reduce the wake area. A reduction in the wake area increases the flow velocity behind the tube and reduces the recirculation area, resulting in an increased heat transfer in this area [9], [10].

CBR comparison

A low value of CBR means a more economical value from the use of VGs. In general, CBR on the use of PCRWP VGs with staggered configuration is the best, as informed in Figure 10. The lowest CBR value results were obtained with three pairs of staggered type VGs PCRW. Three pairs of VGs lower CBR than one and two VG pairs. This is because the installation of three pairs VGs results in a higher Nusselt number increase than one and two pairs VGs, resulting in a lower pressure drop increase and therefore a lower CBR. These results show that lower CBR improves thermal performance relative to resistivity [11].

Discuss how number of pairs contribute in improving TEF

Thank you for your question. Figure 9 shows the effect of the number of pairs and configuration of VGs on TEF, this is also found in Ref. (He et al., 2013), (Sun et al., 2020), and (Ranjan et al., 2022).

One pair

Two pairs

Three pairs

From the experimental results, as shown in Figure 9, the TEF with three pairs of VGs for both inline and staggered was the highest. The TEF for the case of 3 pairs of PCRWs was 5.02% greater than that of one and two pairs of PRWP VGs. The main reason is because the concave surface of the PCR causes the flow to be thrown away due to the centrifugal force which results in a stronger flow vorticity [12]. Larger and stronger flow vortices can reduce the recirculation zone which has an impact on increasing heat transfer from the rear surface of the tube to the flow [10]. The presence of flow that is formed in each gap between the VGA and the tube causes the TEF in the staggered configuration to be greater than that of the in-line [13]. The increase in TEF for the three-pair case with the staggered configuration was 1.50% and 4.91% greater than that of the inline PCRWP and PRWP, respectively.

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Reviewer 2: This paper presented experimental results of the heat transfer, pressure drop and thermal performance characteristics of the perforated concave rectangular winglet pair vortex generators on plates in rectangular ducts to increase the heat transfer through the six heated tubes to the air stream. Perforated concave rectangular winglets were compared with perforated rectangular winglet pairs vortex generator mounted on rectangular plates. The subject of the article falls within the scope of the Results in Engineering. In my view, unless the paper is rewritten in a proper way, I think it is inadequate to be published in a scientific journal in its present state. I would like to provide the following comment:

There are many spelling and grammar mistakes in this paper, and many sentences are not easy to follow. The grammar and structure of sentences in this paper need to be modified carefully, such as the title of the article. The language of this paper need to be improved by a native English speaker.

Thank you for your suggestion. To improve grammar and structure, I have sent this paper to elsevier service for proof reading.

The velocity ranges studied are given in the Abstract, whereas the Reynolds number ranges studied should be given.

Thank you for your correction. In this experiment, the intake air velocity is in the range of 0.4 to 2 m/s at intervals of 0.2 m/s or in the Reynolds number range of 2.143 to 11.763. I have included the Reynolds number range in the abstract.

Please review the keywords and add a few, for instance heat transfer, thermal performance.

Thank you for your suggestion. The keywords for this study are vortex generators, heat transfer, thermal-hydraulic performance, economic benefit. I've added keywords in the paper.

The method section can be expanded.

Thank you for your suggestion. I have improved this research method in the paper. Based on Fig. 1, the rectangular channel is equipped with a blower (50 Hz, Wipro with a rated voltage of 220V), an inverter (Mitsubishi Electric type FR-D700 with an accuracy of 0.01), straightener, hot wire anemometer (Lutron type AM-4204 with an accuracy of 0.1), wattmeter (Lutron DW-6060 with an accuracy \pm 1.0), central processing unit (CPU), micromanometer, thermocouple (K type with a temperature interval of -200 –1250°C and an accuracy \pm 0.5) where one thermocouple was placed in the air inlet area, six thermocouples on the back surface of the tubes and 15 on the outlet side of the wire, data acquisition (Advantech USB-4718 type with an accuracy of 0.001) and heater regulator. The heater was connected to six tubes with a diameter of

19.05 mm and height of 65.8 mm, with each tube having the same power. Total heating power of 40 W was applied to the six tubes using a regulator. The heating air flowing through the tubes occurs via convection. Thus, the air at the outlet side becomes hotter than that at the inlet side.

A pressure micromanometer (Fluke type 922, with an accuracy of \pm 0.05) was used to monitor the flow pressure drop. Two pitot tubes, each set 26 cm ahead of the inlet of the test specimen and 2.5 cm behind it, were connected to a micromanometer to measure the pressure drop. The pressure drop measurements were recorded 30 times for 5 sekon at each speed variation. Furthermore, flow visualisation was performed by directing the smoke from vaporised fluid in the fluid vaporator into the mainflow.

The arrangement of vortex generators is given in Figure 2, but Figure 2 is not sufficient for a clear understanding of the construction of vortex generators. Additional figures should be drawn which clearly show the construction of vortex generators and the inline and staggered arrangement.

Thank you for your suggestion. The following shows the construction of vortex generators inline and staggered (figure 2, 4 and 5 in the paper).

Figure 2. Geometry of the VGs

For PRW and PCRW VGs in-line configurations with one, two, and three pairs are shown in Figures 4. For one pair, the VGs are placed on the left and right sides of the first row of tubes. VGs are placed on the first and third row tubes for two pairs. As for the three pairs, VGs are placed on the first, third, and fifth row tubes.

one pair PCRW inllinetwo pairs PCRW inline

three pairs PCRW inline

one pair PRW inlinetwo pairs PRW inline

three pairs PRW inline

Figure 4. VGs pairs in-line configurations

For PRW and PCRW VGs staggered configurations with one, two, and three pairs are shown in Figures 5. For one pair, the VGs are placed on the right side of the first row tube and the left side of the second row tube. VGs are placed on the right side of the first, third row tubes and on the left side of the second and fourth tubes for two pairs. As for the three pairs, the VGs are placed on the right side of the first, third, fifth row of tubes and on the left side of the second, fourth and sixth tubes.

one pair PCRW staggeredtwo pairs PCRW staggered

three pairs PCRW staggered

one pair PRW staggeredtwo pairs PRW staggered

three pairs PRW staggered

Figure 5. VGs pairs staggered configurations

Thermal characteristics are given in terms of the convective heat transfer coefficient (h), friction characteristics are given in terms of pressure drop. Why are thermal characteristics not given in terms of Nusselt number (Nu) and friction characteristics in terms of coefficient of friction (f)?

Thank you for your correction. h versus Re and P vs Re are used instead of Nu - Re and f - Re in this experiment because, in the TEF calculation, the value of Nu represents the value of h resulting from the equation in formula 3 (in the paper), thus

 $h=(N_u k)/D_h (3)$

Figure 7 (in the paper) that h rises as Re rises shows that the Nu increases as Re rises, where h rises as Re rises [1]. While f in this experiment represents the ΔP as shown in the following formula

 $f=(2 \Delta P D_h)/(\rho V^2 (L+6D))(8)$

Formula 8 states that the friction factor (f) in the flow rate is determined by using the pressure drop (ΔP) characteristic where increasing the Reynolds number in figure 8 will decrease the friction factor [2].

The following are examples of several studies that use h as a representative of Nu and (ΔP) as a representative of f.

The experimental results of Yafid et al indicate that perforated VGs can increase the heat transfer rate and decrease the pressure drop using the parameters h, ΔP , and TEF as shown in the following graph[3].

Al Asadi et al represents heat transfer coefficient and pressure drop to show the results of their investigation that the addition of span wise gap variations can increase heat transfer performance and reduce pressure drop [4], as shown in the figure below.

Increased heat transfer in heat sinks, Zhang et al described the micro gap by pairing more VGs which resulted in a larger heat transfer coefficient and a reduced pressure drop value [5], as shown in the figure below.

Hosseinirad et al showed that the increase in heat transfer coefficient and pressure drop vs. Reynold number had a tendency to increase with the increase in Re to evaluate heat transfer [6]. There is an increase in heat transfer with an increase in Re which is indicated by an increase in the heat transfer coefficient and an increase in pressure drop along with an increase in Re.

It is not specified how the hydraulic diameter (Dh) is calculated. How the Reynolds number (Re) was calculated is not specified.

Thank you for your corrections. The calculation of the hydraulic diameter in this experiment uses a rectangular air duct with a side length of a = 0.165 m and a side width of b = 0.064 m with the resulting Dh of 0.0106 from the following formula. D h=(4A c)/p=4ab/2(a+b)=2ab/(a+b) (5)

The result of Dh is used to calculate Re with the formula Re=(pu in D h)/u(7)

With u_in of 0.4 - 2 m/s with an interval of 0.2 m/s, on the physical properties of air at a pressure of 1 atm and is the viscosity of the fluid so that Re used in this experiment ranges from 2,143 - 11,763.

No correlation (Nu-Re), (f-Re) is given.

Thank you for the correction. In this experiment, Nu - Re and f - Re are not shown but use h vs Re and ΔP vs Re because the Nu value represents the value of h that arises in this experiment as in equation 4 (in the manuscript)

 $h=(N_u k)/D_h (3)$

While f in this experiment represents the ΔP as shown in the following formula

 $f=(2 \Delta P D_h)/(\rho V^2 (L+6D))(8)$

The formula states that the friction factor (f) in the flow rate is determined by using the pressure drop (ΔP) characteristic where increasing the Reynolds number in figure 8 will

Error analysis is given, but uncertainty analysis is not done. Thank you for the correction. In the following, uncertainty analysis calculation data will be shown for the temperature at base line conditions with a velocity of 0.4 m/s as shown in Table 3 below. Table 3 Base-line test temperature data at a speed of 0.4 m/s T (Tube1)T (Tube2)T (Tube3)T (Tube4)T (Tube5)T (Tube6) 49.1909351.2136848.3231349.7691547.8021951.27142 49.183451.1772848.315649.7905347.765751.2639 49.1454551.1682648.3065549.752647.785651.25489 49.1210551.1727748.2821449.7282147.7611851.2594 49.1529751.2046548.2851549.7312247.7352451.2624 49.0996651.1514148.2896749.7357347.7687151.26691 49.0981551.1499148.2302949.7342347.7382651.29428 49.0891251.1408948.2501949.6673947.7292251.22751 From these data, it is found that T_Tube can be calculated by the equation as T Tube=(T Tube1+T Tube2+T Tube3+T Tube4+T Tube5+T Tube6)/6=49.56°C (11)Then, the average standard deviation is obtained by the following formula. s tube= $\sqrt{((\sum (i=1)^N)^{(i=1)^N})^{(i=1)^N}}$ (T tubei-T tube)^2)/N(N-1))=0.029(12) Therefore, the average T_tube can be written as 49.5 ± 0.029°C. T_out calculation results obtained 32.95°C. The average standard deviation was calculated using the following equation: s Tout= $\sqrt{((\sum (i=1)^N)(T \text{ out})^2)/N(N-1)} = 0.051(13)$ Furthermore, the average value of T_out can be written as 32.95 ± 0.051°C. Using the same equation, the standard deviation of T_in was found to be 0.033. Thus, the average T in value was 29.75 ± 0.016°C. The value of g at a speed of 0.4 m/s was found to be 19.48 W. To determine of the standard deviation of q, the following equation was used: [RSS]] $q=\sqrt{([s(\Delta T)] \text{ out})\partial q/(\partial T \text{ out}))^2+([s(\Delta T)] \text{ in})\partial q/(\partial T \text{ in}))^2)(14)}$ $\partial q/(\partial T \text{ out }) = (\partial (m \cdot c p \cdot T \text{ out-} m \cdot c p \cdot T \text{ in)})/(\partial T \text{ out }) = m \cdot c \cdot p$ $\partial q/(\partial T_{in}) = (\partial (m \cdot c_p \cdot T_{out} - m \cdot c_p \cdot T_{in}))/(\partial T_{in}) = -(m \cdot c \cdot p)$ where $[s(\Delta T)]$ _out)=0.051°C and $[s(\Delta T)]$ _in)=0.033°C, ensuring, that [RSS] _q= ± 0.290 W. Therefore, the heat transfer rate g becomes 19.48 ± 0.290 W. The value of [[\Delta T]] _Imtd at a speed of 0.4 m/s was found to be 18.56°C. To determine the value of the standard deviation of $\llbracket \Delta T \rrbracket$ Imtd we used the following equation: $[RSS] _{-}([\Delta T] _{-} Imtd) = \sqrt{([s(\Delta T] _{-} tube) \partial ([\Delta T] _{-} Imtd)/(\partial T_{-} tube))^2 + ([s(\Delta T] _{-} tube) \partial ([\Delta T] _{-} Imtd)/(\partial T_{-} tube))^2 + ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube)))^2 + ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube)))^2 + ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube)))^2 + ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube)))^2 + ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube)))^2 + ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube)))^2 + ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube)))^2 + ([s(\Delta T] _{-} tube) \partial ([s(\Delta T] _{-} tube)))^2 + ([s(\Delta T] _{-} tube))^2 + ([s(\Delta T] _{$ $s(\Delta T) = out)\partial(\Delta T) = Imtd /(\partial T_out)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd /(\partial T_in)^2 + (s(\Delta T) = in)\partial(\Delta T) = Imtd$)(15) $\partial(\Delta T)$ Imtd $\partial(\Delta T)$ Im $T_out)/(T_tube-T_in)$)/(∂T_tube) $\partial([\Delta T]]$ Imtd)/(∂T out)= $\partial(((T \text{ tube-T out })-(T \text{ tube-T in }))/In [(T \text{ tube-T out })-(T \text{ tube-T in })]/In [(T \text{ tube-T out })-(T \text{ tube-T in })]/In [(T \text{ tube-T out })-(T \text{ tube-T in })]/In [(T \text{ tube-T out })-(T \text{ tube-T in })]/In [(T \text{ tube-T out })-(T \text{ tube-T in })]/In [(T \text{ tube-T out })-(T \text{ tube-T in })]/In [(T \text{ tube-T out })-(T \text{ tube-T in })]/In [(T \text{ tube-T out })-(T \text{ tube-T in })]/In [(T \text{ tube-T out })-(T \text{ tube-T in })]/In [(T \text{ tube-T out })-(T \text{ tube-T out }$ T_{out} /(T_{tube} - T_{in})])/(∂T_{out}) ∂ (ΔT Imtd)/(∂T_i n)= ∂ (((T_t ube- T_i)-(T_t ube- T_i n))/In T_t $T_{out}/(T_{tube}-T_{in})$ $)/(\partial T_{in})$ where $[s(\Delta T)]$ _tube)=0,029°C, $[s(\Delta T)]$ _out)=0,051°C and $[s(\Delta T)]$ _in)=0,033°C; we get $[RSS]_{(\Delta T)}$ _Imtd) of \pm 0.043, ensuring, that the obtained $[\Delta T]$ _Imtd is 8.56 ± 0.043 . The value of Nu at a speed of 0.4 m/s was found to be 155.31. The standard deviation of Nu was obtained using following equation $[RSS] _Nu=\sqrt{((s(q)\partial Nu/\partial Q)^2 + [s(\Delta T)] _lmtd)\partial Nu/(\partial [\Delta T]] _lmtd))(16)}$ $\partial Nu/\partial q = \partial (q \bullet D_h. \text{ [At] } ^{-1}) \bullet \text{ [[AT] } \underline{ \text{lmtd] } ^{-1}} \bullet k^{-1}) \cdot \partial q = D_h/((At)(\text{ [AT] } \underline{ \text{lmtd] } ^{-1}})$ Imtd)(k)) $\partial Nu/(\partial [\Delta T]]$ Imtd $)=\partial (g \cdot [D h \cdot At]] ^{(-1) \cdot [AT]}$ Imtd $) ^{(-1) \cdot k^{(-1)}} /(\partial [\Delta T]]$ $Imtd)=(g.D_h)/((At)([\Delta T] Imtd)^2 (k))$ With the values of s(q)=0.290 W and $[s(\Delta T)]$ _lmtd)=0.043, the obtained [RSS] _Nu was \pm 2.889 W/(m2°C). Therefore, the value of [RSS] Nu is 155.31 \pm 2.889 Wm2°C. The value of h at a speed of 0.4 m/s was found to be 44.86. To determine the standard

decrease the friction factor [2].

deviation of Nu the following equation is used

```
[RSS]] _h=\sqrt{((s(Nu)\partial h/Nu)^2)(17)}
\partial h/\partial Nu = \partial (h.D_h.k^{-1})/\partial h = k/D_h
Furthermore, the value of Dh is 0.092 m and k at Tf= 40.24 is 0.026. So the value of
h at a speed of 0.4 m/s is:
[RSS]] h=\sqrt{((s(Nu)\partial h/Nu)^2)} = 0.83
Thus, the number h at a speed of 0.4 \text{ m/s} is 44.86 \pm 0.83. So, the error h for the
baseline at a speed of 0.4 m/s is
Error= [RSS] _h/h×100(18)
Error=0.83/44.86×100=1.51%
From the test in the baseline case with a speed of 2.0 m/s, the results of the pressure
drop are listed in Table 4, which show that the average P can be calculated as follows:
(\Delta P)^{-} = ( [\Delta P] _1 + [\Delta P] _2 + [\Delta P] _3 + \dots + [\Delta P] _30)/30 = 3.51 Pa(19)
The average standard deviation of the pressure drop can then be calculated using the
equation
s=\sqrt{((\sum_{i=1}^{N})^{N})([\Delta P]_{i-(\Delta P)})^{2}}/N(N-1)=8.9\times10^{(-5)}(20)
Baseline case for the pressure drop value at a speed of 2.0 m/s is 3.51±8.9×10<sup>(-5)</sup>
Pa. Then, the error in the form of percentage can be calculated using the following
equation:
(8.9×10<sup>(-5)</sup>)/3.51×100=0.71
Table 4 Baseline pressure drop data at a speed of 2.0 m/s
Data to 2.0 m/s Data to 2.0 m/s
 10.013160.012
  20.013170.013
30.013180.012
40.013190.012
50.012200.013
60.013210.013
70.013220.012
80.012230.013
90.013240.012
100.013250.013
110.013260.013
120.013270.013
130.012280.013
140.012290.012
150.013300.012
The equal calculation approach changed into used for all data. Therefore, the overall
error outputs for the pressure-drop vortex generator with placement variations (in-line
and staggered), Re and amount of VG sets (one, two and three) are listed in Table 5.
Table 5. Overall Pressure Drop (\Delta P)
Vortex Generator VariationsOverall Error P
(perforated)
1 PRWP in-line2.94%
2 PRWP in-line2.87%
3 PRWP in-line1.98%
1 PRWP staggered2.88%
2 PRWP staggered2.34%
3 PRWP staggered1.36%
1 PCRWP in-line2.72%
2 PCRWP in-line1.80%
3 PCRWP in-line1.80%
1PCRWPstaggered2.43%
2PCRWPstaggered1.91%
3 PCRWP staggered0.97%
The average TEF results from the experimental results can be calculated as follows.
(TEF) = ( [TEF] 1+ [TEF] 2+ [TEF] 3+···+ [TEF] 12)/12=1.12 (21)
Then, the average standard deviation of the TEF can be calculated with the equation
s=\sqrt{((\sum_{i=1}^{N})^{N})([TEF])_{i-(TEF)})^{2}}/N(N-1)=1.07(22)
Therefore, the TEF value was 1.12 ± 1.07. Then, the error in the form of percentage
can be calculated using the following equation:
1.07/1.12×100=0.94%
```

The overall error results for the TEF vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are are listed inTable 6.

Table 6. Overall error TEF

Variasi Vortex GeneratorOverall Error TEF

(Berlubang)

- 1 RWP in-line
- 0.47 %
- 2 RWP in-line0.47%
- 3 RWP in-line0.43%
- 1 RWP staggered0.47%
- 2 RWP staggered0.47%
- 3 RWP staggered0.43%
- 1 CRWP in-line0.45%
- 2 CRWP in-line0.45%
- 3 CRWP in-line0.42%
- 1 CRWP staggered0.45% 2 CRWP staggered0.45%

3 CRWP staggered0.41%

First, find the average CBR of the experimental results with the following formula. (CBR) = ([CBR] _1+ [CBR] _2+ [CBR] _3+···+ [CBR] _12)/12=2.14 (23)

The average standard deviation of the pressure drop CBR can then be calculated using the following equation:

 $s=\sqrt{((\sum_{i=1}^{N})^{N})((CBR))^{-i}-(CBR)^{-i})^{2}}/N(N-1)=1.60(24)$

The CBR value is 2.14 ± 1.60. Then the error in the form of percentage can be calculated using thefollowing equation:

1.60/2.14×100=0.63%

The overall error results for the CBR vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are listed in Table 7 Table 7. Overall error CBR

Variasi Vortex GeneratorOverall Error CBR

(Berlubang)

- 1 RWP in-line 0.32%
- 2 RWP in-line0.29%
- 3 RWP in-line0.45%
- 1 RWP staggered0.32%
- 2 RWP staggered0.31%
- 3 RWP staggered0.45%
- 1 CRWP in-line0.4%
- 2 CRWP in-line0.42%
- 3 CRWP in-line0.56%
- 1 CRWP staggered0.43%
- 2 CRWP staggered0.42%
- 3 CRWP staggered0.66%

Reference

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[2]A. J. Modi and M. K. Rathod, "Comparative study of heat transfer enhancement and pressure drop for fin-and-circular tube compact heat exchangers with sinusoidal wavy and elliptical curved rectangular winglet vortex generator," Int J Heat Mass Transf, vol. 141, pp. 310-326, Oct. 2019, doi: 10.1016/J.IJHEATMASSTRANSFER.2019.06.088. [3]Y. Effendi, A. Prayogo, Syaiful, M. Djaeni, and E. Yohana, "Effect of perforated concave delta winglet vortex generators on heat transfer and flow resistance through the heated tubes in the channel," Experimental Heat Transfer, vol. 35, no. 5, pp. 553-576, 2022, doi: 10.1080/08916152.2021.1919245.

[4]M. T. Al-Asadi, F. S. Alkasmoul, and M. C. T. Wilson, "Benefits of spanwise gaps in cylindrical vortex generators for conjugate heat transfer enhancement in microchannels," Appl Therm Eng, vol. 130, pp. 571-586, Feb. 2018, doi: 10.1016/J.APPLTHERMALENG.2017.10.157.

[5]J. F. Zhang, Y. K. Joshi, and W. Q. Tao, "Single phase laminar flow and heat

transfer characteristics of microgaps with longitudinal vortex generator array," Int J Heat Mass Transf, vol. 111, pp. 484–494, Aug. 2017, doi: 10.1016/J.IJHEATMASSTRANSFER.2017.03.036.

[6]E. Hosseinirad, M. Khoshvaght-Aliabadi, and F. Hormozi, "Evaluation of heat transfer and pressure drop in a mini-channel using transverse rectangular vortex-generators with various non-uniform heights," Appl Therm Eng, vol. 161, p. 114196, Oct. 2019, doi: 10.1016/J.APPLTHERMALENG.2019.114196.

Reviewer 3: In connection with climate change and an increase in the average annual temperature on Earth, there is a new danger of the negative impact of high temperatures on human life.

In this case, the improvement of air conditioning systems, including the search for the best thermal enhancement factor, cost-benefit ratio, etc., takes on a new sense, which is one of the main targets of this article. The topic is timely and of great practical significance to environmental protection, enhancing safety, and people's life comfort. The manuscript is well-structured and includes all necessary parts.

Two key strengths of the paper are a good introduction section and an analysis and discussion of the results. Both research objectives and content are clear. The key scientific issues to be solved are moderate. The research experimental method is reasonable.

There are also several shortages worthy to be mentioned:

Seriously revise the formulas

If you use an italic font in formulas, use an italic font in their descriptions. For example, in formula (1), the Nusselt number (Nu) and friction factor (f); in formula (3), heat transfer coefficient (h). It may confuse the reader.

Thank you for the corrections. For formula and description fonts, improvements have been made where all formula and description fonts are italicized consistently.

The Nusselt numbers in formula (1) and formula (2) have different designations. It may confuse the reader.

Thanks for the correction. Consistent improvements have been made to writing the Nu symbol on paper.

What are Nusselt number and friction factor with subscript 0 in formula (1)?

Thanks for the corrections. The subscript 0 for Nusselt number and friction factor is meant for the baseline condition. This additional explanation has been added to the paper. The following is an explanation of formula 1.

TEF= $((Nu/ [Nu] _0))/(f/f_0)^{(1/3)}(1)$

Di mana: [Nu] _0= Nusselt number pada kondisi baseline f_0 = friction factor pada kondisi baseline

In formula (5), a pressure drop is the lowercase letter Δp , but in formula (6), a pressure drop is uppercase ΔP . Are these different pressures?

Thank you for the corrections. I'm so sorry for the error in writing the pressure drop symbol which is inconsistent. Improvements in writing pressure drop have been made with uppercase P for the formula on the paper.

What is the error of pressure drop measurement with the Fluke 922 Airflow Micromanometer described in section "3.2 Effect of perforated vortex generators on pressure drop"? Did it cover the necessary range of pressures to be investigated? Could micromanometer error have affected the conclusions of the section? Because the pressure drop values of 4.58 Pa, 5 Pa and 5.4 Pa are close to each other.

Thanks for the question. Pressure measurement errors with the Fluke 922 Airflow Micromanometer are explained in the uncertainty analysis section with the calculation results as below.

From the test in the baseline case with a speed of 2.0 m/s, the results of the pressure drop are listed in Table 4, which show that the average P can be calculated as follows: $(\triangle P) = (\triangle P) _ 1 + (\triangle P) _ 2 + (\triangle P) _ 3 + \cdots + (\triangle P) _ 30)/30 = 3.51 \ Pa(19)$

The average standard deviation of the pressure drop can then be calculated using the equation

 $s=\sqrt{((\sum_{i=1}^{N})^{N})([\Delta P]_{i-(\Delta P)})^{2}}/N(N-1)=8.9\times10^{(-5)}(20)$

Baseline case for the pressure drop value at a speed of 2.0 m/s is 3.51±8.9×10^(-5) Pa. Then, the error in the form of percentage can be calculated using the following equation:

(8.9×10⁽⁻⁵⁾)/3.51×100=0.71

Table 4 Baseline pressure drop data at a speed of 2.0 m/s

∆P (Pa)

Data to 2.0 m/sData to 2.0 m/s

10.013160.012

20.013170.013

30.013180.012

40.013190.012

50.012200.013

60.013210.013

70.013220.012

80.012230.013

90.013240.012

100.013250.013

110.013260.013

120.013270.013

130.012280.013

140.012290.012

150.013300.012

The equal calculation approach changed into used for all data. Therefore, the overall error outputs for the pressure-drop vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are listed in Table 5. Table 5. Overall Pressure Drop (\triangle P)

Vortex Generator VariationsOverall Error P (perforated)

1 PRWP in-line2.94%

2 PRWP in-line2.87%

3 PRWP in-line1.98%

1 PRWP staggered2.88%

2 PRWP staggered2.34%

3 PRWP staggered1.36%

1 PCRWP in-line2.72%

2 PCRWP in-line1.80%

3 PCRWP in-line1.80%

1 PCRWP staggered2.43%

2 PCRWP staggered1.91%

```
3 PCRWP staggered0.97%
```

The results of the measurement error calculation are still below the maximum accuracy limit of the tool by 5% so that it does not affect the conclusion section.

The measurement error in the section "3.3 Effect of perforated VGs on thermal enhancement factor" and "Effects of perforated VGs on the cost-benefit ratio" is not clear. Can you show the error bar or describe it in the description?

Thanks for the question. The measurement error for the thermal increase factor and the cost benefit ratio has been added to the explanation in the paper in the data uncertainty section, as follows.

error bar TEF

The average TEF results from the experimental results can be calculated as follows. (TEF) =([TEF] _1+ [TEF] _2+ [TEF] _3+···+ [TEF] _12)/12=1.12 (21)

Then, the average standard deviation of the TEF can be calculated with the equation $s=\sqrt{((\sum_{i=1}^{N})^{N})([TEF]^{-})^{2}}/N(N-1))=1.07(22)$

Therefore, the TEF value was 1.12 ± 1.07. Then, the error in the form of percentage can be calculated using the following equation:

1.07/1.12×100=0.94%

The overall error results for the TEF vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are are listed inTable 6.

Table 6. Overall error TEF

Variasi Vortex GeneratorOverall Error TEF

(Berlubang)

1 RWP in-line

0.47 %

2 RWP in-line0.47%

3 RWP in-line0.43%

1 RWP staggered0.47%

2 RWP staggered0.47%

3 RWP staggered0.43%

1 CRWP in-line0.45%

2 CRWP in-line0.45%

3 CRWP in-line0.42%

1 CRWP staggered0.45%

2 CRWP staggered0.45%

3 CRWP staggered0.41%

b error bar CBR

First, find the average CBR of the experimental results with the following formula. (CBR) = ([CBR]] _1+ [CBR]] _2+ [CBR]] _3+···+ [CBR]] _12)/12=2.14 (23)

The average standard deviation of the pressure drop CBR can then be calculated using the following equation:

 $s=\sqrt{((\sum_{i=1}^{N})^{N})([CBR]]_{i-(CBR)^{-i}}^2)/N(N-1)}=1.60(24)$

The CBR value is 2.14 ± 1.60. Then the error in the form of percentage can be calculated using thefollowing equation:

1.60/2.14×100=0.63%

The overall error results for the CBR vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are listed in Table 7.

Table 7. Overall error CBR

Variasi Vortex GeneratorOverall Error CBR

(Berlubang)

1 RWP in-line 0.32%

2 RWP in-line0.29%

3 RWP in-line0.45%

1 RWP staggered0.32%

2 RWP staggered0.31%

3 RWP staggered0.45%

1 CRWP in-line0.4%

2 CRWP in-line0.42%

3 CRWP in-line0.56%

1 CRWP staggered0.43%

| | 2 CRWP staggered0.42% 3 CRWP staggered0.66% In my humble opinion, the section "3.5 Flow visualisation" is better presented first in section "3. Results and Discussion". Thanks for the suggestions. The discussion of section 3.5 on visualization has been moved to the earlier section to 3.1 in the paper. We tried our best to improve the manuscript and made some changes in the revised paper, and here we did not list the specific changes but marked in red in revised paper. We appreciate for Editors and Reviewrs' warm work earnestly, and hope that the correction will meet with approval. Once again, thank you very much for your comments and suggestions. |
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| | Yours Sincerely Oktarina Heriyani |
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CRediT author statement

Oktarina Heriyani: Conceptualization, Methodology, Investigation, Writing – Original Draft preparation.: **Mohammad Djaeni**: Supervision – Reviewing and Editing.: **Syaiful**: Conceptualization, Writing – Reviewing and Editing, Conceptualization.: *Aldila Kurnia Putri*: Visualization, Validation.

Declaration of Competing Interests

Declaration of interests

| ⊠The authors declare that they have no known competing financial interests or personal relationships |
|---|
| that could have appeared to influence the work reported in this paper. |
| |
| □The authors declare the following financial interests/personal relationships which may be considered |
| as potential competing interests: |

HIGHLIGHTS

- Perforated vortex generators installed to increase heat transfer and reduce pressure drop through six heated tubes to the air stream.
- Perforated concave rectangular winglets compared with perforated rectangular winglet pairs vortex generator mounted on rectangular plates were investigated experimentally.
- Perforated concave rectangular winglets improve thermal performance better than perforated rectangular winglet pairs vortex generators.
- The best thermal performance improvement is at the lowest cost benefit ratio.

Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel

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Abstract

A significant increase in the rate of heat transfer in a heat exchanger system is made possible by increasing the convection heat-transfer coefficient using a passive method. The addition of vortex generators (VGs) to the fins and tubes of the heat exchanger is currently the most effective passive method. However, the augment in heat is accompanied by an increase in pressure drop. Therefore, in this study, we installed perforated concave rectangular winglet pair vortex generators (PCRWP VGs) on plates in rectangular ducts to increase the heat transfer through the six heated tubes to the air stream with lowering the enhance in pressure drop. We sought to decide the best cost-benefit ratio (CBR) with the difference in fluid flow velocity from 0.4 to 2 m/s with 0.2 m/s intervals in the ducts. PCRWP VGs were composed using in-line and staggered configurations. The outcomes showed a lower CBR (3.56) for the in-line configuration than staggered. Moreover, the lowest CBR was accompanied by a thermal performance (TEF) increase of 1.29.

1. Introduction

The global energy demand is expected to triple in the next few years. The main driver is the increasing use of air conditioning (AC) machines, according to a statement by the International Energy Agency (IEA) [1]. Thus, promoting energy efficiency in air conditioners becomes important, which requires us to maximise their thermal performance. Improving the thermal performance of an air conditioner involves increasing the rate of heat transfer in its main component, the condenser. A condenser commonly used in air conditioners comprises a fin and tube, which function as a refrigerant cooling medium. However, the high thermal resistance (reaching 75%) of the fin air side of the condenser lowers the heat transfer rate in the heat exchanger [2]. That thermal resistance must be lowered to enhance the heat transfer rate.

One of the most commonly used active methods to increase the rate of heat transfer is adding vortex generators (VGs), which according to the research results of Mugisidi et al., increases the performance of a condenser [3]. The added VGs cause longitudinal vortices (LVs), damage the primary flow, make the second flow as large as the first, and increase the air mixing in the area [4,5]. The size of the LVs, shape of the flow, and mixing formed are influenced by the shape, geometry, and position of the VGs added to the fins and tubes of the heat exchanger [6].

Samidifat et al., in their research, showed that simple rectangular vortex generators (RVGs) can increase the heat transfer rate by 7%, although this causes a pressure drop in the heat exchanger system [7]. Meanwhile, modified RVGs with a concave shape on the front and rear surfaces decrease the heat transfer performance of the heat exchanger tube. A better option is RVGs with a double convex front surface and a single concave back surface, which can strengthen the primary vortex, thus increasing the rate of heat transfer from the plate to the fluid, as demonstrated in a study by Kashyap et al. [8]. Further research conducted by Kashyap et al. in the same year concluded that modifying the surface shape of RWVGs can create longitudinal eddies that interact with the boundary layer, increasing the rate of convection heat transfer [9]. The increase in the optimal heat transfer rate, based on their research, is 14.4. Beyond this, Song et al. attempted to compare changes in the rate of heat transfer in a heat

exchanger system paired with concave or convex curved, delta winglet VGs [10]. The results showed that the concave VGs could improve heat transfer better than the convex VGs. Yet, it is not only the difference in shape that affects the change in the heat transfer rate; changes in the geometry of VGs also play a role.

Research conducted by Zeeshan et al. showed that increasing the angle of attack increased the rate of heat transfer (to 37.01–64.54%) if a pair of rectangular winglet vortex generators (RWVGs) were placed at the back of the tube; yet, this did not reduce the pressure drop [11]. Beyond this, the number of pairs of RWVGs affected the increase in heat transfer, as discovered by Oheriyani et al., with a 15.17% better hydraulic thermal performance for three pairs of RWVGs compared to the baseline [12]. The more pairs of VGs placed in the crossflow, the higher the increase in the heat transfer rate, as found by Wang et al. [13]. A study by Sun et al. further discovered that increasing the number of RWVGs in the heat exchanger tube increased the heat transfer, with a maximum thermal enhancement factor (TEF) of 1.27 [14]. The TEF value of a V delta winglet VG, meanwhile, reached 1.82–3% higher than a V rectangular winglet VG, as revealed by Promvonge et al. [15]. These results were obtained with the optimal blockage ratio (BR)=0.15 and pitch ratio (PR)=1.0. Elsewhere, Skullong et al. modified the shapes of RWVGs with optimal BRs and PRs to achieve an optimum heat transfer performance and reduced pressure drop; their shape modification involved perforating RWVGs [16].

The positions of the holes in RWVGs do not significantly affect the increase in heat transfer, but it greatly affects the flow resistance of VGs. The heat transfer rate increases as the height (vertical position) of the hole increases. Width-wise, although there is an initial increase, the heat transfer rate decreases with increasing lateral distance [17]. An increase in the number of holes in RWVGs indicates an increase in fluid flow; this forces fluid to flow behind RWVGs, which increases the heat transfer [18]. The heat transfer rate increases during laminar flow when the Reynolds number (Re) increases, then decreases with an increase in Re during turbulent flow [19]. Positioning the tube in-line, with a pair of RWVGs in a common flow-down configuration, provides better performance than the common flow-up [20]. However, a staggered tube position is superior to this, resulting in a 25.85% higher heat-transfer performance than when a pair of RWVGs is not used [21].

In the existing studies, no detailed analyses of heat transfer were carried out from the surfaces of several cylinders heated and arranged in-line when using a perforated vortex generator. Therefore, this study focused on investigating the advantages of using perforated concave rectangular winglet pair vortex generators (PCRWP VGs) to increase the heat transfer of airflow through heated tubes arranged in-line in the ducts.

2. Experimental Approach

2.1 Experimental setup

This research was conducted experimentally with a test equipment scheme comprising a rectangular channel sized 370 x 18 x 8 cm. The duct was made of 1 cm-thick glass, as shown in Figure 1.

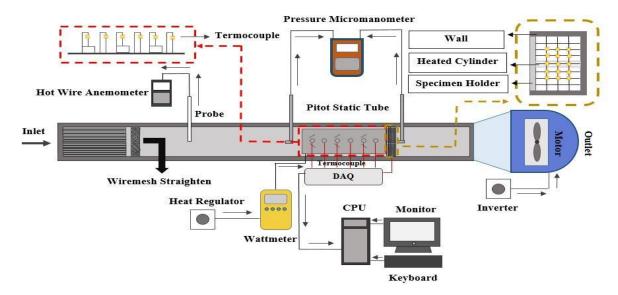
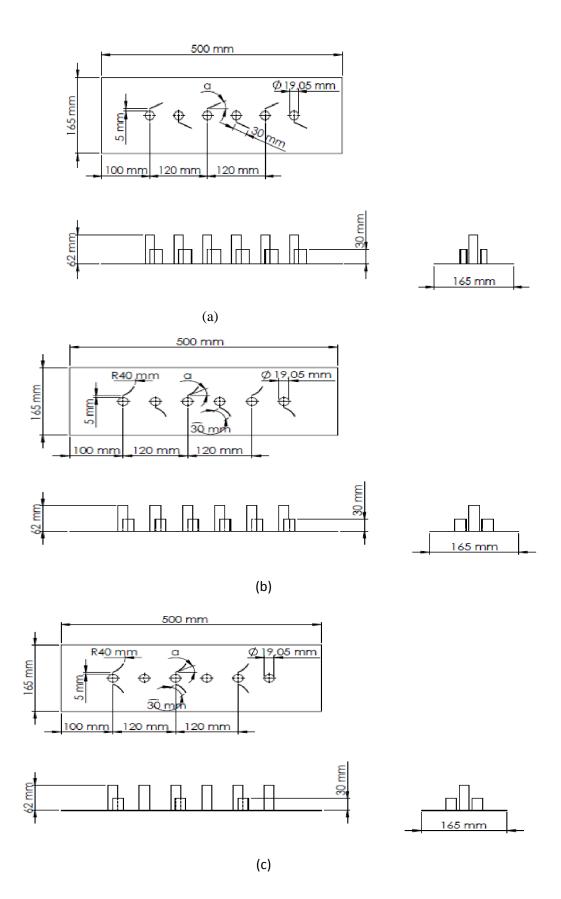


Figure 1 Experimental Tool Schematic

Based on Figure 1 is equipped with a blower (50 Hz, Wipro with a rated voltage of 220V), an inverter (Mitsubishi Electric type FR-D700 with an accuracy of 0.01), a straightener, a hot wire anemometer (Lutron type AM-4204 with an accuracy of 0.1), heater regulator, wattmeter (Lutron DW-6060 with an accuracy \pm 1.0), thermocouple (K type with a temperature interval of -200 –1250 °C and an accuracy \pm 0.5), data acquisition (Advantech USB-4718 type with an accuracy of 0.001), CPU, pressure micromanometer (Fluke type 922 with an accuracy of 0.05), and micromanometer.

The test in this experiment varied the inlet air velocity from 0.4 to 2 m/s with 0.2 m/s intervals. Air flowing at a constant heat rate of 40 W passed through six cylindrical tubes with a diameter of 19.05 mm and a height of 65.8 mm. The six cylindrical tubes were composed in-line with 60 mm between the centers of the cylinders.

The VGs used as test specimens were perforated rectangular winglet pair and perforated concave rectangular winglet pair types sized 30 x 30 mm. The VGs were be assembled on aluminium plates measuring 500 x 155 x 1 mm in one, two, or three pairs. Naik et al. verified the merits of using rectangular, winglet pair vortex generators with a CFD configuration to improve the heat transfer rate [22]. Thus, the VGs pairs were divided into in-line and staggered arrangements, with a common flow-down (CFD) configuration, as shown in Figure 2. We fixed our angle of attack at 15° since Lu and Zhai represented that reaps the best hydraulic thermal performance [23].



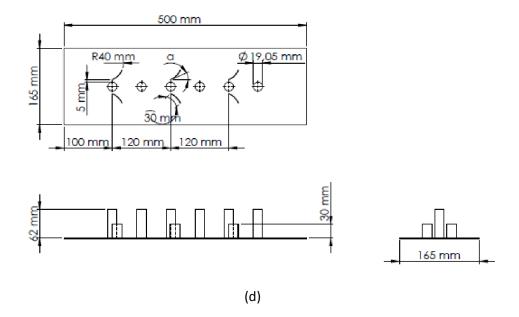


Figure 2 Arrangement of test specimens; rectangular winglet vortex generators with (a) staggered or (b) in-line arrangement; concave rectangular winglet vortex generators with (c) staggered or (d) in-line arrangement

2.2 Parameter definitions

The parameters in this study were derived from the equation used by Oneissi et al. to obtain the Thermal Enhancement Factor (TEF) [24], namely

$$TEF = \frac{\frac{Nu}{Nu_0}}{\left(\frac{f}{f_0}\right)^{\frac{1}{3}}} \tag{1}$$

The Nusselt number (Nu) and friction factor (f), based on the research of Zeeshan et al., were formulated via equations (2) and (3) [11]

$$N_u = \frac{Q}{A_t \Delta T_{IMTD}} \cdot \frac{D_h}{k} \tag{2}$$

where Q, A_t , ΔT_{LTMD} , D_h , and k are the convection heat transfer rate, surface area, log mean temperature difference, hydraulic diameter, and thermal conductivity, respectively.

$$f = \frac{2 \Delta P D_h}{\rho V^2 (L+6D)} \tag{3}$$

where ρ , V, and L are the density of the air, velocity of the inlet airflow, and length of the test specimen plate, while the convection heat-transfer coefficient (h) is counted by the equation

$$h = \frac{N_u \, k}{D_h} \tag{4}$$

The further equation required to determine the cost-benefit ratio (CBR), defined as the ratio of pressure drop per variation in Nu number, as formulated by Tian et al. [25], is as follows

$$CBR = \frac{\%\Delta p}{\%Nu} \tag{5}$$

This concept investigates whether or not the method used to enhance the heat transfer rate is economically efficient. In the hydrodynamic test, the pressure drop (ΔP) P is measured by the pressure difference on the sides of the *inlet* (P_{inlet}) and *outlet* (P_{outlet}) of the test specimen in the tested part, using equation (6):

$$\Delta P = P_{inlet} - P_{outlet} \tag{6}$$

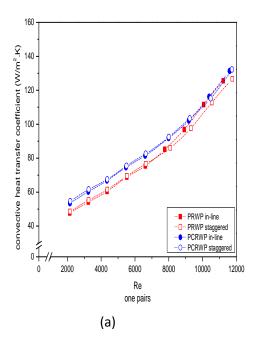
2.3 Validation

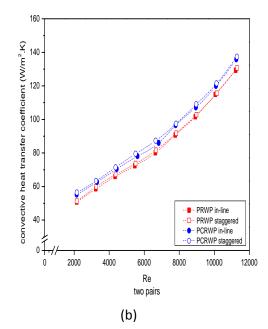
To ensure that the experimental procedure was sound, validation was done by comparing the current experiment with baseline conditions. Those were determined using the experimental results for the Nusselt number (Nu) from the works of Whitaker [26] and Syaiful et al. [27], which both showed the same trend, where Nu increased with an increasing Reynolds number (Re) from 2000 to 10,000.

3. Results and Discussion

3.1 Effect of perforated vortex generators on heat transfer

The increase in the convection heat transfer was due to the mixing of fluids caused by strong LVs [28]. The strength of the LVs is caused by the number of pairs of VGs. Increasing the number of pairs of VGs in the test specimen can increase the coefficient value of the convection heat transfer [29], as shown in Figure 3.





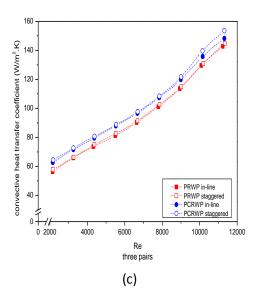


Figure 3 Graphs of convective heat-transfer coefficient against Reynolds number; (a) one, (b) two, (c) three pairs

In Figure 3, we can see the convective heat-transfer coefficient with respect to the Reynolds number (Re), analysed after installing PCRWP VGs or (non-concave) perforated, rectangular, winglet pair vortex generators (PRWP VGs), with one, two, and three pairs, arranged in-line or staggered. Based on Figure 3, it can be seen that there was an increase in the heat transfer with a rise in Re due to an increase in flow vortices and high turbulence intensity in the channel [30], alongside a reduction in the wage region and stagnation area [31] for each increase in flow velocity. The increase in heat transfer for staggered VGs was better than in-line for PCRW VGs with any number of pairs at the highest Re (11,000). The results in Figure 3 show that the PCRWP VGs worked better than the PRWP VGs, and the staggered arrangement of the former with three pairs gave the highest yield, of 153.5 W/m².K, as shown in Figure 3(c). Meanwhile, two PCRW pairs (137.33 W/m².K, Figure 3(b)) were better than one (132.25 W/m².K, Figure 3(a)) because VGs with a concave surface destabilise the centrifugal force of the fluid flow, which strengthens the flow vortices. This makes the mixing of the hot fluid near the wall with the cold fluid of the main flow more robust [32,33].

3.2 Effect of perforated vortex generators on pressure drop

The use of VGs can affect the increase in heat transfer, but there is often an accompanying increase in pressure drop, as shown in Figure 4 where an increase in pressure drop can be seen along with the increases in Re and pair numbers for both the VG types, PCRW and PRW. This was caused by the resistance to fluid flow against the walls of the VGs and the addition of the frontal area of VGs in the next pair arrangement [34]. The pressure drop in the staggered arrangement was lower than in the inline one, while the PRW VGs type created a lower pressure drop than the PCRW VGs due to the latter reducing the frontal area hit by the airflow, resulting in a decrease in drag [35]. In addition, jet flow from the VG hole can reduce the stagnation flow, which can, in turn, reduce the pressure drop [36]. A significant decrease noted in the pressure drop was due to the VGS perforation used [37]. The best pressure drop value for one pair with a staggered arrangement was 4.58 Pa (see Figure 4(a)), while two pairs (5 Pa, Figure 4(b)) were better than three (5.4 Pa, Figure 4(c)).

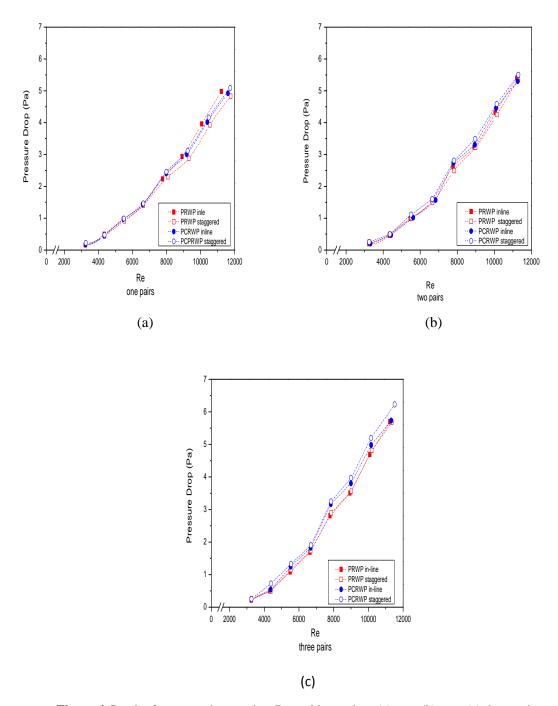


Figure 4 Graph of pressure drop against Reynolds number; (a) one, (b) two, (c) three pairs

3.3 Effect of perforated VGs on thermal enhancement factor

The TEF showed the hydraulic thermal performance while using VGs, which played a role in restructuring the incoming fluid flow pattern. The increase in TEF was due to the influence of complex overlapping structures, which meant the flow developed into a turbulent structure, greatly affecting the heat transfer increase [38]. The experimental TEF values are shown in Figure 5.

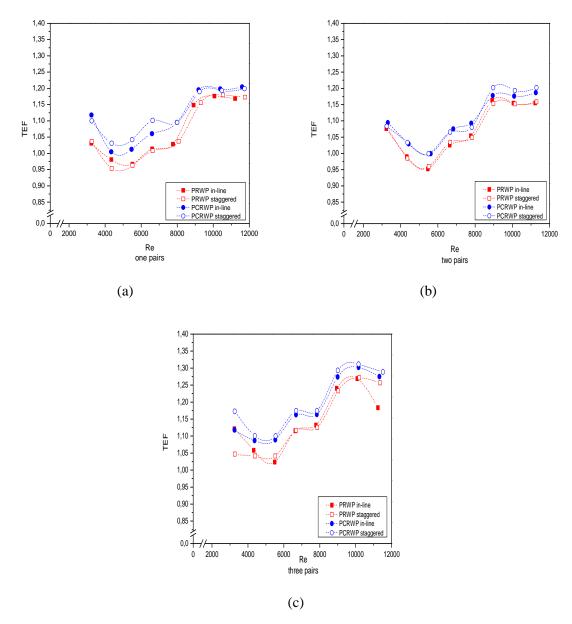


Figure 5 Graph of thermal enhancement factor against Reynolds number; (a) one, (b) two, (c) three pairs

It can be seen in Figure 5 that there was an increase in TEF with greater pairs of VGs used for both PCRW and PRW VGs. This happened because PCRW produced wider flow vortices, which reduced the wake region behind the cylinder, thereby reducing the recirculation zone, which impacted the heat transfer increase from the rear cylinder surface to the flow [39]. In this process, the large of longitudinal vortices with high intensities can reduce the wake area, which increases the flow velocity downstream of the tube and reduces the recirculation region, leading to an increased heat transfer rate in the region [40,41]. Based on Figure 5, the best TEF increase occurred at Re 8000–9000. The best TEF values with one, two, or three pairs occurred in the staggered arrangement with PCRW VGs, at 1.18, 1.20, and 1.29, respectively (see Figure 5).

3.4 Effects of perforated VGs on the cost-benefit ratio

Economic evaluation cannot be carried out based just on the TEF and the net profit from the transferred unit's heat load [42]. Instead, it must be determined by evaluating the economic value of the heat transfer improvement, by calculating the CBR, as shown in Figure 6.

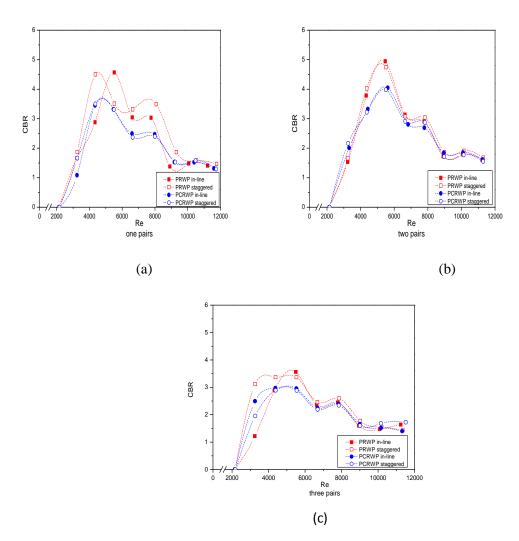


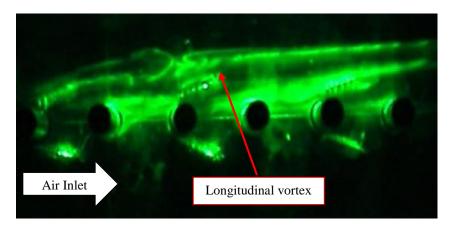
Figure 6 Graph of cost-benefit ratio against Reynolds number; (a) one, (b) two, (c) three pairs

Figure 6 is the result of the CBR calculation to compare the percentage raise in pressure drop with the percentage increase in the Nusselt number when using VGs. These results indicate that a lower CBR results in an improvement in thermal performance, which was higher than the drag force [25]. In Figure 6, the greatest increase in CBR values occurs with the PRW VGs, with an in-line arrangement, totalling 4.57, 4.95, or 3.56 for one, two, or three pairs, respectively. The lowest CBR value was measured when PCRW VGs, with a staggered arrangement and three pairs, were used. Three VG pairs showed a lower CBR than one and two pairs because they brought about the greatest increase in the Nusselt number, accompanied by a lower pressure drop increase, which lowered the CBR.

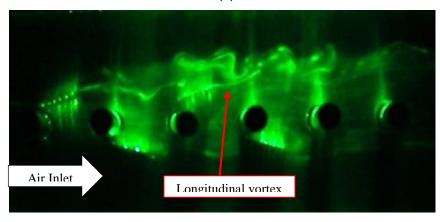
3.5 Flow visualisation

The flow visualisation test was carried out to observe the LV formed after the flow passed through the VGs in the rectangular channel. This test was conducted under low-light conditions to clarify the formed

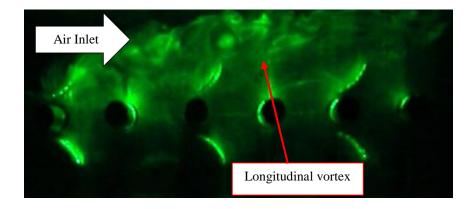
LV. The laser beam was refracted by a cylindrical glass (diameter 5 mm), which produced a cross-sectional area perpendicular to the direction of flow. The smoke formed from the evaporation of the liquid was used to visualize the LV in the flow. The VGs used in this visualisation test were PRWP and PCRWP with an in-line arrangement, as can be seen in Figure 7.



(a)



(b)



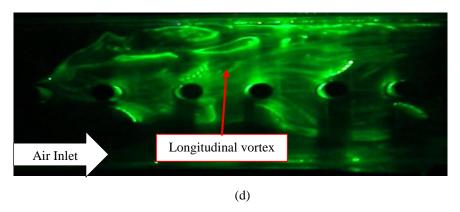


Figure 7 Visualisation of LV generated by (a) in-line (b) staggered PRWP (c) in-line (d) staggered PCRWP

In Figure 7 (c,d), the PCRWP VGs appear to produce LV in a wide flow area compared to the PRWP VGs in Figure 7 (a,b) downstream. The back region of the PCRWP VGs has a wider frontal surface area than the PRWP VGs. As a result, the mixing of near fluid the channel walls with the fluid in the mainstream is better, meaning the heat transfer rate is increased [32]. Downstream, the LV compression in the wake area increases the fluid flow velocity pass the cylindrical structure, thereby increasing the heat transfer rate from the channel surface to the fluid flow in the wake region [43]. The increase in heat transfer produced when using PCRWP VGs is better than with PRWP VGs.

3.6 Error analysis

The deviation (error) is the difference between the measured and actual values, which introduces uncertainty to a result. When using large amounts of data, scientific data-processing is necessary to determine the deviations (errors) that occurred during data collection, which may affect the results of analysis.

The deviation in the average pressure data was calculated using the following equation (7) [44, 45]

$$\overline{\Delta P} = \frac{\Delta P_1 + \Delta P_2 + \dots + \Delta P_{30}}{30} \tag{7}$$

The next step calculated the mean standard deviation using equation (8) [44, 45]

$$S_{\Delta p} = \sqrt{\frac{\sum_{i=1}^{n} (\Delta P_i - \overline{\Delta P})^2}{N(N-1)}}$$
 (8)

To calculate the overall error, equation (9) was used as follows [44, 45]

$$\%error_{\Delta P} = \left(\frac{S_{\Delta P}}{\overline{\Delta P}}\right) 100\% \tag{9}$$

The value of the overall error for VGs with holes was 0.97%, while the value of the overall error in the pressure drop for VGs with holes was 2.2%

Conclusion

Based on the experimental results for perforated concave rectangular winglet pair vortex generators (PCRWP VGs) use in increasing the heat transfer of airflow through heated tubes arranged in-line in

the duct, we conclude that using PCRWP VGs affects the convection heat transfer coefficient, pressure drop in achieving hydraulic thermal performance, and cost-benefit ratio. In our investigations, the best heat-transfer convection coefficient was 153.5 W/m².K for three pairs of PCRW VGs composed in a staggered manner. The greatest improvement in the pressure drop value (4.58 Pa), meanwhile, occurred for one pair of PCRW VGs arranged in a staggered manner, while the hydraulic thermal performance was best (1.29) in this experiment with three pairs of PCRW VGs composed in a staggered manner. Finally, the best CBR (3.56) was recorded again for three pairs of PCRW VGs composed in a staggered manner.

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Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel

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Abstract

A significant increase in the rate heat transfer in a heat exchanger system is made possible by increasing the convection heat-transfer coefficient using a passive method. The addition of vortex generators (VGs) to the fins and tubes of a heat exchanger is currently the most effective passive method. However, the increase in heat was accompanied by an increase in pressure drop. Therefore, in this study, we installed perforated concave rectangular winglet pair vortex generators (PCRWP VGs) on plates in rectangular ducts to increase the heat transfer through the six heated tubes to the air stream by lowering the enhancement in the pressure drop. We attempted to determine the best cost-benefit ratio (*CBR*) with a fluid flow velocity difference of 0.4 –2 m/s at intervals of 0.2 m/s (Reynolds number (*Re*) of 2,143 to 11,763) in the channel. The PCRWP VGs were composed of in-line and staggered configurations. The results showed a lower *CBR* (3.56) for the in-line configuration than for the staggered configuration. Moreover, the lowest *CBR* was accompanied by an increase in thermal performance (*TEF*) of 1.29.

Keywords: Perforated; Rectangular winglet; Concave; Pressure drop; Vortex generator; Heat transfer; Thermal performance

1. Introduction

The global energy demand is expected to triple over the next few years. According to a statement by the International Energy Agency (IEA), the main driver is the increasing use of air conditioning (AC) machines [1]. Thus, promoting energy efficiency in air conditioners is important and requires maximising their thermal performance, which involves increasing the rate of heat transfer in its main component, i.e., the condenser. A condenser, commonly used in air conditioners, comprises a fin and a tube and functions as a refrigerant cooling medium. However, the high thermal resistance (75%) of the fin air side of the condenser lowers the heat-transfer rate in the heat exchanger[2]. Thus, the thermal resistance must be lowered to enhance the heat transfer rate.

A commonly used active methods to increase the rate of heat transfer involves adding vortex generators (VGs), which, according to the research results obtained by Mugisidi et al., increases the performance of a condenser[3]. The added VGs cause longitudinal vortices (LVs), damage the primary flow, make the second flow as large as the first and increase air mixing in the area[4][5]. The size of the LVs, shape of the flow, and mixing are influenced by the shape, geometry and position of the VGs added to the fins and tubes of the heat exchanger[6].

Samidifat et al. showed that simple rectangular vortex generators (RVGs) can increase the heat transfer rate by 7%; however, this causes a pressure drop in the heat exchanger system[7]. Meanwhile, modified RVGs with a concave shape on the front and rear surfaces decreased the heat transfer performance of the heat exchanger tube. A better option is to use RVGs with a double convex front surface and a single concave back surface, which can strengthen the primary vortex, increasing the rate of heat transfer from the plate to the fluid, as demonstrated in a study by Kashyap et al.[8]. Further research conducted by Kashyap et al. in the same year concluded that modifying the surface shape of rectangular winglet vortex generators (RWVGs) can create longitudinal eddies that interact with the boundary layer, thereby increasing the rate of convection heat transfer[9]. Based on their research, the

increase in the optimal heat transfer rate was 14.4. The optimal heat transfer performance was also obtained from the results of experiments conducted by Adnan et al. on rectangular ducts by adding delta and rectangular winglet VGs[10]. Concave curved delta winglet VGs were compared with convex curved delta winglet VGs by Song et al. to observe changes in the heat transfer rate[11]. The results showed that the concave VGs improved the heat transfer better than the convex VGs. The differences in the shape of the VGs affects the change in the heat transfer rate and the change in the geometry of the VG, such as a new rib geometry in the cylinder channel[12].

Zeeshan et al. showed that increasing the angle of attack increased the rate of heat transfer (to 37.01-64.54%) if a pair of RWVGs were placed at the back of the tube even though this did not reduce the pressure drop[13]. A decrease in the value of the pressure drop also did not occur significantly, even though there was an increase in heat of 260% in heat, as per the results of the research conducted by Linardo et al. using the batched heat and channelled pipe (BHCP) approach[14]. The increase in heat transfer performance is influenced by the number of RWVG pairs based on the research results of Heriyani et al., where there is an increase in the hydraulic thermal performance evaluation criteria by 15.17% for three pairs of RWVG compared with the baseline [15]. Wang et al. found that the more pairs of VGs placed in the crossflow, the higher the increase in the heat transfer coefficient[16]. Sun et al. further discovered that increasing the number of RWVGs in the heat exchanger tube increased the heat transfer, with a maximum thermal enhancement factor (TEF) of 1.27 [17]. The TEF value of a V-delta winglet VG reached 1.82-3% higher than that of a V-rectangular winglet VG, as revealed by Promvonge et al.[18]. These results were obtained with an optimal blockage ratio (BR) of 0.15 and pitch ratio (PR) 1.0. Skullong et al. modified the shapes of RWVGs with optimal BRs and PRs to achieve an optimum heat transfer performance and reduced pressure drop; their shape modification involved perforating RWVGs [19].

The positions of the holes in the RWVGs did not significantly affect the increase in heat transfer; however, they significantly affected the flow resistance of the VGs. The heat-transfer rate increased as the height (vertical position) of the hole increased. Widthwise, although there is an initial increase, the heat transfer rate decreased with increasing lateral distance[20]. An increase in the number of holes in the RWVGs indicates an increase in fluid flow, which forces the fluid to flow behind the RWVGs, thereby increasing heat transfer[4]. The heat transfer rate increased during laminar flow when the Reynolds number (Re) increased and then decreased with an increase in Re during turbulent flow [20]. Positioning the tube in-line with a pair of RWVGs in a common flow-down configuration provides better performance than the common flow-up configuration. However, a staggered tube position is superior, resulting in a 25.85% higher heat-transfer performance than when a pair of RWVGs is not used[21].

In the existing studies, no detailed analyses of heat transfer were conducted on from the surfaces of several cylinders heated and arranged in-line when using a perforated vortex generator. Therefore, the focus herein is on investigating the advantages of using perforated concave rectangular winglet pair vortex generators (PCRWP VGs) to increase the heat transfer of the airflow through heated tubes arranged in-line in the ducts.

2. Experimental Approach

2.1 Experimental setup

This research was conducted experimentally with a test equipment scheme comprising a rectangular channel sized 370 x 18 x 8 cm. The duct was made of 1 cm thick glass, as shown in Fig. 1.

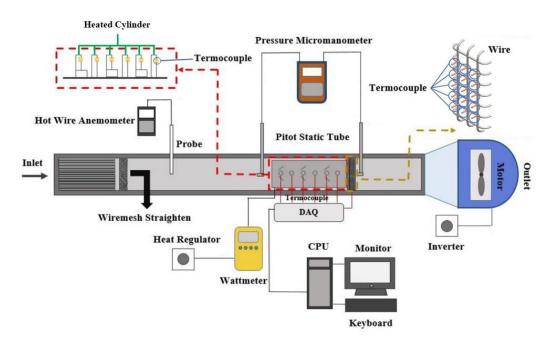


Fig. 1. Experimental tool schematic

Based on Fig. 1, the rectangular channel is equipped with a blower (50 Hz, Wipro with a rated voltage of 220V), an inverter (Mitsubishi Electric type FR-D700 with an accuracy of 0.01), straightener, hot wire anemometer (Lutron type AM-4204 with an accuracy of 0.1), wattmeter (Lutron DW-6060 with an accuracy \pm 1.0), central processing unit (CPU), micromanometer, thermocouple (K type with a temperature interval of -200 –1250°C and an accuracy \pm 0.5) where one thermocouple was placed in the air inlet area, six thermocouples on the back surface of the tubes and 15 on the outlet side of the wire, data acquisition (Advantech USB-4718 type with an accuracy of 0.001) and heater regulator. The heater was connected to six tubes with a diameter of 19.05 mm and height of 65.8 mm, with each tube having the same power. Total heating power of 40 W was applied to the six tubes using a regulator. The heating air flowing through the tubes occurs via convection. Thus, the air at the outlet side becomes hotter than that at the inlet side.

A pressure micromanometer (Fluke type 922, with an accuracy of \pm 0.05) was used to monitor the flow pressure drop. Two pitot tubes, each set 26 cm ahead of the inlet of the test specimen and 2.5 cm behind it, were connected to a micromanometer to measure the pressure drop. The pressure drop measurements were recorded 30 times for 5 sekon at each speed variation. Furthermore, flow visualisation was performed by directing the smoke from vaporised fluid in the fluid vaporator into the mainflow.

The VGs used as test specimens were perforated rectangular winglet pair (PRWP) and perforated concave rectangular winglet pair (PCRWP) vortex generators (VGs). Perforated is a term for holes in the VGs, as shown in Fig. 2. The VGs have dimensions of the same length and width of 30 mm, with

36 holes. The bore diameter on the VGs was 2.5 mm. The distance between the holes was 5 mm from the center.

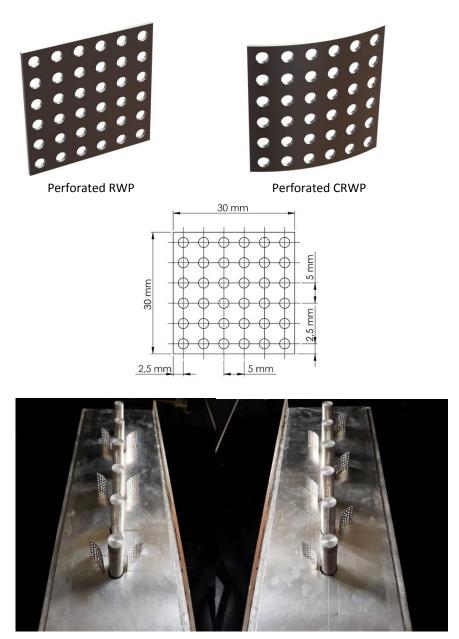


Fig. 2. Geometry of the VGs

The VGs are placed on an aluminium plate measuring $500 \times 165 \times 1$ mm. The geometry and the pitch between VGs for both in-line and staggered configurations are shown in Fig 3, with an angle of attack (α) of 150[2]. The distance between the cylinders is 120 mm, with a cylinder diameter of 19.05 mm.

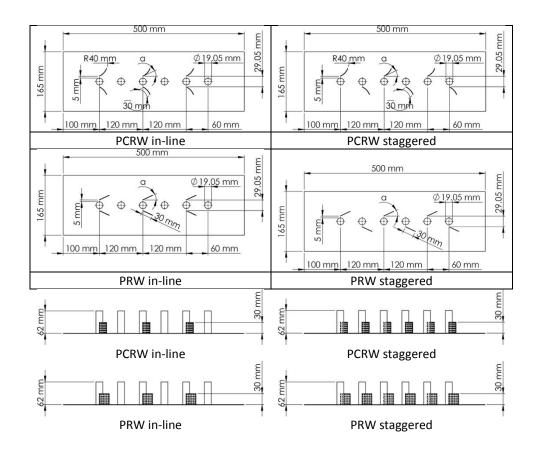
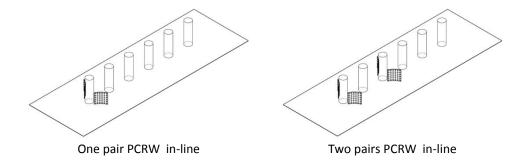


Fig. 3. Geometry and pitch of the VGs

The VGs configurations were arranged in-line and staggered on the plate. The perforated rectangular winglet (PRW) and perforated concave rectangular winglet (PCRW) VGs in-line configurations with one, two and three pairs are shown in Fig. 4. For each pair, the VGs were placed on the left and right sides of the first row of tubes. VGs were placed in the first- and third- row tubes for two pairs. For the three pairs, VGs were placed on the first-, third-, and fifth-row tubes.



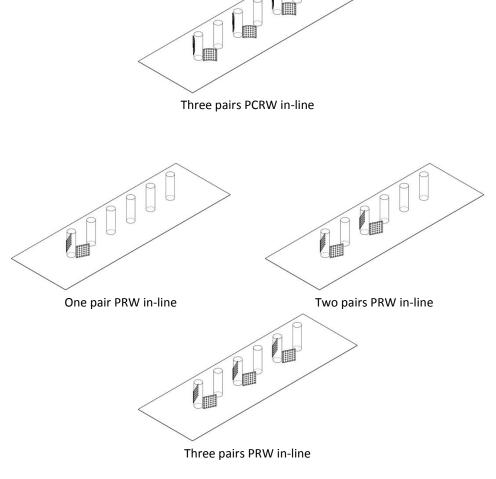
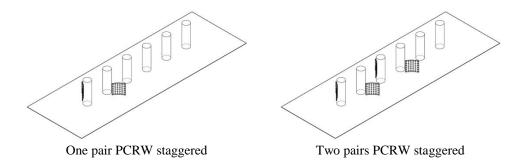
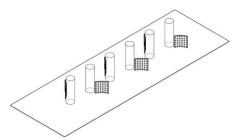


Fig. 4. VGs pairs in-line configurations

The PRW and PCRW VGs staggered configurations with one, two, and three pairs are shown in Fig. 5. For one pair, the VGs are placed on the right side of the first-row tube and on the left side of the second. The VGs are placed on the right side of the first and third row tubes and on the left side of the second and fourth tubes for two pairs. For the three pairs, the VGs are placed on the right side of the first, third and fifth rows of the tubes and on the left side of the second, fourth and sixth tubes.





Three pairs PCRW staggered

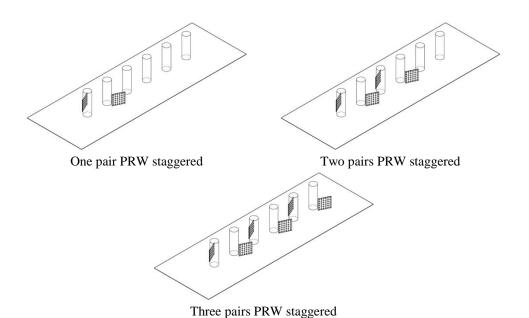


Fig. 5. VGs pairs staggered configurations

2.2 Parameter definitions

The parameters in this study were derived from the equation used by Oneissi et al. to obtain the thermal enhancement factor (TEF) [22]

$$TEF = \frac{\frac{Nu}{Nu_0}}{\left(\frac{f}{f_0}\right)^{\frac{1}{3}}} \tag{1}$$

The Nusselt number dan friction factor for the baseline conditions are symbolised as (Nu_o) and (f_0) , and (Nu) and (f) based on the research of Zeeshan et al[23]

$$Nu = \frac{q D_h}{A_{tube} \Delta T_{LMTD} k} \tag{2}$$

$$h = \frac{q}{A_{tube} \, \Delta T_{LMTD}} \tag{3}$$

$$q = \dot{m} c_p \left(T_{out} - T_{in} \right) \tag{4}$$

where D_h , A_{tube} , ΔT_{LMTD} , \dot{m} , c_p , T_{out} and T_{in} , are hydraulic diameter, tube surface area, log mean temperature difference, mass flow rate, specific heat, outlet temperature, and inlet temperature, respectively

$$D_h = \frac{4A_c}{p} = \frac{4ab}{2(a+b)} = \frac{2ab}{a+b} \tag{5}$$

$$\Delta T_{LMTD} = \frac{(\overline{T}_{tube} - \overline{T}_{out}) - (\overline{T}_{tube} - \overline{T}_{in})}{\ln[(\overline{T}_{tube} - \overline{T}_{out}) - (\overline{T}_{tube} - \overline{T}_{in})]}$$
(6)

where A_c dan T_{tube} are channel surface area and tube temperature, respectively.

The result of \mathcal{D}_h is used to calculate Re with the formula

$$Re = \frac{\rho u_{in} D_h}{\mu} \tag{7}$$

and friction factor (f) was determined to evalute the performance of hydro dynamic using

$$f = \frac{2 \Delta P D_h}{\rho V^2 (L+6D)} \tag{8}$$

where ρ , V, and L are the air density, inlet airflow velocity and length of the test specimen, respectively. The equation required to determine the cost-benefit ratio (CBR), defined as the ratio of pressure drop per variation in Nu number, as formulated by Tian et al. [25], is as follows:

$$CBR = \frac{\%\Delta P}{\%Nu} \tag{9}$$

This concept investigates whether the method used to enhance the heat-transfer rate is economically efficient. In the hydrodynamic test, the pressure drop (ΔP) is measured by the pressure difference on the sides of P_{inlet} and P_{outlet} of the test specimen in the tested part using equation (10):

$$\Delta P = P_{inlet} - P_{outlet} \tag{10}$$

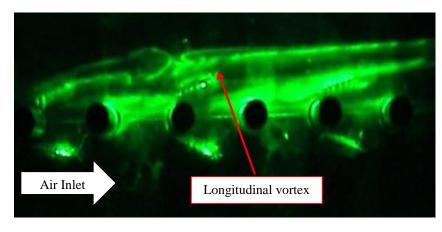
2.3 Validation

The current study is a follow-up investigation to the work of Yafid et al. [24], and the experimental setup was similar to that of Yafid et al. The difference between the current study and the experiment of Yafid et al. is a test object in which the current study uses concave rectangular winglet (CRW) VGs; in Yafid et al.'s experiment concave delta winglet (CDW) VGs are used. Whitaker et al. [25] studied the heat transfer characteristics of airflow through a single cylinder in a rectangular duct. The results of Yafid et al. were valid, and the same experimental set-up was determined. The *Nu* value from the experiment of Yafid et al. were comparable with the *Nu* values from the experiments of Whitaker et al. in the Reynolds number (*Re*) range of 2,143 to 11,763.

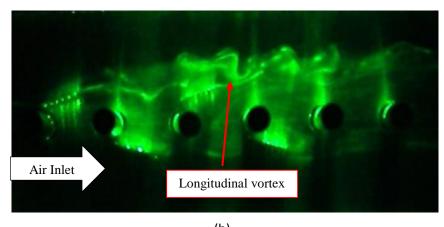
3. Results and Discussion

3.1 Flow visualisation

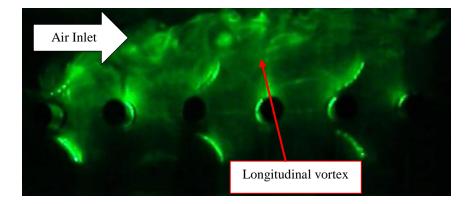
A flow visualisation test was performed to observe the longitudinal vortices (LV) formed after the flow passed through the VGs in the rectangular channel. This test was conducted under low-light conditions to clarify the LV. The laser beam was refracted by a cylindrical glass (diameter 5 mm), which produced a cross-sectional area perpendicular to the direction of the flow. Smoke formed from the evaporation of the liquid was used to visualise the LV in the flow. The VGs used in this visualisation test were PRWP and PCRWP with an in-line arrangement, as shown in Fig. 6.



(a)



(b)



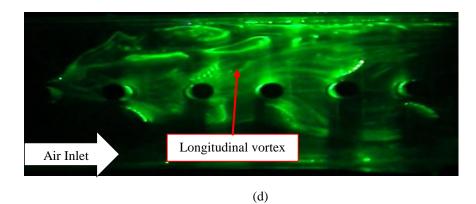


Fig. 6. Visualisation of LV generated by (a) in-line PRWP, (b) staggered PRWP, (c) in-line PCRWP and (d) staggered PCRWP

In Fig. 6 (c) and (d), the PCRWP VGs appear to produce longitudinal vortices (LV) in a wide flow area compared with the PRWP VGs in Fig 6 (a) and (b) downstream. The back region of the PCRWP VGs had a wider frontal surface area than the PRWP VGs. Consequently, mixing the near-fluid the channel walls with the fluid in the mainstream is better, meaning that the heat transfer rate is increased [26]. Downstream, the LV compression in the wake area increases the fluid flow velocity passing through the cylindrical structure, thereby increasing the heat transfer rate from the channel surface to the fluid flow in the wake region [27]. The increase in heat transfer produced when using PCRWP VGs was better than that with PRWP VGs.

3.2 Perforated vortex generators effect on heat transfer

The increase in the convection heat transfer was due to the mixing of fluids caused by the strong longitudinal vortices (LVs)[28]. The strength of the LVs is caused by the amount of VGs sets; increasing the amount of VGs pairs in the test specimen can increase the coefficient of the convection heat transfer [29], as shown in Fig. 7.

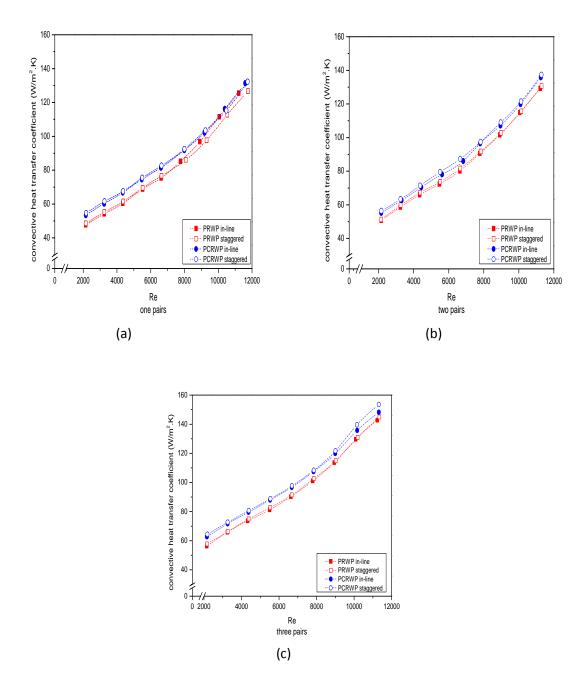


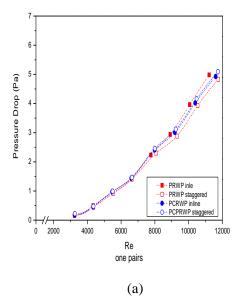
Fig. 7 Graphs of convective heat-transfer coefficient against Reynolds number: (a) one, (b) two and (c) three pairs.

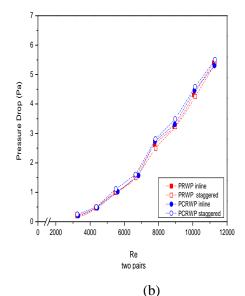
In Fig 7, we can see the convective heat transfer coefficient with respect to the Reynolds number (*Re*), analysed after installing the PCRWP and PRWP with pairs ranging from one, two and three, arranged in-line or staggered. Based on Fig. 7, the convective heat transfer coefficient increased with a rise in *Re* due to an increase in flow vortices and high turbulence intensity in the channel[30], alongside a reduction in the wage region and stagnation area for each increase in flow velocity[31]. The improve in heat transfer for the staggered was better than that for the PCRW VGs with any number of pairs at the

highest *Re* (11,000). The results in Fig. 7 show that the PCRWP VGs worked better than the PRWP VGs, and the staggered arrangement of the former, with three pairs, gave the highest yield (153.5 W/m²·K), as shown in Fig. 7(c). Two PCRW pairs (137.33 W/m²·K, Fig. 7(b)) were better than one (132.25 W/m²·K) (Fig. 7(a)) because the VGs with a concave surface destabilise the force of centrifugal of the fluid flow, strengthening the flow vortices and making the mixing of the hot fluid near the wall with the cold fluid of the main flow more robust[32]. In Fig. 7(a), the convection heat-transfer coefficient for the case of the in-line PRW VGs has the same value as that of the in-line or staggered PCRW VGs in a pair of VGs. In one pair of VGs, a longitudinal vortex is generated after the flow hits and weakens the VGs[29]. This result contrasts with the cases with two and three pairs of VGs, where the longitudinal vortex produced after striking the first VGs is amplified again when the flow strikes the second VGs and so on. Therefore, the value of the heat transfer coefficient in the case of a pair of PRW VGs is the same value as that of PCRW VGs at Reynolds numbers above 8,000.

3.3 Effect of perforated vortex generators on pressure drop

Using VGs can affect the increase in heat transfer, but there is often an accompanying increase in pressure drop, as shown in Fig. 8, where an increase in pressure drop can be seen along with the increases in *Re* and pair numbers for both the VG types PCRW and PRW. In general, the highest pressure drop was observed using the PCRWP VGs with a staggered configuration for all *Re*, except for one pair of VGs. The highest pressure drop was found in the PRWP VGs with an in-line configuration at *Re* greater than 8,000. The pressure drop on the staggered VGs was found to be higher than that on the in-line configuration because of the shorter distance between the VGs of the staggered configuration than that of the in-line[29], caused by the resistance of fluid flow against the walls of the VGs and the expansion of the frontal zone of the VGs in the next-pair arrangement [33]. The pressure drop in the staggered arrangement was lower than that in the in-line arrangement, whereas the PRW VGs type created a lower pressure drop than the PCRW VGs because the latter reduced the frontal area hit by the airflow, resulting in a decrease in drag [34]. In addition, the jet flow from the VG hole can reduce the stagnation flow, which can reduce the pressure drop [35]. A significant decrease in the pressure drop was due to the VG perforation[36]. The best pressure drop value for one pair with a staggered arrangement was 4.58 Pa (see Figure 8(a)), whereas two pairs (5 Pa, Figure 8(b)) were better than three (5.4 Pa, Figure 8(c)).





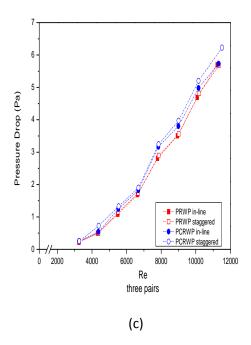
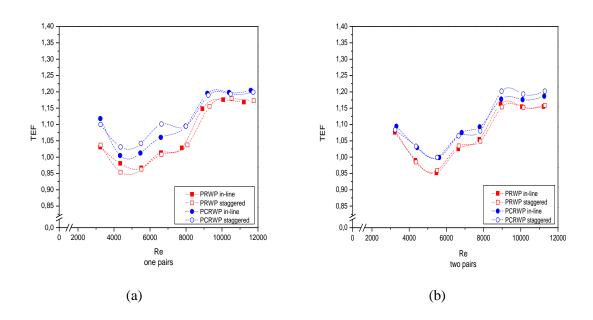


Fig. 8 Graph of pressure drop against Reynolds number: (a) one, (b) two and (c) three pairs

3.4 Effect of perforated VGs on thermal enhancement factor

TEF exhibited the hydraulic thermal performance while using VGs, which played a role in restructuring the incoming fluid flow pattern. The increase in the TEF was due to the influence of complex overlapping structures, which meant that the flow developed into a turbulent structure, significantly affecting the heat transfer increase [37]. The experimental TEF values are shown in Fig. 9.



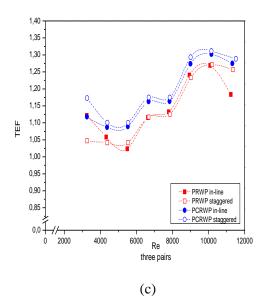


Fig. 9 Graph of thermal enhancement factor against Reynolds number: (a) one, (b) two and (c) three pairs

The TEF is the thermal-hydraulic performance which is the ratio of the increase in heat transfer to the pressure drop ratio. In general, the highest TEF was observed when the PCRWP VGs were used with a staggered configuration, as depicted in Fig 9. The PCRW creates wider flow vortices that can reduce the wake area behind the cylinder. Reducing the wake area can reduce the recirculation zone, affecting the heat transfer from the back of the cylinder to the stream [26]. A large-radius, high-intensity anterior-posterior vortex can reduce the wake area. A lessening within the wake zone increased the flow velocity behind the tube and reduced the recirculation area, resulting in increased heat transfer in this area [27, 38]. As shown in Fig 9, there was an increase in the TEF with greater pairs of VGs used for both the PCRW and PRW VG because the PCRW produced wider flow vortices, which reduced the wake region behind the cylinder, thereby reducing the recirculation zone and impacting the heat transfer increment from the rear cylinder surface to the stream[39]. In this process, a large number of longitudinal vortices with high intensities can reduce the wake area, which increases the flow velocity downstream of the tube and reduces the recirculation region, leading to an increased heat-ransfer rate in the region [40, 41]. Based on Figure 9, the best TEF increase occurred at Re between 8,000–9,000. The best TEF values, with one, two and three pairs occurred in the staggered arrangement with PCRW VGs, at 1.18, 1.20 and 1.29, respectively (see Fig. 9).

3.5 Effects of perforated VGs on the cost-benefit ratio

Economic evaluation cannot be conducted based only on the *TEF* and the net profit from the heat load of the transferred unit[26]. Instead, it must be determined by evaluating the economic value of the heat-transfer improvement by calculating the cost benefit ratio (*CBR*), as shown in Fig. 10.

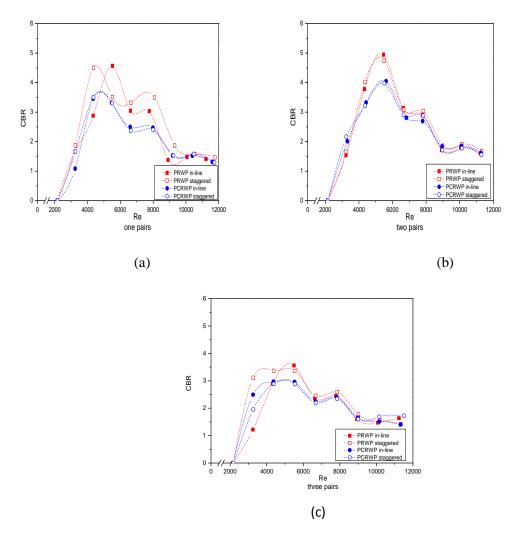


Fig. 10 Graph of cost-benefit ratio against Reynolds number: (a) one, (b) two and (c) three pairs

Fig. 10 show the result of the *CBR* calculation to compare the percentage increase in the pressure drop with the percentage increase in the Nusselt number when using VGs. These results indicate that a lower *CBR* improves thermal performance, which is greater than the drag force [25]. The greatest increase in *CBR* occured with the PRW VGs, with an in-line arrangement, totalling 4.57, 4.95 and 3.56 for one, two and three pairs, respectively. The lowest *CBR* was measured when three sets of PCRW vortex generators with a staggered arrangement were used. The lowest *CBR* were obtained with the three pairs of staggered-type VGs PCRW. The three VG pairs showed a lower *CBR* than the one and two pairs because they resulted in the greatest increase in the Nusselt number, accompanied by a lower pressure drop increase, which lowered the *CBR*. These results show that a lower *CBR* improves thermal performance relative to resistivity [26]. A low value *CBR* inicates a more economical value using VGs. In general, using PCRWP VGs with a staggered configuration is the best.

3.6 Heat Loss Analysis

Heat loss analysis was performed by considering the convection heat transfer from the six tubes to the surrounding fluid flow. The heat transfer rate was calculated for laminar and turbulent flows.

The heat loss in this experiment was calculated by calculating the difference between the induced electric power and total heat through convection from the surface of the tubes to the fluid. In this experiment, six tubes in a wind tunnel were heated using a heater at a power of 40 W; the velocity of the inlet fluid is varied from 0.4 to 2 m/s at intervals of 0.2 m/s or in the Reynolds number range from 2,143 to 11,763. Based on the Reynolds number range, two types of flows were determined; laminar and turbulent. Therefore, the heat loss was determined from the correlation between laminar at 0.4 m/s and turbulent for other velocities. The experimental data for the hydraulic diameter D_h , tube surface area A_{tube} , channel surface area A_c and air specific heat c_p are 0.09223 m, 0.02338908 m², 0.01056 m² and 1.007 J/kgK, respectively. Table 1 is a baseline for calculating heat loss

Table 1 Heat Loss Baseline

| } - - | v (m/s) | Re | Mass flow rate (kg/s) | Density (kg/m³) | Dynamics viscous (kg/ms) | k | Pr | T inlet (C) | T outlet (C) | T tube (C) | Δ T LMTD | ΔT (T tube - T inlet) | Nu | h (W/mK) | q conv (W) | q input (W) | q loss (W) |
|---------------|------------|-------|-----------------------------|--------------------|--------------------------------|------|------|-------------|--------------|------------|--------------------|-------------------------------|-----|-------------|------------------|-------------|------------------|
| | 0.4 | 2165 | 0.004757 | 1.13 | 1.9.E-05 | 0.03 | 0.73 | 29 | 33 | 50 | 19 | 21 | 155 | 45 | 19.48 | 40 | 20.52 |
| : | 0.6 | 3291 | 0.00719 | 1.13 | 1.9.E-05 | 0.03 | 0.73 | 28 | 31 | 46 | 16 | 18 | 174 | 50 | 18.98 | 40 | 21.02 |
| , baseline | 0.8 | 4413 | 0.009618 | 1.14 | 1.9.E-05 | 0.03 | 0.03 | 28 | 30 | 44 | 15 | 16 | 192 | 50 | 19.19 | 40 | 20.81 |
| } | 1 | 5545 | 0.012056 | 1.14 | 1.9.E-05 | 0.03 | 0.73 | 28 | 30 | 43 | 14 | 15 | 214 | 55 | 19.84 | 40 | 20.16 |
|) | 1.2 | 6661 | 0.014477 | 1.14 | 1.9.E-05 | 0.03 | 0.73 | 28 | 29 | 43 | 14 | 15 | 228 | 61 | 21.15 | 40 | 18.85 |
| - | 1.4 | 7826 | 0.016958 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 28 | 29 | 41 | 12 | 13 | 247 | 70 | 20.30 | 40 | 19.70 |
| | 1.6 | 8965 | 0.019407 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 27 | 29 | 40 | 12 | 13 | 263 | 75 | 21.03 | 40 | 18.97 |
| | 1.8 | 10110 | 0.021863 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 27 | 28 | 39 | 12 | 12 | 296 | 84 | 22.54 | 40 | 17.46 |
| | 2 | 11272 | 0.024341 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 27 | 28 | 38 | 11 | 11 | 342 | 97 | 24.17 | 40 | 15.83 |

From Table 1, the greater the velocity with an increase in the *Re*, the lower the heat loss. It can be observed that the heat flow from the heater not only spreads into the tube, but convection also occurs outside the tube. The heat output increased with *Re*, i.e., the higher the flow velocity, the greater the turbulence through the cylinder and the higher the turbulence intensity. An increase in the turbulence intensity between a cold airflow and hot cylinder with a constant surface temperature is caused by the airflow velocity[26]. In row-tube arrays, this recirculation area increased for the second and subsequent columns. A lower air velocity in the circulation region indicated less airflow in the region participating in the local heating process[37]. The heat loss under all conditions in this experiment is listed in table 2.

Table 2 Calculation of heat loss for the whole case

| type VGs | q conv (W) | q input (W) | q loss (W) |
|----------|------------|-------------|------------|
| Baseline | 20.74 | 40 | 19.26 |
| PCRWPI1 | 25.15 | 40 | 14.85 |
| PCRWPI2 | 27.55 | 40 | 12.45 |

| PCRWPI3 | 27.61 | 40 | 12.39 |
|---------|-------|----|-------|
| PCRWPS1 | 26.43 | 40 | 13.57 |
| PCRWPS2 | 26.43 | 40 | 13.57 |
| PCRWPS3 | 27.94 | 40 | 12.06 |
| PRWPI1 | 24.09 | 40 | 15.91 |
| PRWPI2 | 27.25 | 40 | 12.75 |
| PRWPI3 | 28.82 | 40 | 11.18 |
| PRWPS1 | 23.94 | 40 | 16.06 |
| PRWPS2 | 26.37 | 40 | 13.63 |
| PRWPS3 | 28.12 | 40 | 11.88 |
| | | | |

Table 2 shows that the lowest heat loss occurs when three sets of PCRWPs are staggered. The placement of the VGs can increase heat transfer in square ducts as the VGs create longitudinal vortices, which in turn increase vortex strength in the wake region downstream of the tube. Longitudinal vortices make the overall temperature field more uniform, improve heat mixing and boundary layer modification, and improve heat transfer performance. A higher number of vortex generators creates more longitudinal vortices and significantly increases heat transfer [29, 26].

3.7 Uncertainty Analysis

In this section, uncertainty analysis calculation data will be shown for the temperature at base-line conditions with a velocity of 0.4 m/s as shown in Table 3.

Table 3 Base-line test temperature data at a speed of 0.4 m/s

| $T(Tube_1)$ | $T(Tube_2)$ | $T(Tube_3)$ | $T(Tube_4)$ | $T(Tube_5)$ | $T(Tube_6)$ |
|-------------|-------------|-------------|-------------|-------------|-------------|
| 49.19093 | 51.21368 | 48.32313 | 49.76915 | 47.80219 | 51.27142 |
| 49.1834 | 51.17728 | 48.3156 | 49.79053 | 47.7657 | 51.2639 |
| 49.14545 | 51.16826 | 48.30655 | 49.7526 | 47.7856 | 51.25489 |
| 49.12105 | 51.17277 | 48.28214 | 49.72821 | 47.76118 | 51.2594 |
| 49.15297 | 51.20465 | 48.28515 | 49.73122 | 47.73524 | 51.2624 |
| 49.09966 | 51.15141 | 48.28967 | 49.73573 | 47.76871 | 51.26691 |
| 49.09815 | 51.14991 | 48.23029 | 49.73423 | 47.73826 | 51.29428 |
| 49.08912 | 51.14089 | 48.25019 | 49.66739 | 47.72922 | 51.22751 |

From these data, it is found that \overline{T}_{Tube} can be calculated by the equation as

$$\overline{T}_{Tube} = \frac{\overline{T}_{Tube1} + \overline{T}_{Tube2} + \overline{T}_{Tube3} + \overline{T}_{Tube4} + \overline{T}_{Tube5} + \overline{T}_{Tube6}}{6} = 49.56^{\circ}\text{C}$$
Then, the average standard deviation is obtained by the following formula.

$$s_{tube} = \sqrt{\frac{\sum_{i=1}^{N} (T_{tubei} - \overline{T}_{tube})^2}{N(N-1)}} = 0.029$$
 (12)

Therefore, the average T_{tube} can be written as 49.5 ± 0.029 °C. \bar{T}_{out} calculation results obtained 32.95°C. The average standard deviation was calculated using the following equation:

$$s_{Tout} = \sqrt{\frac{\sum_{i=1}^{N} (T_{outi} - \overline{T}_{out})^2}{N(N-1)}} = 0,051$$
 (13)

Furthermore, the average value of T_{out} can be written as 32.95 \pm 0.051°C. Using the same equation, the standard deviation of T_{in} was found to be 0.033. Thus, the average T_{in} value was 29.75 \pm 0.016°C.

The value of q at a speed of 0.4 m/s was found to be 19.48 W. To determine of the standard deviation of q, the following equation was used:

$$RSS_{q} = \sqrt{\left(s(\Delta T_{out})\frac{\partial q}{\partial T_{out}}\right)^{2} + \left(s(\Delta T_{in})\frac{\partial q}{\partial T_{in}}\right)^{2}}$$

$$\frac{\partial q}{\partial T_{out}} = \frac{\partial (m \cdot c_{p} \cdot T_{out} - m \cdot c_{p} \cdot T_{in})}{\partial T_{out}} = m \cdot c \cdot p$$

$$\frac{\partial q}{\partial T_{in}} = \frac{\partial (m \cdot c_{p} \cdot T_{out} - m \cdot c_{p} \cdot T_{in})}{\partial T_{in}} = -(m \cdot c \cdot p)$$
(14)

where $s(\Delta T_{out}) = 0.051$ °C and $s(\Delta T_{in}) = 0.033$ °C, ensuring, that $RSS_q = \pm 0.290$ W. Therefore, the heat transfer rate q becomes 19.48 ± 0.290 W. The value of ΔT_{lmtd} at a speed of 0.4 m/s was found to be 18.56°C. To determine the value of the standard deviation of ΔT_{lmtd} we used the following equation:

$$RSS_{\Delta T_{lmtd}} = \sqrt{\left(s(\Delta T_{tube})\frac{\partial(\Delta T_{lmtd})}{\partial T_{tube}}\right)^2 + \left(s(\Delta T_{out})\frac{\partial(\Delta T_{lmtd})}{\partial T_{out}}\right)^2 + \left(s(\Delta T_{in})\frac{\partial(\Delta T_{lmtd})}{\partial T_{in}}\right)^2}$$
(15)

$$\frac{\partial (\Delta T_{lmtd})}{\partial T_{tube}} = \frac{\partial \left(\frac{(T_{tube} - T_{out}) - (T_{tube} - T_{in})}{\ln \frac{T_{tube} - T_{out}}{T_{tube} - T_{in}}} \right)}{\partial T_{tube}}$$

$$\frac{\partial (\Delta T_{lmtd})}{\partial T_{out}} = \frac{\partial \left(\frac{(T_{tube} - T_{out}) - (T_{tube} - T_{in})}{\ln \frac{T_{tube} - T_{out}}{T_{tube} - T_{in}}} \right)}{\partial T_{out}}$$

$$\frac{\partial (\Delta T_{lmtd})}{\partial T_{in}} = \frac{\partial \left(\frac{(T_{tube} - T_{out}) - (T_{tube} - T_{in})}{\ln \frac{T_{tube} - T_{out}}{T_{tube} - T_{in}}} \right)}{\partial T_{in}}$$

where $s(\Delta T_{tube}) = 0.029$ °C, $s(\Delta T_{out}) = 0.051$ °C and $s(\Delta T_{in}) = 0.033$ °C; we get $RSS_{\Delta T_{lmtd}}$ of ± 0.043 , ensuring, that the obtained ΔT_{lmtd} is 8.56 ± 0.043 .

The value of Nu at a speed of 0.4 m/s was found to be 155.31. The standard deviation of Nu was obtained using following equation

$$RSS_{Nu} = \sqrt{\left(s(q)\frac{\partial Nu}{\partial O}\right)^2 + s(\Delta T_{lmtd})\frac{\partial Nu}{\partial \Delta T_{lmtd}}}$$
(16)

$$\frac{\partial Nu}{\partial q} = \frac{\partial \left(q \cdot D_h \cdot At^{-1} \cdot \Delta T_{lmtd}^{-1} \cdot k^{-1}\right)}{\partial q} = \frac{D_h}{(At)(\Delta T_{lmtd})(k)}$$

$$\frac{\partial Nu}{\partial \Delta T_{lmtd}} = \frac{\partial \left(q \cdot D_h \cdot At^{-1} \cdot \Delta T_{lmtd}^{-1} \cdot k^{-1}\right)}{\partial \Delta T_{lmtd}} = \frac{q \cdot D_h}{(At)(\Delta T_{lmtd})^2(k)}$$

With the values of $s(q) = 0.290 \ W$ and $s(\Delta T_{lmtd}) = 0.043$, the obtained RSS_{Nu} was $\pm 2.889 \ \text{W/(m}^{2\circ}\text{C)}$. Therefore, the value of RSS_{Nu} is $155.31 \pm 2.889 \ \text{W/m}^{2\circ}\text{C}$.

The value of h at a speed of 0.4 m/s was found to be 44.86. To determine the standard deviation of *Nu* the following equation is used

$$RSS_{h} = \sqrt{\left(s(Nu)\frac{\partial h}{Nu}\right)^{2}}$$

$$\frac{\partial h}{\partial Nu} = \frac{\partial (h.D_{h}.k^{-1})}{\partial h} = \frac{k}{D_{h}}$$
(17)

Furthermore, the value of D_h is 0.092 m and k at $T_f = 40.24$ is 0.026. So the value of h at a speed of 0.4 m/s is:

$$RSS_h = \sqrt{\left(s(Nu)\frac{\partial h}{Nu}\right)^2} = 0.83$$

Thus, the number h at a speed of 0.4 m/s is 44.86 \pm 0.83. So, the error h for the baseline at a speed of 0.4 m/s is

$$Error = \frac{RSS_h}{h} \times 100 \tag{18}$$

$$Error = \frac{0.83}{44.86} \times 100 = 1.51\%$$

From the test in the baseline case with a speed of 2.0 m/s, the results of the pressure drop are listed in Table 4, which show that the average P can be calculated as follows:

$$\overline{\Delta P} = \frac{\Delta P_1 + \Delta P_2 + \Delta P_3 + \dots + \Delta P_{30}}{30} = 3.51 \text{ Pa}$$

$$\tag{19}$$

The average standard deviation of the pressure drop can then be calculated using the equation

$$s = \sqrt{\frac{\sum_{i=1}^{N} (\Delta P_i - \overline{\Delta P})^2}{N(N-1)}} = 8.9 \times 10^{-5}$$
 (20)

Baseline case for the pressure drop value at a speed of 2.0 m/s is $3.51 \pm 8.9 \times 10^{-5}$ Pa. Then, the error in the form of percentage can be calculated using the following equation:

$$\frac{8.9 \times 10^{-5}}{3.51} \times 100 = 0.71$$

Table 4 Baseline pressure drop data at a speed of 2.0 m/s

| | | ΔP (Pa) | |
|---------|---------|-----------------|---------|
| Data to | 2.0 m/s | Data to | 2.0 m/s |
| 1 | 0.013 | 16 | 0.012 |
| 2 | 0.013 | 17 | 0.013 |
| 3 | 0.013 | 18 | 0.012 |
| 4 | 0.013 | 19 | 0.012 |
| 5 | 0.012 | 20 | 0.013 |
| 6 | 0.013 | 21 | 0.013 |
| 7 | 0.013 | 22 | 0.012 |
| 8 | 0.012 | 23 | 0.013 |
| 9 | 0.013 | 24 | 0.012 |
| 10 | 0.013 | 25 | 0.013 |
| 11 | 0.013 | 26 | 0.013 |
| 12 | 0.013 | 27 | 0.013 |
| 13 | 0.012 | 28 | 0.013 |
| 14 | 0.012 | 29 | 0.012 |
| 15 | 0.013 | 30 | 0.012 |

The equal calculation approach changed into used for all data. Therefore, the overall error outputs for the pressure-drop vortex generator with placement variations (in-line and staggered), *Re* and amount of VG sets (one, two and three) are listed in Table 5.

Table 5. Overall Pressure Drop (ΔP)

| Vortex Generator | Overall Error P |
|-------------------|-----------------|
| Variations | (perforated) |
| 1 PRWP in-line | 2.94% |
| 2 PRWP in-line | 2.87% |
| 3 PRWP in-line | 1.98% |
| 1 PRWP staggered | 2.88% |
| 2 PRWP staggered | 2.34% |
| 3 PRWP staggered | 1.36% |
| 1 PCRWP in-line | 2.72% |
| 2 PCRWP in-line | 1.80% |
| 3 PCRWP in-line | 1.80% |
| 1 PCRWP staggered | 2.43% |
| 2 PCRWP staggered | 1.91% |

The average *TEF* results from the experimental results can be calculated as follows.

$$\overline{TEF} = \frac{TEF_1 + TEF_2 + TEF_3 + \dots + TEF_{12}}{12} = 1.12$$
 (21)

Then, the average standard deviation of the TEF can be calculated with the equation

$$s = \sqrt{\frac{\sum_{i=1}^{N} (TEF_i - \overline{TEF})^2}{N(N-1)}} = 1.07$$
 (22)

Therefore, the TEF value was 1.12 ± 1.07 . Then, the error in the form of percentage can be calculated using the following equation:

$$\frac{1.07}{1.12} \times 100 = 0.94\%$$

The overall error results for the *TEF* vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are are listed in Table 6.

Table 6. Overall error TEF

| Variasi Vortex Generator | Overall Error TEF |
|--------------------------|-------------------|
| | (Berlubang) |
| 1 RWP in-line | 0.47 % |
| 2 RWP in-line | 0.47% |
| 3 RWP in-line | 0.43% |
| 1 RWP staggered | 0.47% |
| 2 RWP staggered | 0.47% |
| 3 RWP staggered | 0.43% |
| 1 CRWP in-line | 0.45% |
| 2 CRWP in-line | 0.45% |
| 3 CRWP in-line | 0.42% |
| 1 CRWP staggered | 0.45% |
| 2 CRWP staggered | 0.45% |
| 3 CRWP staggered | 0.41% |

First, find the average *CBR* of the experimental results with the following formula.
$$\overline{CBR} = \frac{CBR_1 + CBR_2 + CBR_3 + \dots + CBR_{12}}{12} = 2.14$$
(23)

The average standard deviation of the pressure drop CBR can then be calculated using the following equation:

$$s = \sqrt{\frac{\sum_{i=1}^{N} (CBR_i - \overline{CBR})^2}{N(N-1)}} = 1.60$$
 (24)

The CBR value is 2.14 ± 1.60 . Then the error in the form of percentage can be calculated using thefollowing equation:

$$\frac{1.60}{2.14} \times 100 = 0.63\%$$

The overall error results for the *CBR* vortex generator with placement variations (in-line and staggered), *Re* and amount of VG sets (one, two and three) are listed in Table 7

Table 7. Overall error CBR

| Variasi Vortex Generator | Overall Error CBR |
|--------------------------|-------------------|
| | (Berlubang) |
| 1 RWP in-line | 0.32% |
| 2 RWP in-line | 0.29% |
| 3 RWP in-line | 0.45% |
| 1 RWP staggered | 0.32% |
| 2 RWP staggered | 0.31% |
| 3 RWP staggered | 0.45% |
| 1 CRWP in-line | 0.4% |
| 2 CRWP in-line | 0.42% |
| 3 CRWP in-line | 0.56% |
| 1 CRWP staggered | 0.43% |
| 2 CRWP staggered | 0.42% |
| 3 CRWP staggered | 0.66% |

Conclusion

Based on the experimental results for perforated concave rectangular winglet pair vortex generators (PCRWP VGs) used to increase the heat transfer of airflow through heated tubes arranged in-line in the duct, we conclude that using PCRWP VGs affects the convection heat transfer coefficient, pressure drop in achieving hydraulic thermal performance and cost-benefit ratio. In our investigation, the best heat-transfer convection coefficient was 153.5 W/m²·K for the three pairs of PCRW VGs, in a staggered manner. The greatest improvement in the pressure drop value (4.58 Pa), occurred for one pair of PCRW VGs arranged in a staggered manner, whereas the hydraulic thermal performance was the best (1.29) in this experiment with the three pairs of PCRW VGs arranged in a staggered manner. Finally, the best *CBR* (3.56) was recorded for the three pairs of PCRW VGs composed in a staggered manner.

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Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel

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Abstract

A significant increase in the rate heat transfer in a heat exchanger system is made possible by increasing the convection heat-transfer coefficient using a passive method. The addition of vortex generators (VGs) to the fins and tubes of a heat exchanger is currently the most effective passive method. However, the increase in heat was accompanied by an increase in pressure drop. Therefore, in this study, we installed perforated concave rectangular winglet pair vortex generators (PCRWP VGs) on plates in rectangular ducts to increase the heat transfer through the six heated tubes to the air stream by lowering the enhancement in the pressure drop. We attempted to determine the best cost-benefit ratio (*CBR*) with a fluid flow velocity difference of 0.4 –2 m/s at intervals of 0.2 m/s (Reynolds number (*Re*) of 2,143 to 11,763) in the channel. The PCRWP VGs were composed of in-line and staggered configurations. The results showed a lower *CBR* (3.56) for the in-line configuration than for the staggered configuration. Moreover, the lowest *CBR* was accompanied by an increase in thermal performance (*TEF*) of 1.29.

Keywords: Perforated; Rectangular winglet; Concave; Pressure drop; Vortex generator; Heat transfer; Thermal performance

1. Introduction

The global energy demand is expected to triple over the next few years. According to a statement by the International Energy Agency (IEA), the main driver is the increasing use of air conditioning (AC) machines [1]. Thus, promoting energy efficiency in air conditioners is important and requires maximising their thermal performance, which involves increasing the rate of heat transfer in its main component, i.e., the condenser. A condenser, commonly used in air conditioners, comprises a fin and a tube and functions as a refrigerant cooling medium. However, the high thermal resistance (75%) of the fin air side of the condenser lowers the heat-transfer rate in the heat exchanger[2]. Thus, the thermal resistance must be lowered to enhance the heat transfer rate.

A commonly used active methods to increase the rate of heat transfer involves adding vortex generators (VGs), which, according to the research results obtained by Mugisidi et al., increases the performance of a condenser[3]. The added VGs cause longitudinal vortices (LVs), damage the primary flow, make the second flow as large as the first and increase air mixing in the area[4][5]. The size of the LVs, shape of the flow, and mixing are influenced by the shape, geometry and position of the VGs added to the fins and tubes of the heat exchanger[6].

Samidifat et al. showed that simple rectangular vortex generators (RVGs) can increase the heat transfer rate by 7%; however, this causes a pressure drop in the heat exchanger system[7]. Meanwhile, modified RVGs with a concave shape on the front and rear surfaces decreased the heat transfer performance of the heat exchanger tube. A better option is to use RVGs with a double convex front surface and a single concave back surface, which can strengthen the primary vortex, increasing the rate of heat transfer from the plate to the fluid, as demonstrated in a study by Kashyap et al.[8]. Further research conducted by Kashyap et al. in the same year concluded that modifying the surface shape of rectangular winglet vortex generators (RWVGs) can create longitudinal eddies that interact with the boundary layer, thereby increasing the rate of convection heat transfer[9]. Based on their research, the

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increase in the optimal heat transfer rate was 14.4. The optimal heat transfer performance was also obtained from the results of experiments conducted by Adnan et al. on rectangular ducts by adding delta and rectangular winglet VGs[10]. Concave curved delta winglet VGs were compared with convex curved delta winglet VGs by Song et al. to observe changes in the heat transfer rate[11]. The results showed that the concave VGs improved the heat transfer better than the convex VGs. The differences in the shape of the VGs affects the change in the heat transfer rate and the change in the geometry of the VG, such as a new rib geometry in the cylinder channel[12].

Zeeshan et al. showed that increasing the angle of attack increased the rate of heat transfer (to 37.01-64.54%) if a pair of RWVGs were placed at the back of the tube even though this did not reduce the pressure drop[13]. A decrease in the value of the pressure drop also did not occur significantly, even though there was an increase in heat of 260% in heat, as per the results of the research conducted by Linardo et al. using the batched heat and channelled pipe (BHCP) approach[14]. The increase in heat transfer performance is influenced by the number of RWVG pairs based on the research results of Heriyani et al., where there is an increase in the hydraulic thermal performance evaluation criteria by 15.17% for three pairs of RWVG compared with the baseline [15]. Wang et al. found that the more pairs of VGs placed in the crossflow, the higher the increase in the heat transfer coefficient [16]. Sun et al. further discovered that increasing the number of RWVGs in the heat exchanger tube increased the heat transfer, with a maximum thermal enhancement factor (TEF) of 1.27 [17]. The TEF value of a V-delta winglet VG reached 1.82-3% higher than that of a V-rectangular winglet VG, as revealed by Promvonge et al.[18]. These results were obtained with an optimal blockage ratio (BR) of 0.15 and pitch ratio (PR) 1.0. Skullong et al. modified the shapes of RWVGs with optimal BRs and PRs to achieve an optimum heat transfer performance and reduced pressure drop; their shape modification involved perforating RWVGs [19].

The positions of the holes in the RWVGs did not significantly affect the increase in heat transfer; however, they significantly affected the flow resistance of the VGs. The heat-transfer rate increased as the height (vertical position) of the hole increased. Widthwise, although there is an initial increase, the heat transfer rate decreased with increasing lateral distance[20]. An increase in the number of holes in the RWVGs indicates an increase in fluid flow, which forces the fluid to flow behind the RWVGs, thereby increasing heat transfer[4]. The heat transfer rate increased during laminar flow when the Reynolds number (Re) increased and then decreased with an increase in Re during turbulent flow [20]. Positioning the tube in-line with a pair of RWVGs in a common flow-down configuration provides better performance than the common flow-up configuration. However, a staggered tube position is superior, resulting in a 25.85% higher heat-transfer performance than when a pair of RWVGs is not used[21].

In the existing studies, no detailed analyses of heat transfer were conducted on from the surfaces of several cylinders heated and arranged in-line when using a perforated vortex generator. Therefore, the focus herein is on investigating the advantages of using perforated concave rectangular winglet pair vortex generators (PCRWP VGs) to increase the heat transfer of the airflow through heated tubes arranged in-line in the ducts.

2. Experimental Approach

2.1 Experimental setup

This research was conducted experimentally with a test equipment scheme comprising a rectangular channel sized 370 x 18 x 8 cm. The duct was made of 1 cm thick glass, as shown in Fig. 1.

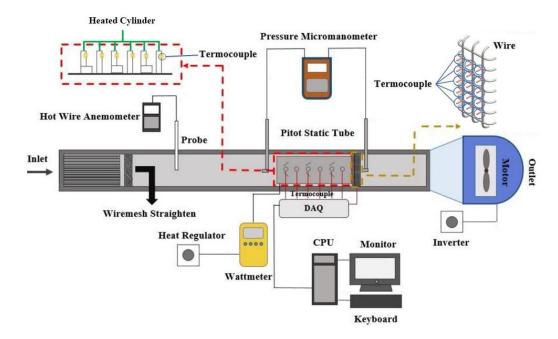


Fig. 1. Experimental tool schematic

Based on Fig. 1, the rectangular channel is equipped with a blower (50 Hz, Wipro with a rated voltage of 220V), an inverter (Mitsubishi Electric type FR-D700 with an accuracy of 0.01), straightener, hot wire anemometer (Lutron type AM-4204 with an accuracy of 0.1), wattmeter (Lutron DW-6060 with an accuracy \pm 1.0), central processing unit (CPU), micromanometer, thermocouple (K type with a temperature interval of -200 –1250°C and an accuracy \pm 0.5) where one thermocouple was placed in the air inlet area, six thermocouples on the back surface of the tubes and 15 on the outlet side of the wire, data acquisition (Advantech USB-4718 type with an accuracy of 0.001) and heater regulator. The heater was connected to six tubes with a diameter of 19.05 mm and height of 65.8 mm, with each tube having the same power. Total heating power of 40 W was applied to the six tubes using a regulator. The heating air flowing through the tubes occurs via convection. Thus, the air at the outlet side becomes hotter than that at the inlet side.

A pressure micromanometer (Fluke type 922, with an accuracy of \pm 0.05) was used to monitor the flow pressure drop. Two pitot tubes, each set 26 cm ahead of the inlet of the test specimen and 2.5 cm behind it, were connected to a micromanometer to measure the pressure drop. The pressure drop measurements were recorded 30 times for 5 sekon at each speed variation. Furthermore, flow visualisation was performed by directing the smoke from vaporised fluid in the fluid vaporator into the mainflow.

The VGs used as test specimens were perforated rectangular winglet pair (PRWP) and perforated concave rectangular winglet pair (PCRWP) vortex generators (VGs). Perforated is a term for holes in the VGs, as shown in Fig. 2. The VGs have dimensions of the same length and width of 30 mm, with

36 holes. The bore diameter on the VGs was 2.5 mm. The distance between the holes was 5 mm from the center.

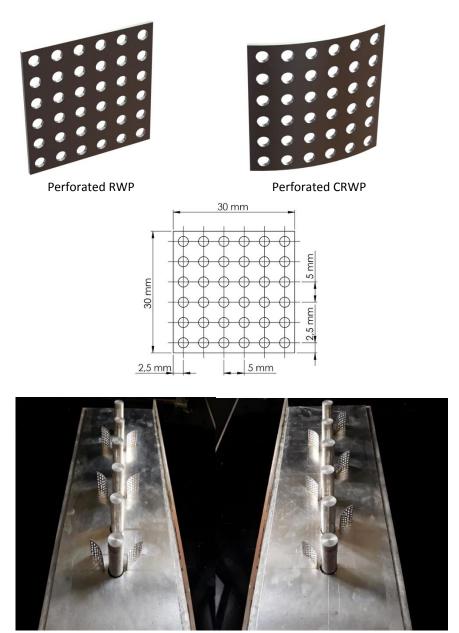


Fig. 2. Geometry of the VGs

The VGs are placed on an aluminium plate measuring $500 \times 165 \times 1$ mm. The geometry and the pitch between VGs for both in-line and staggered configurations are shown in Fig 3, with an angle of attack (α) of 150[2]. The distance between the cylinders is 120 mm, with a cylinder diameter of 19.05 mm.

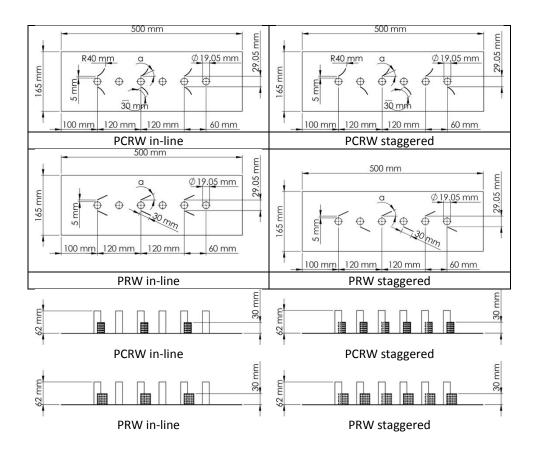
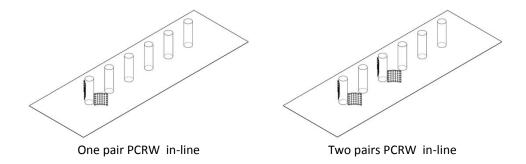


Fig. 3. Geometry and pitch of the VGs

The VGs configurations were arranged in-line and staggered on the plate. The perforated rectangular winglet (PRW) and perforated concave rectangular winglet (PCRW) VGs in-line configurations with one, two and three pairs are shown in Fig. 4. For each pair, the VGs were placed on the left and right sides of the first row of tubes. VGs were placed in the first- and third- row tubes for two pairs. For the three pairs, VGs were placed on the first-, third-, and fifth-row tubes.



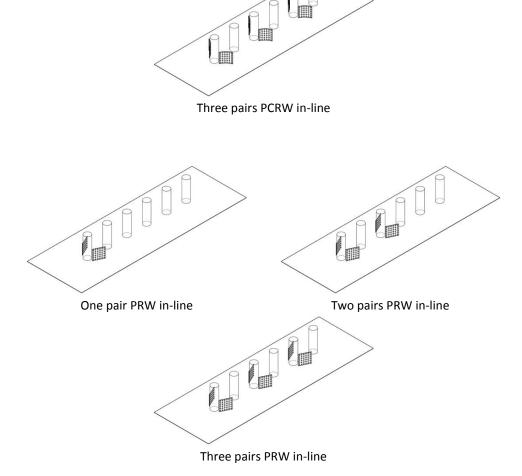
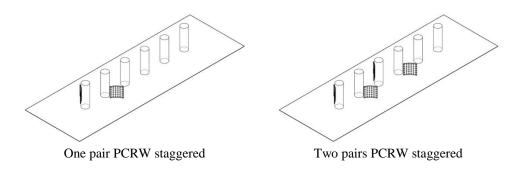
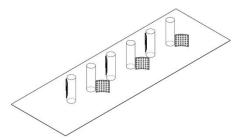


Fig. 4. VGs pairs in-line configurations

The PRW and PCRW VGs staggered configurations with one, two, and three pairs are shown in Fig. 5. For one pair, the VGs are placed on the right side of the first-row tube and on the left side of the second. The VGs are placed on the right side of the first and third row tubes and on the left side of the second and fourth tubes for two pairs. For the three pairs, the VGs are placed on the right side of the first, third and fifth rows of the tubes and on the left side of the second, fourth and sixth tubes.





Three pairs PCRW staggered

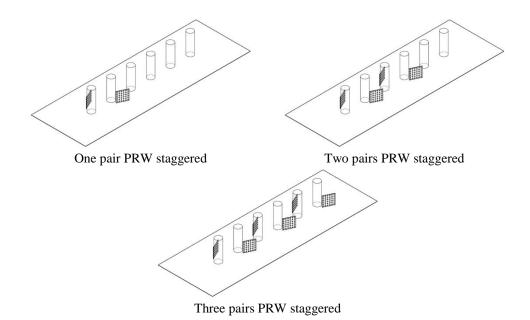


Fig. 5. VGs pairs staggered configurations

2.2 Parameter definitions

The parameters in this study were derived from the equation used by Oneissi et al. to obtain the thermal enhancement factor (TEF) [22]

$$TEF = \frac{\frac{Nu}{Nu_0}}{\left(\frac{f}{f_0}\right)^{\frac{1}{3}}} \tag{1}$$

The Nusselt number dan friction factor for the baseline conditions are symbolised as (Nu_0) and (f_0) , and (Nu) and (f) based on the research of Zeeshan et al[23]

$$Nu = \frac{q D_h}{A_{tube} \Delta T_{LMTD} k} \tag{2}$$

$$h = \frac{q}{A_{TUDe} \Delta T_{LMTD}} \tag{3}$$

$$q = \dot{m} c_p \left(T_{out} - T_{in} \right) \tag{4}$$

where D_h , A_{tube} , ΔT_{LMTD} , \dot{m} , c_p , T_{out} and T_{in} , are hydraulic diameter, tube surface area, log mean temperature difference, mass flow rate, specific heat, outlet temperature, and inlet temperature, respectively

$$D_h = \frac{4A_c}{p} = \frac{4ab}{2(a+b)} = \frac{2ab}{a+b} \tag{5}$$

$$\Delta T_{LMTD} = \frac{(\overline{T}_{tube} - \overline{T}_{out}) - (\overline{T}_{tube} - \overline{T}_{in})}{\ln[(\overline{T}_{tube} - \overline{T}_{out}) - (\overline{T}_{tube} - \overline{T}_{in})]} \tag{6}$$

where A_c dan T_{tube} are channel surface area and tube temperature, respectively.

The result of D_h is used to calculate Re with the formula

$$Re = \frac{\rho u_{in} D_h}{\mu} \tag{7}$$

and friction factor (f) was determined to evalute the performance of hydro dynamic using

$$f = \frac{2 \frac{\Delta P}{D_h}}{\rho V^2 (L + 6D)} \tag{8}$$

where ρ , V, and L are the air density, inlet airflow velocity and length of the test specimen, respectively. The equation required to determine the cost-benefit ratio (CBR), defined as the ratio of pressure drop per variation in Nu number, as formulated by Tian et al. [25], is as follows:

$$CBR = \frac{\%\Delta P}{\%Nu} \tag{9}$$

This concept investigates whether the method used to enhance the heat-transfer rate is economically efficient. In the hydrodynamic test, the pressure drop (ΔP) is measured by the pressure difference on the sides of P_{inlet} and P_{outlet} of the test specimen in the tested part using equation (10):

$$\Delta P = P_{inlet} - P_{outlet} \tag{10}$$

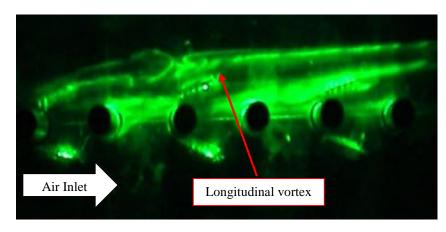
2.3 Validation

The current study is a follow-up investigation to the work of Yafid et al. [24], and the experimental setup was similar to that of Yafid et al. The difference between the current study and the experiment of Yafid et al. is a test object in which the current study uses concave rectangular winglet (CRW) VGs; in Yafid et al.'s experiment concave delta winglet (CDW) VGs are used. Whitaker et al. [25] studied the heat transfer characteristics of airflow through a single cylinder in a rectangular duct. The results of Yafid et al. were valid, and the same experimental set-up was determined. The *Nu* value from the experiment of Yafid et al. were comparable with the *Nu* values from the experiments of Whitaker et al. in the Reynolds number (*Re*) range of 2.143 to 11.763.

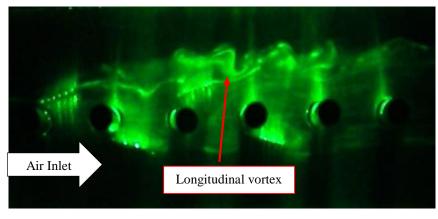
3. Results and Discussion

3.1 Flow visualisation

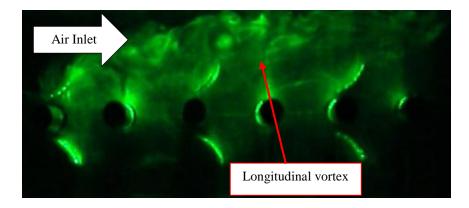
A flow visualisation test was performed to observe the longitudinal vortices (LV) formed after the flow passed through the VGs in the rectangular channel. This test was conducted under low-light conditions to clarify the LV. The laser beam was refracted by a cylindrical glass (diameter 5 mm), which produced a cross-sectional area perpendicular to the direction of the flow. Smoke formed from the evaporation of the liquid was used to visualise the LV in the flow. The VGs used in this visualisation test were PRWP and PCRWP with an in-line arrangement, as shown in Fig. 6.



(a)



(b)



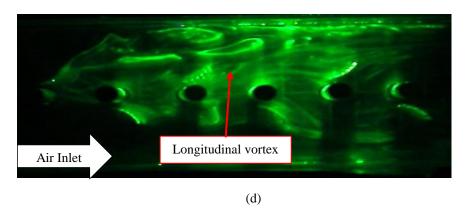


Fig. 6. Visualisation of LV generated by (a) in-line PRWP, (b) staggered PRWP, (c) in-line PCRWP and (d) staggered PCRWP

In Fig. 6 (c) and (d), the PCRWP VGs appear to produce longitudinal vortices (LV) in a wide flow area compared with the PRWP VGs in Fig 6 (a) and (b) downstream. The back region of the PCRWP VGs had a wider frontal surface area than the PRWP VGs. Consequently, mixing the near-fluid the channel walls with the fluid in the mainstream is better, meaning that the heat transfer rate is increased [26]. Downstream, the LV compression in the wake area increases the fluid flow velocity passing through the cylindrical structure, thereby increasing the heat transfer rate from the channel surface to the fluid flow in the wake region [27]. The increase in heat transfer produced when using PCRWP VGs was better than that with PRWP VGs.

3.2 Perforated vortex generators effect on heat transfer

The increase in the convection heat transfer was due to the mixing of fluids caused by the strong longitudinal vortices (LVs)[28]. The strength of the LVs is caused by the amount of VGs sets; increasing the amount of VGs pairs in the test specimen can increase the coefficient of the convection heat transfer [29], as shown in Fig. 7.

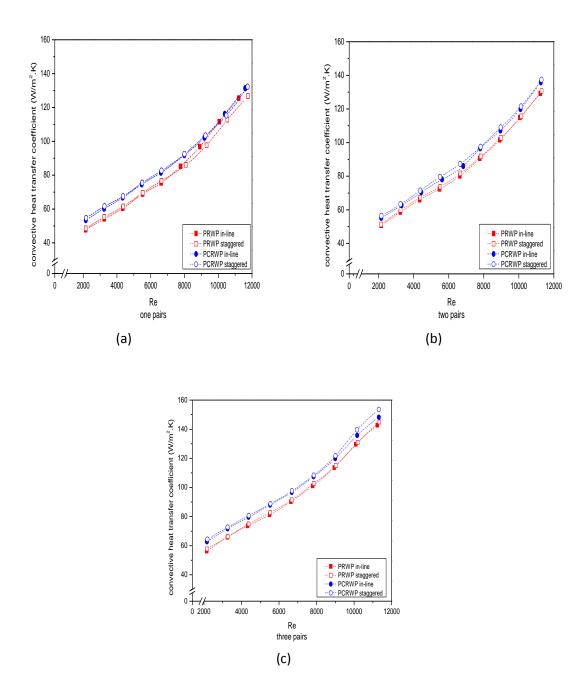


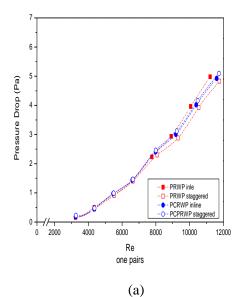
Fig. 7 Graphs of convective heat-transfer coefficient against Reynolds number: (a) one, (b) two and (c) three pairs.

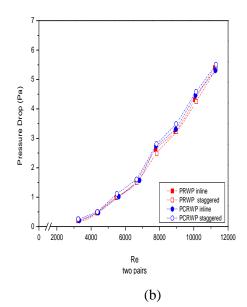
In Fig 7, we can see the convective heat transfer coefficient with respect to the Reynolds number (*Re*), analysed after installing the PCRWP and PRWP with pairs ranging from one, two and three, arranged in-line or staggered. Based on Fig. 7, the convective heat transfer coefficient increased with a rise in *Re* due to an increase in flow vortices and high turbulence intensity in the channel[30], alongside a reduction in the wage region and stagnation area for each increase in flow velocity[31]. The improve in heat transfer for the staggered was better than that for the PCRW VGs with any number of pairs at the

highest *Re* (11,000). The results in Fig. 7 show that the PCRWP VGs worked better than the PRWP VGs, and the staggered arrangement of the former, with three pairs, gave the highest yield (153.5 W/m²·K), as shown in Fig. 7(c). Two PCRW pairs (137.33 W/m²·K, Fig. 7(b)) were better than one (132.25 W/m²·K) (Fig. 7(a)) because the VGs with a concave surface destabilise the force of centrifugal of the fluid flow, strengthening the flow vortices and making the mixing of the hot fluid near the wall with the cold fluid of the main flow more robust[32]. In Fig. 7(a), the convection heat-transfer coefficient for the case of the in-line PRW VGs has the same value as that of the in-line or staggered PCRW VGs in a pair of VGs. In one pair of VGs, a longitudinal vortex is generated after the flow hits and weakens the VGs[29]. This result contrasts with the cases with two and three pairs of VGs, where the longitudinal vortex produced after striking the first VGs is amplified again when the flow strikes the second VGs and so on. Therefore, the value of the heat transfer coefficient in the case of a pair of PRW VGs is the same value as that of PCRW VGs at Reynolds numbers above 8,000.

3.3 Effect of perforated vortex generators on pressure drop

Using VGs can affect the increase in heat transfer, but there is often an accompanying increase in pressure drop, as shown in Fig. 8, where an increase in pressure drop can be seen along with the increases in *Re* and pair numbers for both the VG types PCRW and PRW. In general, the highest pressure drop was observed using the PCRWP VGs with a staggered configuration for all *Re*, except for one pair of VGs. The highest pressure drop was found in the PRWP VGs with an in-line configuration at *Re* greater than 8,000. The pressure drop on the staggered VGs was found to be higher than that on the in-line configuration because of the shorter distance between the VGs of the staggered configuration than that of the in-line[29], caused by the resistance of fluid flow against the walls of the VGs and the expansion of the frontal zone of the VGs in the next-pair arrangement [33]. The pressure drop in the staggered arrangement was lower than that in the in-line arrangement, whereas the PRW VGs type created a lower pressure drop than the PCRW VGs because the latter reduced the frontal area hit by the airflow, resulting in a decrease in drag [34]. In addition, the jet flow from the VG hole can reduce the stagnation flow, which can reduce the pressure drop [35]. A significant decrease in the pressure drop was due to the VG perforation[36]. The best pressure drop value for one pair with a staggered arrangement was 4.58 Pa (see Figure 8(a)), whereas two pairs (5 Pa, Figure 8(b)) were better than three (5.4 Pa, Figure 8(c)).





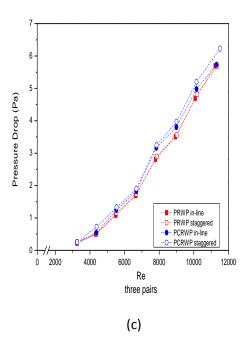
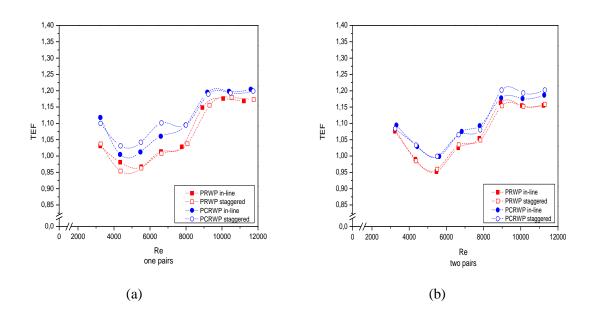


Fig. 8 Graph of pressure drop against Reynolds number: (a) one, (b) two and (c) three pairs

3.4 Effect of perforated VGs on thermal enhancement factor

TEF exhibited the hydraulic thermal performance while using VGs, which played a role in restructuring the incoming fluid flow pattern. The increase in the TEF was due to the influence of complex overlapping structures, which meant that the flow developed into a turbulent structure, significantly affecting the heat transfer increase [37]. The experimental TEF values are shown in Fig. 9.



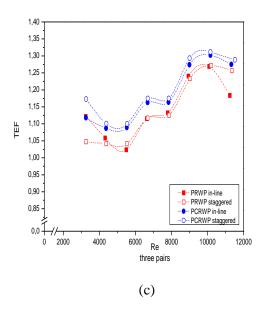


Fig. 9 Graph of thermal enhancement factor against Reynolds number: (a) one, (b) two and (c) three pairs

The TEF is the thermal-hydraulic performance which is the ratio of the increase in heat transfer to the pressure drop ratio. In general, the highest TEF was observed when the PCRWP VGs were used with a staggered configuration, as depicted in Fig 9. The PCRW creates wider flow vortices that can reduce the wake area behind the cylinder. Reducing the wake area can reduce the recirculation zone, affecting the heat transfer from the back of the cylinder to the stream [26]. A large-radius, high-intensity anterior-posterior vortex can reduce the wake area. A lessening within the wake zone increased the flow velocity behind the tube and reduced the recirculation area, resulting in increased heat transfer in this area [27, 38]. As shown in Fig 9, there was an increase in the TEF with greater pairs of VGs used for both the PCRW and PRW VG because the PCRW produced wider flow vortices, which reduced the wake region behind the cylinder, thereby reducing the recirculation zone and impacting the heat transfer increment from the rear cylinder surface to the stream[39]. In this process, a large number of longitudinal vortices with high intensities can reduce the wake area, which increases the flow velocity downstream of the tube and reduces the recirculation region, leading to an increased heat-ransfer rate in the region [40, 41]. Based on Figure 9, the best TEF increase occurred at Re between 8,000–9,000. The best TEF values, with one, two and three pairs occurred in the staggered arrangement with PCRW VGs, at 1.18, 1.20 and 1.29, respectively (see Fig. 9).

3.5 Effects of perforated VGs on the cost-benefit ratio

Economic evaluation cannot be conducted based only on the *TEF* and the net profit from the heat load of the transferred unit[26]. Instead, it must be determined by evaluating the economic value of the heat-transfer improvement by calculating the cost benefit ratio (*CBR*), as shown in Fig. 10.

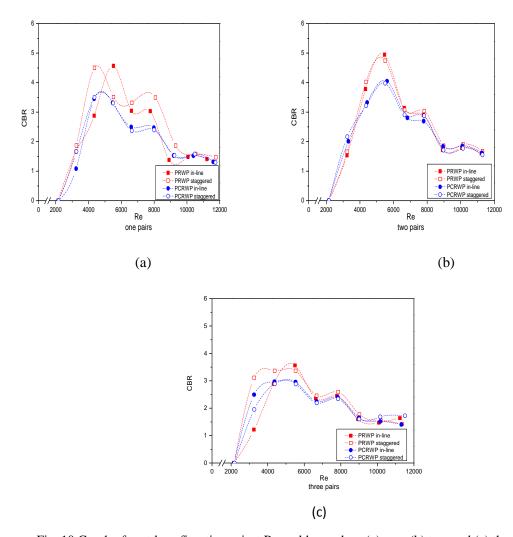


Fig. 10 Graph of cost-benefit ratio against Reynolds number: (a) one, (b) two and (c) three pairs

Fig. 10 show the result of the *CBR* calculation to compare the percentage increase in the pressure drop with the percentage increase in the Nusselt number when using VGs. These results indicate that a lower *CBR* improves thermal performance, which is greater than the drag force [25]. The greatest increase in *CBR* occured with the PRW VGs, with an in-line arrangement, totalling 4.57, 4.95 and 3.56 for one, two and three pairs, respectively. The lowest *CBR* was measured when three sets of PCRW vortex generators with a staggered arrangement were used. The lowest *CBR* were obtained with the three pairs of staggered-type VGs PCRW. The three VG pairs showed a lower *CBR* than the one and two pairs because they resulted in the greatest increase in the Nusselt number, accompanied by a lower pressure drop increase, which lowered the *CBR*. These results show that a lower *CBR* improves thermal performance relative to resistivity [26]. A low value *CBR* inicates a more economical value using VGs. In general, using PCRWP VGs with a staggered configuration is the best.

3.6 Heat Loss Analysis

Heat loss analysis was performed by considering the convection heat transfer from the six tubes to the surrounding fluid flow. The heat transfer rate was calculated for laminar and turbulent flows.

The heat loss in this experiment was calculated by calculating the difference between the induced electric power and total heat through convection from the surface of the tubes to the fluid. In this experiment, six tubes in a wind tunnel were heated using a heater at a power of 40 W; the velocity of the inlet fluid is varied from 0.4 to 2 m/s at intervals of 0.2 m/s or in the Reynolds number range from 2,143 to 11,763. Based on the Reynolds number range, two types of flows were determined; laminar and turbulent. Therefore, the heat loss was determined from the correlation between laminar at 0.4 m/s and turbulent for other velocities. The experimental data for the hydraulic diameter D_h , tube surface area A_{tube} , channel surface area A_c and air specific heat c_p are 0.09223 m, 0.02338908 m², 0.01056 m² and 1.007 J/kgK, respectively. Table 1 is a baseline for calculating heat loss

Table 1 Heat Loss Baseline

| | v (m/s) | Re | Mass flow rate (kg/s) | Density (kg/m³) | Dynamics viscous (kg/ms) | k | Pr | T inlet (C) | T outlet (C) | T tube (C) | Δ T LMTD | ΔT (T tube - T inlet) | Nu | h (W/mK) | q conv (W) | q input (W) | q loss (W) |
|----------|------------|-------|-----------------------------|--------------------|--------------------------------|------|------|-------------|--------------|------------|--------------------|-------------------------------|-----|-------------|------------------|-------------|------------------|
| | 0.4 | 2165 | 0.004757 | 1.13 | 1.9.E-05 | 0.03 | 0.73 | 29 | 33 | 50 | 19 | 21 | 155 | 45 | 19.48 | 40 | 20.52 |
| | 0.6 | 3291 | 0.00719 | 1.13 | 1.9.E-05 | 0.03 | 0.73 | 28 | 31 | 46 | 16 | 18 | 174 | 50 | 18.98 | 40 | 21.02 |
| baseline | 0.8 | 4413 | 0.009618 | 1.14 | 1.9.E-05 | 0.03 | 0.03 | 28 | 30 | 44 | 15 | 16 | 192 | 50 | 19.19 | 40 | 20.81 |
| ouseine | 1 | 5545 | 0.012056 | 1.14 | 1.9.E-05 | 0.03 | 0.73 | 28 | 30 | 43 | 14 | 15 | 214 | 55 | 19.84 | 40 | 20.16 |
| | 1.2 | 6661 | 0.014477 | 1.14 | 1.9.E-05 | 0.03 | 0.73 | 28 | 29 | 43 | 14 | 15 | 228 | 61 | 21.15 | 40 | 18.85 |
| | 1.4 | 7826 | 0.016958 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 28 | 29 | 41 | 12 | 13 | 247 | 70 | 20.30 | 40 | 19.70 |
| | 1.6 | 8965 | 0.019407 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 27 | 29 | 40 | 12 | 13 | 263 | 75 | 21.03 | 40 | 18.97 |
| | 1.8 | 10110 | 0.021863 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 27 | 28 | 39 | 12 | 12 | 296 | 84 | 22.54 | 40 | 17.46 |
| | 2 | 11272 | 0.024341 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 27 | 28 | 38 | 11 | 11 | 342 | 97 | 24.17 | 40 | 15.83 |

From Table 1, the greater the velocity with an increase in the *Re*, the lower the heat loss. It can be observed that the heat flow from the heater not only spreads into the tube, but convection also occurs outside the tube. The heat output increased with *Re*, i.e, the higher the flow velocity, the greater the turbulence through the cylinder and the higher the turbulence intensity. An increase in the turbulence intensity between a cold airflow and hot cylinder with a constant surface temperature is caused by the airflow velocity[26]. In row-tube arrays, this recirculation area increased for the second and subsequent columns. A lower air velocity in the circulation region indicated less airflow in the region participating in the local heating process[37]. The heat loss under all conditions in this experiment is listed in table 2.

Table 2 Calculation of heat loss for the whole case

| type VGs | q conv (W) | q input (W) | q loss (W) |
|----------|------------|-------------|------------|
| Baseline | 20.74 | 40 | 19.26 |
| PCRWPI1 | 25.15 | 40 | 14.85 |
| PCRWPI2 | 27.55 | 40 | 12.45 |

| PCRWPI3 | 27.61 | 40 | 12.39 |
|---------|-------|----|-------|
| PCRWPS1 | 26.43 | 40 | 13.57 |
| PCRWPS2 | 26.43 | 40 | 13.57 |
| PCRWPS3 | 27.94 | 40 | 12.06 |
| PRWPI1 | 24.09 | 40 | 15.91 |
| PRWPI2 | 27.25 | 40 | 12.75 |
| PRWPI3 | 28.82 | 40 | 11.18 |
| PRWPS1 | 23.94 | 40 | 16.06 |
| PRWPS2 | 26.37 | 40 | 13.63 |
| PRWPS3 | 28.12 | 40 | 11.88 |

Table 2 shows that the lowest heat loss occurs when three sets of PCRWPs are staggered. The placement of the VGs can increase heat transfer in square ducts as the VGs create longitudinal vortices, which in turn increase vortex strength in the wake region downstream of the tube. Longitudinal vortices make the overall temperature field more uniform, improve heat mixing and boundary layer modification, and improve heat transfer performance. A higher number of vortex generators creates more longitudinal vortices and significantly increases heat transfer [29, 26].

3.7 Uncertainty Analysis

In this section, uncertainty analysis calculation data will be shown for the temperature at base-line conditions with a velocity of 0.4 m/s as shown in Table 3.

Table 3 Base-line test temperature data at a speed of 0.4 m/s

| $T(Tube_1)$ | $T(Tube_2)$ | $T(Tube_3)$ | $T(Tube_4)$ | $T(Tube_5)$ | $T(Tube_6)$ |
|-------------|-------------|-------------|-------------|-------------|-------------|
| 49.19093 | 51.21368 | 48.32313 | 49.76915 | 47.80219 | 51.27142 |
| 49.1834 | 51.17728 | 48.3156 | 49.79053 | 47.7657 | 51.2639 |
| 49.14545 | 51.16826 | 48.30655 | 49.7526 | 47.7856 | 51.25489 |
| 49.12105 | 51.17277 | 48.28214 | 49.72821 | 47.76118 | 51.2594 |
| 49.15297 | 51.20465 | 48.28515 | 49.73122 | 47.73524 | 51.2624 |
| 49.09966 | 51.15141 | 48.28967 | 49.73573 | 47.76871 | 51.26691 |
| 49.09815 | 51.14991 | 48.23029 | 49.73423 | 47.73826 | 51.29428 |
| 49.08912 | 51.14089 | 48.25019 | 49.66739 | 47.72922 | 51.22751 |

From these data, it is found that \overline{T}_{Tube} can be calculated by the equation as

$$\overline{T}_{Tube} = \frac{\overline{T}_{Tube1} + \overline{T}_{Tube2} + \overline{T}_{Tube3} + \overline{T}_{Tube4} + \overline{T}_{Tube5} + \overline{T}_{Tube6}}{6} = 49.56^{\circ} C$$
(11)

Then, the average standard deviation is obtained by the following formula.

$$s_{tube} = \sqrt{\frac{\sum_{i=1}^{N} (T_{tubei} - \overline{T}_{tube})^2}{N(N-1)}} = 0.029$$
 (12)

Therefore, the average T_{tube} can be written as 49.5 ± 0.029 °C. \bar{T}_{out} calculation results obtained 32.95°C. The average standard deviation was calculated using the following equation:

$$s_{Tout} = \sqrt{\frac{\sum_{i=1}^{N} (T_{outi} - \overline{T}_{out})^2}{N(N-1)}} = 0,051$$
 (13)

Furthermore, the average value of T_{out} can be written as 32.95 \pm 0.051°C. Using the same equation, the standard deviation of T_{in} was found to be 0.033. Thus, the average T_{in} value was 29.75 \pm 0.016°C.

The value of q at a speed of 0.4 m/s was found to be 19.48 W. To determine of the standard deviation of q, the following equation was used:

$$RSS_{q} = \sqrt{\left(s(\Delta T_{out})\frac{\partial q}{\partial T_{out}}\right)^{2} + \left(s(\Delta T_{in})\frac{\partial q}{\partial T_{in}}\right)^{2}}$$

$$\frac{\partial q}{\partial T_{out}} = \frac{\partial (m \cdot c_{p} \cdot T_{out} - m \cdot c_{p} \cdot T_{in})}{\partial T_{out}} = m \cdot c \cdot p$$

$$\frac{\partial q}{\partial T_{in}} = \frac{\partial (m \cdot c_{p} \cdot T_{out} - m \cdot c_{p} \cdot T_{in})}{\partial T_{in}} = -(m \cdot c \cdot p)$$
(14)

where $s(\Delta T_{out}) = 0.051$ °C and $s(\Delta T_{in}) = 0.033$ °C, ensuring, that $RSS_q = \pm 0.290$ W. Therefore, the heat transfer rate q becomes 19.48 ± 0.290 W. The value of ΔT_{lmtd} at a speed of 0.4 m/s was found to be 18.56°C. To determine the value of the standard deviation of ΔT_{lmtd} we used the following equation:

$$RSS_{\Delta T_{lmtd}} = \sqrt{\left(s(\Delta T_{tube})\frac{\partial(\Delta T_{lmtd})}{\partial T_{tube}}\right)^{2} + \left(s(\Delta T_{out})\frac{\partial(\Delta T_{lmtd})}{\partial T_{out}}\right)^{2} + \left(s(\Delta T_{in})\frac{\partial(\Delta T_{lmtd})}{\partial T_{in}}\right)^{2}}$$

$$\frac{\partial(\Delta T_{lmtd})}{\partial T_{tube}} = \frac{\partial\left(\frac{\left(T_{tube} - T_{out}\right) - \left(T_{tube} - T_{in}\right)}{\ln\frac{T_{tube} - T_{out}}{T_{tube}}}\right)}{\partial T_{tube}}$$

$$\frac{\partial(\Delta T_{lmtd})}{\partial T_{out}} = \frac{\partial\left(\frac{\left(T_{tube} - T_{out}\right) - \left(T_{tube} - T_{in}\right)}{\ln\frac{T_{tube} - T_{out}}{T_{tube} - T_{in}}}\right)}{\partial T_{out}}$$

$$\frac{\partial(\Delta T_{lmtd})}{\partial T_{out}} = \frac{\partial\left(\frac{\left(T_{tube} - T_{out}\right) - \left(T_{tube} - T_{in}\right)}{\ln\frac{T_{tube} - T_{out}}{T_{tube} - T_{in}}}\right)}{\partial T_{out}}$$

$$\frac{\partial (\Delta T_{lmtd})}{\partial T_{in}} = \frac{\partial \left(\frac{(T_{tube} - T_{out}) - (T_{tube} - T_{in})}{\ln \frac{T_{tube} - T_{out}}{T_{tube} - T_{in}}} \right)}{\partial T_{in}}$$

where $s(\Delta T_{tube}) = 0.029$ °C, $s(\Delta T_{out}) = 0.051$ °C and $s(\Delta T_{in}) = 0.033$ °C; we get $RSS_{\Delta T_{lmtd}}$ of ± 0.043 , ensuring, that the obtained ΔT_{lmtd} is 8.56 ± 0.043 .

The value of Nu at a speed of 0.4 m/s was found to be 155.31. The standard deviation of Nu was obtained using following equation

$$RSS_{Nu} = \sqrt{\left(s(q)\frac{\partial Nu}{\partial Q}\right)^{2} + s(\Delta T_{lmtd})\frac{\partial Nu}{\partial \Delta T_{lmtd}}}$$

$$\frac{\partial Nu}{\partial q} = \frac{\partial \left(q \cdot D_{h} \cdot At^{-1} \cdot \Delta T_{lmtd}^{-1} \cdot k^{-1}\right)}{\partial q} = \frac{D_{h}}{(At)(\Delta T_{lmtd})(k)}$$

$$\frac{\partial Nu}{\partial \Delta T_{lmtd}} = \frac{\partial \left(q \cdot D_{h} \cdot At^{-1} \cdot \Delta T_{lmtd}^{-1} \cdot k^{-1}\right)}{\partial \Delta T_{lmtd}} = \frac{q \cdot D_{h}}{(At)(\Delta T_{lmtd})^{2}(k)}$$

$$(16)$$

With the values of $s(q) = 0.290 \ W$ and $s(\Delta T_{lmtd}) = 0.043$, the obtained RSS_{Nu} was $\pm 2.889 \ \text{W/(m}^{2\circ}\text{C)}$. Therefore, the value of RSS_{Nu} is $155.31 \pm 2.889 \ \text{W/m}^{2\circ}\text{C}$.

The value of h at a speed of 0.4 m/s was found to be 44.86. To determine the standard deviation of *Nu* the following equation is used

$$RSS_{h} = \sqrt{\left(s(Nu)\frac{\partial h}{Nu}\right)^{2}}$$

$$\frac{\partial h}{\partial Nu} = \frac{\partial (h.D_{h}.k^{-1})}{\partial h} = \frac{k}{D_{h}}$$
(17)

Furthermore, the value of D_h is 0.092 m and k at $T_f = 40.24$ is 0.026. So the value of h at a speed of 0.4 m/s is:

$$RSS_h = \sqrt{\left(s(Nu)\frac{\partial h}{Nu}\right)^2} = 0.83$$

Thus, the number h at a speed of 0.4 m/s is 44.86 \pm 0.83. So, the error h for the baseline at a speed of 0.4 m/s is

$$Error = \frac{RSS_h}{h} \times 100$$

$$Error = \frac{0.83}{44.86} \times 100 = 1.51\%$$
(18)

From the test in the baseline case with a speed of 2.0 m/s, the results of the pressure drop are listed in Table 4, which show that the average P can be calculated as follows:

$$\overline{\Delta P} = \frac{\Delta P_1 + \Delta P_2 + \Delta P_3 + \dots + \Delta P_{30}}{30} = 3.51 \text{ Pa}$$

$$\tag{19}$$

The average standard deviation of the pressure drop can then be calculated using the equation

$$s = \sqrt{\frac{\sum_{i=1}^{N} (\Delta P_i - \overline{\Delta P})^2}{N(N-1)}} = 8.9 \times 10^{-5}$$
 (20)

Baseline case for the pressure drop value at a speed of 2.0 m/s is $3.51 \pm 8.9 \times 10^{-5}$ Pa. Then, the error in the form of percentage can be calculated using the following equation:

$$\frac{8.9 \times 10^{-5}}{3.51} \times 100 = 0.71$$

Table 4 Baseline pressure drop data at a speed of 2.0 m/s

| | Δ <i>P</i> (Pa) | | |
|---------|-----------------|---------|---------|
| Data to | 2.0 m/s | Data to | 2.0 m/s |
| 1 | 0.013 | 16 | 0.012 |
| 2 | 0.013 | 17 | 0.013 |
| 3 | 0.013 | 18 | 0.012 |
| 4 | 0.013 | 19 | 0.012 |
| 5 | 0.012 | 20 | 0.013 |
| 6 | 0.013 | 21 | 0.013 |
| 7 | 0.013 | 22 | 0.012 |
| 8 | 0.012 | 23 | 0.013 |
| 9 | 0.013 | 24 | 0.012 |
| 10 | 0.013 | 25 | 0.013 |
| 11 | 0.013 | 26 | 0.013 |
| 12 | 0.013 | 27 | 0.013 |
| 13 | 0.012 | 28 | 0.013 |
| 14 | 0.012 | 29 | 0.012 |
| 15 | 0.013 | 30 | 0.012 |

The equal calculation approach changed into used for all data. Therefore, the overall error outputs for the pressure-drop vortex generator with placement variations (in-line and staggered), *Re* and amount of VG sets (one, two and three) are listed in Table 5.

Table 5. Overall Pressure Drop (ΔP)

| Overall Error P | |
|-----------------|--|
| (perforated) | |
| 2.94% | |
| 2.87% | |
| 1.98% | |
| 2.88% | |
| 2.34% | |
| 1.36% | |
| 2.72% | |
| 1.80% | |
| 1.80% | |
| 2.43% | |
| 1.91% | |
| | |

The average *TEF* results from the experimental results can be calculated as follows.

$$\overline{TEF} = \frac{TEF_1 + TEF_2 + TEF_3 + \dots + TEF_{12}}{12} = 1.12$$
 (21)

Then, the average standard deviation of the TEF can be calculated with the equation

$$s = \sqrt{\frac{\sum_{i=1}^{N} (TEF_i - \overline{TEF})^2}{N(N-1)}} = 1.07$$
 (22)

Therefore, the TEF value was 1.12 ± 1.07 . Then, the error in the form of percentage can be calculated using the following equation:

$$\frac{1.07}{1.12} \times 100 = 0.94\%$$

The overall error results for the *TEF* vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are are listed in Table 6.

Table 6. Overall error TEF

| Variasi Vortex Generator | Overall Error TEF | |
|--------------------------|-------------------|--|
| | (Berlubang) | |
| 1 RWP in-line | 0.47 % | |
| 2 RWP in-line | 0.47% | |
| 3 RWP in-line | 0.43% | |
| 1 RWP staggered | 0.47% | |
| 2 RWP staggered | 0.47% | |
| 3 RWP staggered | 0.43% | |
| 1 CRWP in-line | 0.45% | |
| 2 CRWP in-line | 0.45% | |
| 3 CRWP in-line | 0.42% | |
| 1 CRWP staggered | 0.45% | |
| 2 CRWP staggered | 0.45% | |
| 3 CRWP staggered | 0.41% | |

First, find the average *CBR* of the experimental results with the following formula.
$$\overline{CBR} = \frac{CBR_1 + CBR_2 + CBR_3 + \dots + CBR_{12}}{12} = 2.14$$
(23)

The average standard deviation of the pressure drop CBR can then be calculated using the following equation:

$$s = \sqrt{\frac{\sum_{i=1}^{N} (CBR_i - \overline{CBR})^2}{N(N-1)}} = 1.60$$
 (24)

The CBR value is 2.14 ± 1.60 . Then the error in the form of percentage can be calculated using thefollowing equation:

$$\frac{1.60}{2.14} \times 100 = 0.63\%$$

The overall error results for the *CBR* vortex generator with placement variations (in-line and staggered), *Re* and amount of VG sets (one, two and three) are listed in Table 7

Table 7. Overall error CBR

| Variasi Vortex Generator | Overall Error CBR |
|--------------------------|-------------------|
| | (Berlubang) |
| 1 RWP in-line | 0.32% |
| 2 RWP in-line | 0.29% |
| 3 RWP in-line | 0.45% |
| 1 RWP staggered | 0.32% |
| 2 RWP staggered | 0.31% |
| 3 RWP staggered | 0.45% |
| 1 CRWP in-line | 0.4% |
| 2 CRWP in-line | 0.42% |
| 3 CRWP in-line | 0.56% |
| 1 CRWP staggered | 0.43% |
| 2 CRWP staggered | 0.42% |
| 3 CRWP staggered | 0.66% |

Conclusion

Based on the experimental results for perforated concave rectangular winglet pair vortex generators (PCRWP VGs) used to increase the heat transfer of airflow through heated tubes arranged in-line in the duct, we conclude that using PCRWP VGs affects the convection heat transfer coefficient, pressure drop in achieving hydraulic thermal performance and cost-benefit ratio. In our investigation, the best heat-transfer convection coefficient was 153.5 W/m²·K for the three pairs of PCRW VGs, in a staggered manner. The greatest improvement in the pressure drop value (4.58 Pa), occurred for one pair of PCRW VGs arranged in a staggered manner, whereas the hydraulic thermal performance was the best (1.29) in this experiment with the three pairs of PCRW VGs arranged in a staggered manner. Finally, the best *CBR* (3.56) was recorded for the three pairs of PCRW VGs composed in a staggered manner.

Acknowledgements

The authors would like to thank LEMLITBANG UHAMKA which has funded this research through internal grants from UHAMKA and UPPI UHAMKA what have contributed in facilitating translation and proof reading. The authors also thank the UNDIP thermofluidics laboratory, where the authors carried out this experiment.

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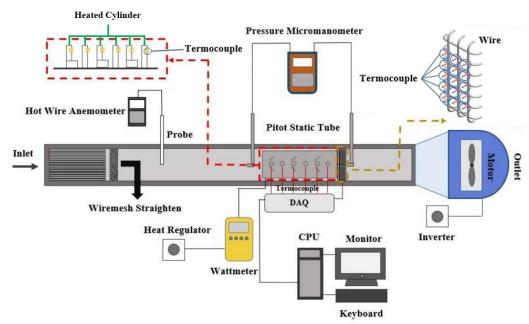
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Dear Editors and Reviewer

Thank you for your letter and for the reviewers' comments concerning our manuscript entitled "Perforated concave rectangular winglet pair vortex generators enhance the heat transfer of air flowing through heated tubes inside a channel" (Manuscript Number: Rineng-D-22-00804R1). The comments are all valuable and very helpful to revise and improve our paper, as well as significant guidelines for our research. We have learned comments carefully and have made the correction that we hope you meet with your approval. We have included the parts requested to be revised in the manuscript. Revised portions are marked in red in the revised paper. The main correction in papers and responses to reviewing comments is flowing.

Reviewer 1: Following are the few observations:

1. Author should add figure of location of thermocouple.



Thank you very much for the proposal. I've revised Figure 1 because I made a mistake in captioning the figure. One thermocouple is placed in the air inlet area, six thermocouples on the back surface of the tubes, and 15 on the outlet side of the wire, as observed in Figure 1.

2. Heaters are placed after test section? How the heating of air takes place?

Thanks for the question. The heater is connected to six tubes with each tube getting the same power. The total heating power of 40 W is induced in the six tubes by a regulator. Heating air flowing through the tubes occurs by convection. So that the air at the outlet side becomes hotter than that from the inlet side.

3. Details of perforations on rectangular and concave winglet is missing. pl add.

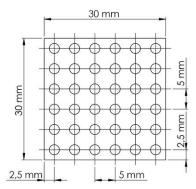
Thank you for the correction. I have added in the paper the detailed geometry of the perforated rectangular winglet (PRW) and perforated concave rectangular winglet (PCRW) vortex generators (VGs), as shown in Figure 2.







Perforated CRWP

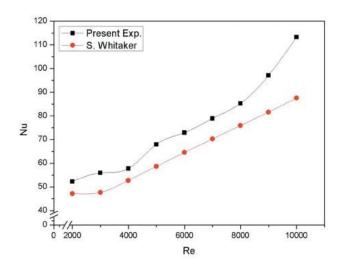


VGs have dimensions of the same length and width of 30 mm and have 36 holes. The bore diameter on the VGs is 2.5 mm. The distance between the holes is 5 mm from the center of the holes.

- 4. Provide details of validation of set up and heat loss analysis.
 - a. Set up validation

Thank you for your suggestion. The current study is a follow-up investigation of the work of Yafid et al [1]. The experimental set-up of this study is similar to that of Yafid et al. experiment. The difference between the current study and the experiment of Yafid et al. is a test object where the current study uses concave rectangular winglet (CRW) VGs, while the work of Yafid et al. uses concave delta winglet (CDW) VGs. Whitaker et al. [2] studied the heat transfer characteristics of airflow through a single cylinder in a rectangular duct. To confirm the results of the experiment Yafid et al. are

valid, the same experimental set-up is determined. The Nu value from the experiment of Yafid et al. compared with Nu values from the experiments of Whitaker et al. in the Reynolds number range of 2,143 to 11,763, as shown in the figure below.



From the figure, it can be observed that Nu from the experimental results of Yafid et al. have the same trend as the experiments of Whitaker et al.

b. Heat Loss analysis

Heat loss analysis is carried out by taking into account the convection heat transfer from the six tubes to the surrounding fluid flow. Calculation of the heat transfer rate is carried out for two types of flow, namely laminar flow and turbulent flow.

The calculation of heat loss in this experiment is determined by calculating the difference between the induced electric power and the total heat through convection from the surface of the tubes to the fluid. In this experiment, six tubes in a wind tunnel are heated by a heater with a power of 40 W. In this work, the velocity of the inlet fluid is varied from 0.4 to 2 m/s at intervals of 0.2 m/s or in the Reynolds number range from 2,143 to 11,763. Based on the Reynolds number range, two types of flow are determined, namely laminar and turbulent. Therefore, the heat loss was determined from the correlation between laminar at 0.4 m/s and turbulent for other velocities.

The described formulas Nu, h, and q were used to determine the heat loss in the conduit of the six tubes.

$$Nu = \frac{q D_h}{A_{tube} \Delta T_{LMTD} k} \tag{2}$$

$$h = \frac{q}{A_{tube} \, \Delta T_{LMTD}} \tag{3}$$

$$q = \dot{m} c_n \left(T_{out} - T_{in} \right) \tag{4}$$

Where D_h , A_{tube} , ΔT_{LMTD} , \dot{m} , c_p , T_{out} , T_{in} , are hydraulic diameter, tube surface area, log mean temperature difference, mass flow rate, specific heat, outlet temperature, and inlet temperature.

$$D_h = \frac{4A_c}{P} \tag{5}$$

$$\Delta T_{LMTD} = \frac{(\overline{T}_{tube} - \overline{T}_{out}) - (\overline{T}_{tube} - \overline{T}_{in})}{\ln[(\overline{T}_{tube} - \overline{T}_{out}) - (\overline{T}_{tube} - \overline{T}_{in})]}$$
(6)

Where A_c dan T_{tube} are channel surface area and tube temperature, respectively. The experimental data for hydraulic diameter D_h ,, tube surface area A_{tube} , channel surface area A_c , and air specific heat c_p are 0.09223 m, 0.02338908 m², 0.01056 m², and 1.007 J/kgK, respectively. The following is a table for calculating the heat loss baseline.

Table 1 Heat Loss Baseline

| | v (m/s) | Re | Mass flow rate (kg/s) | Density (kg/m³) | Dynamics viscous (kg/ms) | k | Pr | T inlet (C) | T outlet (C) | T tube (C) | Δ T LMTD | ΔT (T tube - T inlet) | Nu | h (W/mK) | q conv (W) | q input (W) | q loss (W) |
|----------|------------|-------|-----------------------------|--------------------|--------------------------------|------|------|-------------------|--------------------|------------------|--------------------|-----------------------------------|-----|-------------|------------------|-------------------|---------------|
| | 0.4 | 2165 | 0.004757 | 1.13 | 1.9.E-05 | 0.03 | 0.73 | 29 | 33 | 50 | 19 | 21 | 155 | 45 | 19.48 | 40 | 20.52 |
| | 0.6 | 3291 | 0.00719 | 1.13 | 1.9.E-05 | 0.03 | 0.73 | 28 | 31 | 46 | 16 | 18 | 174 | 50 | 18.98 | 40 | 21.02 |
| baseline | 0.8 | 4413 | 0.009618 | 1.14 | 1.9.E-05 | 0.03 | 0.03 | 28 | 30 | 44 | 15 | 16 | 192 | 50 | 19.19 | 40 | 20.81 |
| | 1 | 5545 | 0.012056 | 1.14 | 1.9.E-05 | 0.03 | 0.73 | 28 | 30 | 43 | 14 | 15 | 214 | 55 | 19.84 | 40 | 20.16 |
| | 1.2 | 6661 | 0.014477 | 1.14 | 1.9.E-05 | 0.03 | 0.73 | 28 | 29 | 43 | 14 | 15 | 228 | 61 | 21.15 | 40 | 18.85 |
| | 1.4 | 7826 | 0.016958 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 28 | 29 | 41 | 12 | 13 | 247 | 70 | 20.30 | 40 | 19.70 |
| | 1.6 | 8965 | 0.019407 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 27 | 29 | 40 | 12 | 13 | 263 | 75 | 21.03 | 40 | 18.97 |
| | 1.8 | 10110 | 0.021863 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 27 | 28 | 39 | 12 | 12 | 296 | 84 | 22.54 | 40 | 17.46 |
| | 2 | 11272 | 0.024341 | 1.15 | 1.9.E-05 | 0.03 | 0.73 | 27 | 28 | 38 | 11 | 11 | 342 | 97 | 24.17 | 40 | 15.83 |

In the table 1, it can be seen that the greater the velocity with the increase in Re number, the lower the heat loss. It can be seen that the heat flow from the heater does not only spread into the tube, but convection occurs to the outside of the tube. Heat output increases with increasing Re. That is, the higher the flow velocity, the greater the turbulence through the silinder and the higher the turbulence intensity. An increase in turbulence intensity between a cold airflow and a hot cylinder with constant surface temperature is caused by the airflow velocity [3]. In row-tube arrays, this recirculation area increases for the second and subsequent columns. A lower air velocity in the circulation region indicates less airflow in that region participating in

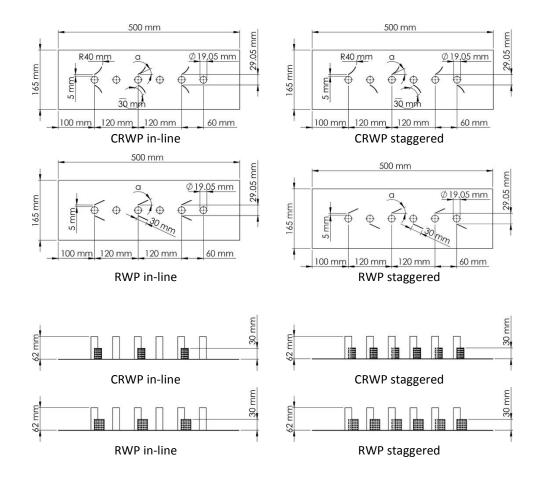
the local heating process [4]. The heat loss of all conditions in this experiment is shown in table 2 below.

Table 2 Calculation of heat loss for the whole case

| type VGs | q conv (W) | q input (W) | q loss (W) |
|----------|------------|-------------|------------|
| Baseline | 20.74 | 40 | 19.26 |
| PCRWPI1 | 25.15 | 40 | 14.85 |
| PCRWPI2 | 27.55 | 40 | 12.45 |
| PCRWPI3 | 27.61 | 40 | 12.39 |
| PCRWPS1 | 26.43 | 40 | 13.57 |
| PCRWPS2 | 26.43 | 40 | 13.57 |
| PCRWPS3 | 27.94 | 40 | 12.06 |
| PRWPI1 | 24.09 | 40 | 15.91 |
| PRWPI2 | 27.25 | 40 | 12.75 |
| PRWPI3 | 28.82 | 40 | 11.18 |
| PRWPS1 | 23.94 | 40 | 16.06 |
| PRWPS2 | 26.37 | 40 | 13.63 |
| PRWPS3 | 28.12 | 40 | 11.88 |

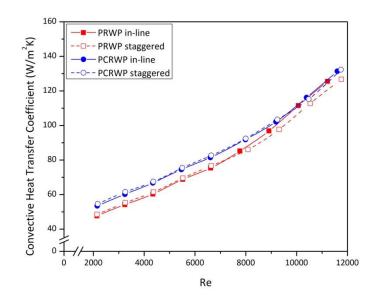
Table 2 shows that the lowest heat loss occurs when three sets of PCRWPs are staggered. The placement of the VGs can increase heat transfer in square ducts as the VGs create longitudinal vortices that increase vortex strength in the wake region downstream of the tube. Longitudinal vortices make the overall temperature field more uniform, improve heat mixing and boundary layer modification, and improve heat transfer performance. A higher number of vortex generators creates more longitudinal vortices and greatly increases heat transfer [5], [3].

5. Mention pitch kept between two pins/winglet.
Thank you for the question. I've added Figure 3 (in the paper) showing the pitch between VGs for both inline and staggered configurations.



6. At lower Re both inline and staggered arrangement gives the same result, while deviation is observed after Re 8000. The author has to justify.

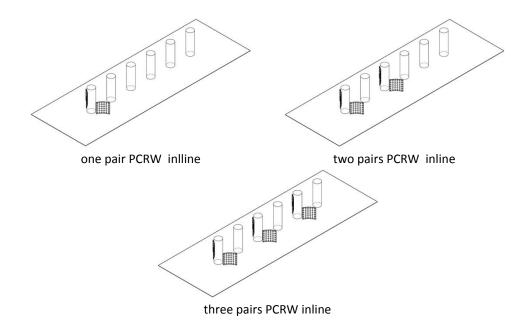
Thank you for your suggestion. In Figure 7(a) (in the paper), the convection heat transfer coefficient for the case of PRW VGs in-line has the same value as that of PCRW VGs in-line or staggered in a pair of VGs. In one pair of VGs, the longitudinal vortex is generated after the flow hits the VGs and weakens (He et al., 2013). This is in contrast to two and three pairs of VGs where the longitudinal vortex produced after striking the first VGs has amplified again when the flow strikes the second VGs and so on. Therefore, the value of the heat transfer coefficient in the case of a pair of PRW VGs has the same value as that of PCRW VGs at Reynolds numbers above 8,000.



7. Provide more clarification about 1,2 & 3 pairs.

Thank you for your suggestion. This study describes the cases of PCRW and PRW VGs for one, two, and three pairs. I have added explanations for cases one, two, and three pairs of perforated VGs to the paper.

For PRW and PCRW VGs in-line configurations with one, two, and three pairs are shown in Figures 4. For one pair, the VGs are placed on the left and right sides of the first row of tubes. VGs are placed on the first and third row tubes for two pairs. As for the three pairs, VGs are placed on the first, third, and fifth row tubes.



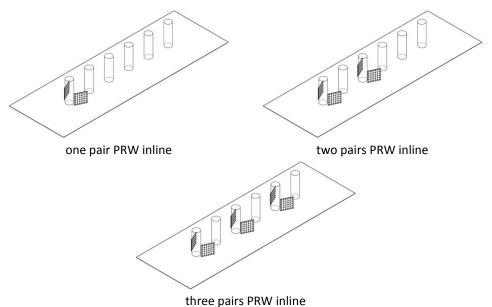
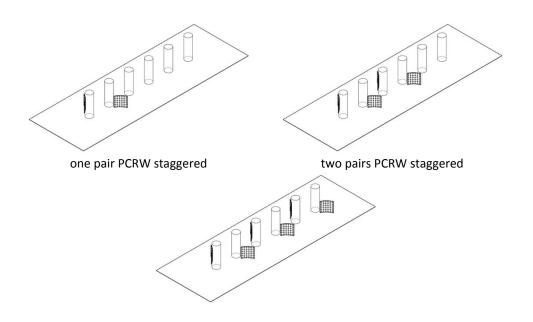


Figure 4. VGs pairs in-line configurations

For PRW and PCRW VGs staggered configurations with one, two, and three pairs are shown in Figures 5. For one pair, the VGs are placed on the right side of the first row tube and the left side of the second row tube. VGs are placed on the right side of the first, third row tubes and on the left side of the second and fourth tubes for two pairs. As for the three pairs, the VGs are placed on the right side of the first, third, fifth row of tubes and on the left side of the second, fourth and sixth tubes.



three pairs PCRW staggered

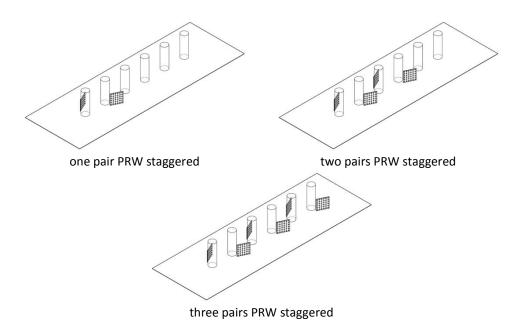


Figure 5. VGs pairs staggered configurations

- 8. Compare the PRWP and PCRWP with without perforation.

 Thank you for your request. In this study, an experiment was conducted to compare PRW VGs and PCRW VGs in improving heat transfer in a rectangular channel, as shown in Figures 7 to 9.
 - a. Comparison of convection heat transfer coefficient values.

Figure 7 provides a comparison of the convection heat transfer coefficients for PRW and PCRW. It can be seen that there was an increase in the convection heat transfer coefficient with a rise in *Re* due to an increase in flow vortices and high turbulence intensity in the channel [6], alongside a reduction in the wage region and stagnation area for each increase in flow velocity [7]. The increase in heat transfer for staggered VGs was better than in-line for PCRW VGs with any number of pairs at the highest *Re* (11,000). The results in Figure 7 show that the PCRWP VGs worked better than the PRWP VGs, and the staggered arrangement of the former with three pairs gave the highest yield, of 153.5 W/m².K, as shown in Figure 7(c). Meanwhile, two PCRW pairs (137.33 W/m².K, Figure 7(b)) were better than one (132.25 W/m².K, Figure 7(a)) because VGs with a concave surface destabilise the centrifugal force of the fluid flow, which strengthens the flow vortices. This makes the mixing of the hot fluid near the wall with the cold fluid of the main flow more robust [8].

b. Pressure drop comparison

In general, the highest pressure drop was observed using PCRWP VGs with staggered configuration for all Reynolds numbers except for one pair of VGs, as shown in Figure 8. The highest pressure drop was found in PRWP VGs with in-line configuration at Reynolds numbers greater than 8,000. The pressure drop on the staggered VGs was found to be higher than that of the in-line due to the shorter distance between the VGs of the staggered configuration than that of the in-line [5].

c. TEF comparison

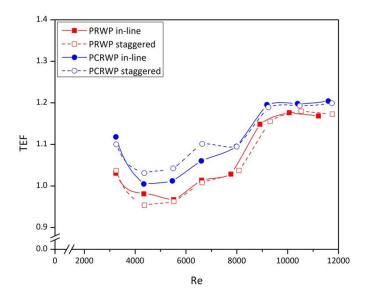
TEF is the thermal-hydraulic performance which is the ratio of the increase in heat transfer to the pressure drop ratio. In general, the highest TEF was observed in the use of PCRWP VGs with staggered configuration, as depicted in Figure 9. PCRW creates wider flow vortices that can reduce the wake area behind the cylinder. Reducing the wake area can reduce the recirculation zone. This affects the increased heat transfer from the back of the cylinder to the stream [3]. A large radius, high intensity antrior-posterior vortex can reduce the wake area. A reduction in the wake area increases the flow velocity behind the tube and reduces the recirculation area, resulting in an increased heat transfer in this area [9], [10].

d. CBR comparison

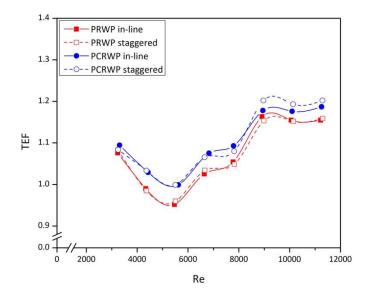
A low value of CBR means a more economical value from the use of VGs. In general, CBR on the use of PCRWP VGs with staggered configuration is the best, as informed in Figure 10. The lowest CBR value results were obtained with three pairs of staggered type VGs PCRW. Three pairs of VGs lower CBR than one and two VG pairs. This is because the installation of three pairs VGs results in a higher Nusselt number increase than one and two pairs VGs, resulting in a lower pressure drop increase and therefore a lower CBR. These results show that lower CBR improves thermal performance relative to resistivity [11].

9. Discuss how number of pairs contribute in improving TEF

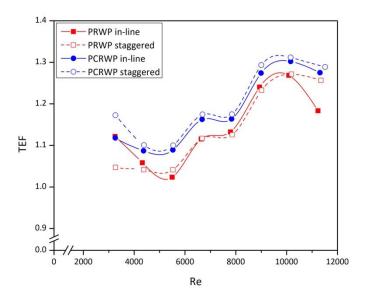
Thank you for your question. Figure 9 shows the effect of the number of pairs and configuration of VGs on TEF, this is also found in Ref. (He et al., 2013), (Sun et al., 2020), and (Ranjan et al., 2022).



a. One pair



b. Two pairs



c. Three pairs

From the experimental results, as shown in Figure 9, the TEF with three pairs of VGs for both inline and staggered was the highest. The TEF for the case of 3 pairs of PCRWs was 5.02% greater than that of one and two pairs of PRWP VGs. The main reason is because the concave surface of the PCR causes the flow to be thrown away due to the centrifugal force which results in a stronger flow vorticity [12]. Larger and stronger flow vortices can reduce the recirculation zone which has an impact on increasing heat transfer from the rear surface of the tube to the flow [10]. The presence of flow that is formed in each gap between the VGA and the tube causes the TEF in the staggered configuration to be greater than that of the in-line [13]. The increase in TEF for the three-pair case with the staggered configuration was 1.50% and 4.91% greater than that of the inline PCRWP and PRWP, respectively.

Reference

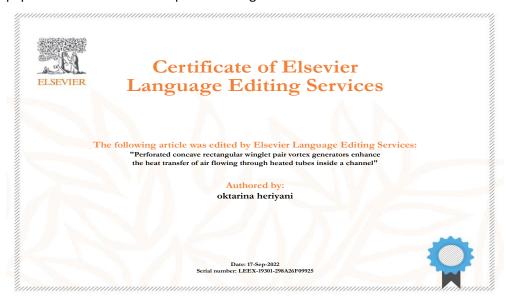
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Reviewer 2: This paper presented experimental results of the heat transfer, pressure drop and thermal performance characteristics of the perforated concave rectangular winglet pair vortex generators on plates in rectangular ducts to increase the heat transfer through the six heated tubes to the air stream. Perforated concave rectangular winglets were compared with perforated rectangular winglet pairs vortex generator mounted on rectangular plates. The subject of the article falls within the scope of the Results in Engineering. In my view, unless the paper is rewritten in a proper way, I think it is inadequate to be published in a scientific journal in its present state. I would like to provide the following comment:

1. There are many spelling and grammar mistakes in this paper, and many sentences are not easy to follow. The grammar and structure of sentences in this paper need to be modified carefully, such as the title of the article. The language of this paper need to be improved by a native English speaker.

Thank you for your suggestion. To improve grammar and structure, I have sent this paper to elsevier service for proof reading.



- 2. The velocity ranges studied are given in the Abstract, whereas the Reynolds number ranges studied should be given.
 - Thank you for your correction. In this experiment, the intake air velocity is in the range of 0.4 to 2 m/s at intervals of 0.2 m/s or in the Reynolds number range of 2.143 to 11.763. I have included the Reynolds number range in the abstract.
- 3. Please review the keywords and add a few, for instance heat transfer, thermal performance.

Thank you for your suggestion. The keywords for this study are vortex generators, heat transfer, thermal-hydraulic performance, economic benefit. I've added keywords in the paper.

4. The method section can be expanded.

Thank you for your suggestion. I have improved this research method in the paper.

Based on Fig. 1, the rectangular channel is equipped with a blower (50 Hz, Wipro with a rated voltage of 220V), an inverter (Mitsubishi Electric type FR-D700 with an accuracy of 0.01), straightener, hot wire anemometer (Lutron type AM-4204 with an accuracy of 0.1), wattmeter (Lutron DW-6060 with an accuracy \pm 1.0), central processing unit (CPU), micromanometer, thermocouple (K type with a temperature interval of -200 -1250°C and an accuracy \pm 0.5) where one thermocouple was placed in the air inlet area, six thermocouples on the back surface of the tubes and 15 on the outlet side of the wire, data acquisition (Advantech USB-4718 type with an accuracy of 0.001) and heater regulator. The heater was connected to six tubes with a diameter of 19.05 mm and height of 65.8 mm, with each tube having the same power. Total heating power of 40 W was applied to the six tubes using a regulator. The heating air flowing through the tubes occurs via convection. Thus, the air at the outlet side becomes hotter than that at the inlet side.

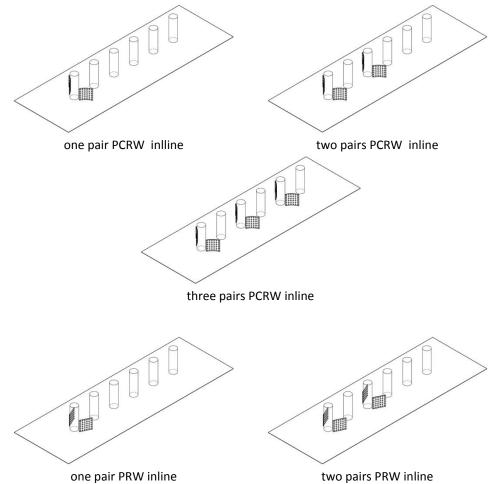
A pressure micromanometer (Fluke type 922, with an accuracy of \pm 0.05) was used to monitor the flow pressure drop. Two pitot tubes, each set 26 cm ahead of the inlet of the test specimen and 2.5 cm behind it, were connected to a micromanometer to measure the pressure drop. The pressure drop measurements were recorded 30 times for 5 sekon at each speed variation. Furthermore, flow visualisation was performed by directing the smoke from vaporised fluid in the fluid vaporator into the mainflow.

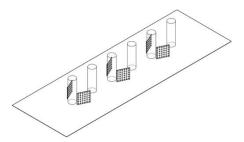
5. The arrangement of vortex generators is given in Figure 2, but Figure 2 is not sufficient for a clear understanding of the construction of vortex generators. Additional figures should be drawn which clearly show the construction of vortex generators and the in-line and staggered arrangement. Thank you for your suggestion. The following shows the construction of vortex generators inline and staggered (figure 2, 4 and 5 in the paper).



Figure 2. Geometry of the VGs

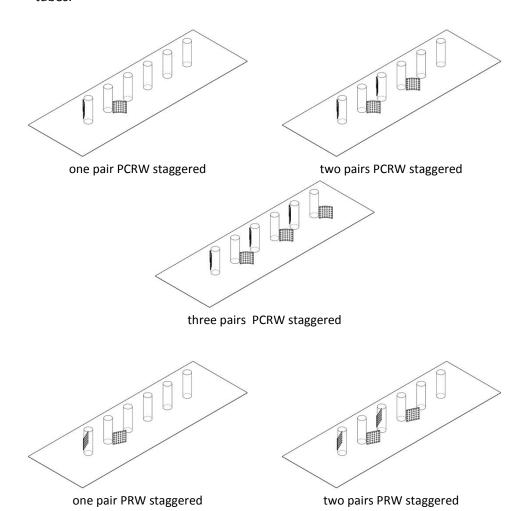
For PRW and PCRW VGs in-line configurations with one, two, and three pairs are shown in Figures 4. For one pair, the VGs are placed on the left and right sides of the first row of tubes. VGs are placed on the first and third row tubes for two pairs. As for the three pairs, VGs are placed on the first, third, and fifth row tubes.

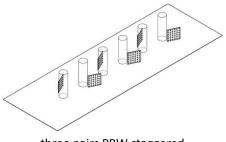




three pairs PRW inline Figure 4. VGs pairs in-line configurations

For PRW and PCRW VGs staggered configurations with one, two, and three pairs are shown in Figures 5. For one pair, the VGs are placed on the right side of the first row tube and the left side of the second row tube. VGs are placed on the right side of the first, third row tubes and on the left side of the second and fourth tubes for two pairs. As for the three pairs, the VGs are placed on the right side of the first, third, fifth row of tubes and on the left side of the second, fourth and sixth tubes.





three pairs PRW staggered

Figure 5. VGs pairs staggered configurations

6. Thermal characteristics are given in terms of the convective heat transfer coefficient (h), friction characteristics are given in terms of pressure drop. Why are thermal characteristics not given in terms of Nusselt number (Nu) and friction characteristics in terms of coefficient of friction (f)?

Thank you for your correction. h versus Re and P vs Re are used instead of Nu - Re and f - Re in this experiment because, in the TEF calculation, the value of Nu represents the value of h resulting from the equation in formula 3 (in the paper), thus

$$h = \frac{N_u k}{D_h} \tag{3}$$

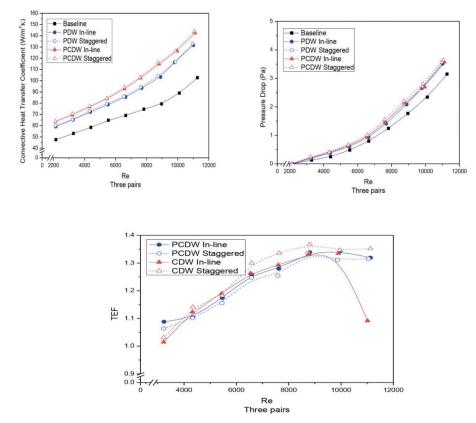
Figure 7 (in the paper) that h rises as Re rises shows that the Nu increases as Re rises, where h rises as Re rises [1]. While f in this experiment represents the ΔP as shown in the following formula

$$f = \frac{2 \Delta P D_h}{\rho V^2 (L+6D)} \tag{8}$$

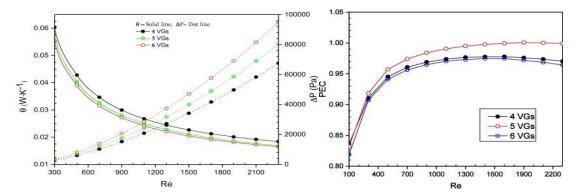
Formula 8 states that the friction factor (f) in the flow rate is determined by using the pressure drop (ΔP) characteristic where increasing the Reynolds number in figure 8 will decrease the friction factor [2].

The following are examples of several studies that use h as a representative of Nu and (ΔP) as a representative of f.

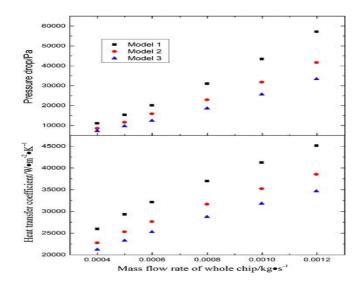
a. The experimental results of Yafid et al indicate that perforated VGs can increase the heat transfer rate and decrease the pressure drop using the parameters h, ΔP , and TEF as shown in the following graph[3].



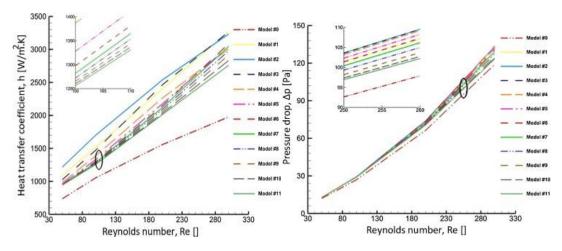
b. Al Asadi et al represents heat transfer coefficient and pressure drop to show the results of their investigation that the addition of span wise gap variations can increase heat transfer performance and reduce pressure drop [4], as shown in the figure below.



c. Increased heat transfer in heat sinks, Zhang et al described the micro gap by pairing more VGs which resulted in a larger heat transfer coefficient and a reduced pressure drop value [5], as shown in the figure below.



d. Hosseinirad et al showed that the increase in heat transfer coefficient and pressure drop vs. Reynold number had a tendency to increase with the increase in Re to evaluate heat transfer [6]. There is an increase in heat transfer with an increase in Re which is indicated by an increase in the heat transfer coefficient and an increase in pressure drop along with an increase in Re.



7. It is not specified how the hydraulic diameter (D_h) is calculated. How the Reynolds number (Re) was calculated is not specified.

Thank you for your corrections. The calculation of the hydraulic diameter in this experiment uses a rectangular air duct with a side length of a = 0.165 m and a side width of b = 0.064 m with the resulting D_h of 0.0106 from the following formula.

$$D_h = \frac{4A_c}{p} = \frac{4ab}{2(a+b)} = \frac{2ab}{a+b} \tag{5}$$

The result of D_h is used to calculate Re with the formula

$$Re = \frac{\rho u_{in} D_h}{\mu} \tag{7}$$

With u_{in} of 0.4 – 2 m/s with an interval of 0.2 m/s, on the physical properties of air at a pressure of 1 atm and is the viscosity of the fluid so that Re used in this experiment ranges from 2,143 – 11,763.

8. No correlation (Nu-Re), (f-Re) is given.

Thank you for the correction. In this experiment, Nu - Re and f - Re are not shown but use h vs Re and ΔP vs Re because the Nu value represents the value of h that arises in this experiment as in equation 4 (in the manuscript)

$$h = \frac{N_u k}{D_h} \tag{3}$$

While f in this experiment represents the ΔP as shown in the following formula

$$f = \frac{2 \Delta P D_h}{\rho V^2 (L+6D)} \tag{8}$$

The formula states that the friction factor (f) in the flow rate is determined by using the pressure drop (ΔP) characteristic where increasing the Reynolds number in figure 8 will decrease the friction factor [2].

9. Error analysis is given, but uncertainty analysis is not done.

Thank you for the correction. In the following, uncertainty analysis calculation data will be shown for the temperature at base line conditions with a velocity of 0.4 m/s as shown in Table 3 below.

Table 3 Base-line test temperature data at a speed of 0.4 m/s

| T (Tube₁) | T (Tube₂) | T (Tube₃) | T (Tube4) | T (Tube₅) | T (Tube ₆) |
|-----------|-----------|-----------|-----------|-----------|------------------------|
| 49.19093 | 51.21368 | 48.32313 | 49.76915 | 47.80219 | 51.27142 |
| 49.1834 | 51.17728 | 48.3156 | 49.79053 | 47.7657 | 51.2639 |
| 49.14545 | 51.16826 | 48.30655 | 49.7526 | 47.7856 | 51.25489 |
| 49.12105 | 51.17277 | 48.28214 | 49.72821 | 47.76118 | 51.2594 |
| 49.15297 | 51.20465 | 48.28515 | 49.73122 | 47.73524 | 51.2624 |
| 49.09966 | 51.15141 | 48.28967 | 49.73573 | 47.76871 | 51.26691 |
| 49.09815 | 51.14991 | 48.23029 | 49.73423 | 47.73826 | 51.29428 |
| 49.08912 | 51.14089 | 48.25019 | 49.66739 | 47.72922 | 51.22751 |

From these data, it is found that \overline{T}_{Tube} can be calculated by the equation as

$$\overline{T}_{Tube} = \frac{\overline{T}_{Tube1} + \overline{T}_{Tube2} + \overline{T}_{Tube3} + \overline{T}_{Tube4} + \overline{T}_{Tube5} + \overline{T}_{Tube6}}{6} = 49.56^{\circ} C$$
(11)

Then, the average standard deviation is obtained by the following formula.

$$s_{tube} = \sqrt{\frac{\sum_{i=1}^{N} (T_{tubei} - \overline{T}_{tube})^2}{N(N-1)}} = 0.029$$
 (12)

Therefore, the average T_{tube} can be written as 49.5 \pm 0.029°C. \overline{T}_{out} calculation results obtained 32.95°C. The average standard deviation was calculated using the following equation:

$$s_{Tout} = \sqrt{\frac{\sum_{i=1}^{N} (T_{outi} - \overline{T}_{out})^2}{N(N-1)}} = 0,051$$
 (13)

Furthermore, the average value of T_{out} can be written as 32.95 \pm 0.051°C. Using the same equation, the standard deviation of T_{in} was found to be 0.033. Thus, the average T_{in} value was 29.75 \pm 0.016°C.

The value of q at a speed of 0.4 m/s was found to be 19.48 W. To determine of the standard deviation of q, the following equation was used:

$$RSS_{q} = \sqrt{\left(s(\Delta T_{out})\frac{\partial q}{\partial T_{out}}\right)^{2} + \left(s(\Delta T_{in})\frac{\partial q}{\partial T_{in}}\right)^{2}}$$

$$\frac{\partial q}{\partial T_{out}} = \frac{\partial (m \cdot c_{p} \cdot T_{out} - m \cdot c_{p} \cdot T_{in})}{\partial T_{out}} = m \cdot c \cdot p$$

$$\frac{\partial q}{\partial T_{in}} = \frac{\partial (m \cdot c_{p} \cdot T_{out} - m \cdot c_{p} \cdot T_{in})}{\partial T_{in}} = -(m \cdot c \cdot p)$$
(14)

where $s(\Delta T_{out})=0.051^{\circ}\text{C}$ and $s(\Delta T_{in})=0.033^{\circ}\text{C}$, ensuring, that RSS_q = \pm 0.290 W. Therefore, the heat transfer rate q becomes 19.48 \pm 0.290 W. The value of ΔT_{lmtd} at a speed of 0.4 m/s was found to be 18.56°C. To determine the value of the standard deviation of ΔT_{lmtd} we used the following equation:

$$RSS_{\Delta T_{lmtd}} = \sqrt{\left(s(\Delta T_{tube})\frac{\partial(\Delta T_{lmtd})}{\partial T_{tube}}\right)^{2} + \left(s(\Delta T_{out})\frac{\partial(\Delta T_{lmtd})}{\partial T_{out}}\right)^{2} + \left(s(\Delta T_{in})\frac{\partial(\Delta T_{lmtd})}{\partial T_{in}}\right)^{2}}$$
(15)

$$\frac{\partial \left(\Delta T_{lmtd}\right)}{\partial T_{tube}} = \frac{\partial \left(\frac{\left(T_{tube} - T_{out}\right) - \left(T_{tube} - T_{in}\right)}{\ln \frac{T_{tube} - T_{out}}{T_{tube} - T_{in}}}\right)}{\partial T_{tube}}$$

$$\frac{\partial (\Delta T_{lmtd})}{\partial T_{out}} = \frac{\partial \left(\frac{(T_{tube} - T_{out}) - (T_{tube} - T_{in})}{\ln \frac{T_{tube} - T_{out}}{T_{tube} - T_{in}}} \right)}{\partial T_{out}}$$

$$\frac{\partial \left(\Delta T_{lmtd}\right)}{\partial T_{in}} = \frac{\partial \left(\frac{\left(T_{tube} - T_{out}\right) - \left(T_{tube} - T_{in}\right)}{\ln \frac{T_{tube} - T_{out}}{T_{tube} - T_{in}}}\right)}{\partial T_{in}}$$

where $s(\Delta T_{tube})=0.029$ °C, $s(\Delta T_{out})=0.051$ °C and $s(\Delta T_{in})=0.033$ °C; we get $RSS_{\Delta T_{imtd}}$ of \pm 0.043, ensuring, that the obtained ΔT_{lmtd} is 8.56 \pm 0.043.

The value of Nu at a speed of 0.4 m/s was found to be 155.31. The standard deviation of *Nu* was obtained using following equation

$$RSS_{Nu} = \sqrt{\left(s(q)\frac{\partial Nu}{\partial Q}\right)^{2} + s(\Delta T_{lmtd})\frac{\partial Nu}{\partial \Delta T_{lmtd}}}$$

$$\frac{\partial Nu}{\partial q} = \frac{\partial \left(q \cdot D_{h} \cdot At^{-1} \cdot \Delta T_{lmtd}^{-1} \cdot k^{-1}\right)}{\partial q} = \frac{D_{h}}{(At)(\Delta T_{lmtd})(k)}$$

$$\frac{\partial Nu}{\partial \Delta T_{lmtd}} = \frac{\partial \left(q \cdot D_{h} \cdot At^{-1} \cdot \Delta T_{lmtd}^{-1} \cdot k^{-1}\right)}{\partial \Delta T_{lmtd}} = \frac{q \cdot D_{h}}{(At)(\Delta T_{lmtd})^{2}(k)}$$

$$(16)$$

With the values of s(q)=0.290~W and $s(\Delta T_{lmtd})=0.043$, the obtained RSS_{Nu} was \pm 2.889 W/(m²°C). Therefore, the value of RSS_{Nu} is 155.31 \pm 2.889 W/m²°C.

The value of h at a speed of 0.4 m/s was found to be 44.86. To determine the standard deviation of Nu the following equation is used

$$RSS_h = \sqrt{\left(s(Nu)\frac{\partial h}{Nu}\right)^2}$$

$$\frac{\partial h}{\partial Nu} = \frac{\partial (h.D_h.k^{-1})}{\partial h} = \frac{k}{D_h}$$
(17)

Furthermore, the value of D_h is 0.092 m and k at T_f = 40.24 is 0.026. So the value of h at a speed of 0.4 m/s is:

$$RSS_h = \sqrt{\left(s(Nu)\frac{\partial h}{Nu}\right)^2} = 0.83$$

Thus, the number h at a speed of 0.4 m/s is 44.86 \pm 0.83. So, the error h for the baseline at a speed of 0.4 m/s is

$$Error = \frac{RSS_h}{h} \times 100$$
 (18)
 $Error = \frac{0.83}{44.86} \times 100 = 1.51\%$

From the test in the baseline case with a speed of 2.0 m/s, the results of the pressure drop are listed in Table 4, which show that the average *P* can be calculated as follows:

$$\overline{\Delta P} = \frac{\Delta P_1 + \Delta P_2 + \Delta P_3 + \dots + \Delta P_{30}}{30} = 3.51 \text{ Pa}$$
 (19)

The average standard deviation of the pressure drop can then be calculated using the equation

$$s = \sqrt{\frac{\sum_{i=1}^{N} (\Delta P_i - \overline{\Delta P})^2}{N(N-1)}} = 8.9 \times 10^{-5}$$
 (20)

Baseline case for the pressure drop value at a speed of 2.0 m/s is $3.51 \pm 8.9 \times 10^{-5}$ Pa. Then, the error in the form of percentage can be calculated using the following equation:

$$\frac{8.9 \times 10^{-5}}{3.51} \times 100 = 0.71$$

Table 4 Baseline pressure drop data at a speed of 2.0 m/s

| | | ΔP (Pa) | |
|---------|---------|-----------------|---------|
| Data to | 2.0 m/s | Data to | 2.0 m/s |
| 1 | 0.013 | 16 | 0.012 |
| 2 | 0.013 | 17 | 0.013 |
| 3 | 0.013 | 18 | 0.012 |
| 4 | 0.013 | 19 | 0.012 |
| 5 | 0.012 | 20 | 0.013 |
| 6 | 0.013 | 21 | 0.013 |
| 7 | 0.013 | 22 | 0.012 |
| 8 | 0.012 | 23 | 0.013 |
| 9 | 0.013 | 24 | 0.012 |
| 10 | 0.013 | 25 | 0.013 |
| 11 | 0.013 | 26 | 0.013 |
| 12 | 0.013 | 27 | 0.013 |
| 13 | 0.012 | 28 | 0.013 |
| 14 | 0.012 | 29 | 0.012 |
| 15 | 0.013 | 30 | 0.012 |

The equal calculation approach changed into used for all data. Therefore, the overall error outputs for the pressure-drop vortex generator with placement variations (in-line and staggered), *Re* and amount of VG sets (one, two and three) are listed in Table 5.

Table 5. Overall Pressure Drop (ΔP)

| Overall Error P |
|-----------------|
| (perforated) |
| 2.94% |
| 2.87% |
| 1.98% |
| 2.88% |
| 2.34% |
| 1.36% |
| 2.72% |
| 1.80% |
| 1.80% |
| 2.43% |
| 1.91% |
| 0.97% |
| |

The average TEF results from the experimental results can be calculated as follows.

$$\overline{TEF} = \frac{TEF_1 + TEF_2 + TEF_3 + \dots + TEF_{12}}{12} = 1.12$$
 (21)

Then, the average standard deviation of the TEF can be calculated with the equation

$$s = \sqrt{\frac{\sum_{i=1}^{N} (TEF_i - \overline{TEF})^2}{N(N-1)}} = 1.07$$
 (22)

Therefore, the TEF value was 1.12 \pm 1.07. Then, the error in the form of percentage can be calculated using the following equation:

$$\frac{1.07}{1.12} \times 100 = 0.94\%$$

The overall error results for the TEF vortex generator with placement variations (in-line and staggered), Re and amount of VG sets (one, two and three) are are listed in Table 6.

Table 6. Overall error TEF

| Overall Error TEF |
|-------------------|
| (Berlubang) |
| 0.47 % |
| 0.47% |
| 0.43% |
| 0.47% |
| 0.47% |
| 0.43% |
| 0.45% |
| 0.45% |
| |

| 3 CRWP in-line | 0.42% |
|------------------|-------|
| 1 CRWP staggered | 0.45% |
| 2 CRWP staggered | 0.45% |
| 3 CRWP staggered | 0.41% |

First, find the average CBR of the experimental results with the following formula.

$$\overline{CBR} = \frac{CBR_1 + CBR_2 + CBR_3 + \dots + CBR_{12}}{12} = 2.14$$
 (23)

The average standard deviation of the pressure drop *CBR* can then be calculated using the following equation:

$$s = \sqrt{\frac{\sum_{i=1}^{N} (CBR_i - \overline{CBR})^2}{N(N-1)}} = 1.60$$
 (24)

The CBR value is 2.14 ± 1.60 . Then the error in the form of percentage can be calculated using the following equation:

$$\frac{1.60}{2.14} \times 100 = 0.63\%$$

The overall error results for the *CBR* vortex generator with placement variations (in-line and staggered), *Re* and amount of VG sets (one, two and three) are listed in Table 7

Table 7. Overall error CBR

| Variasi Vortex Generator | Overall Error CBR |
|--------------------------|-------------------|
| | (Berlubang) |
| 1 RWP in-line | 0.32% |
| 2 RWP in-line | 0.29% |
| 3 RWP in-line | 0.45% |
| 1 RWP staggered | 0.32% |
| 2 RWP staggered | 0.31% |
| 3 RWP staggered | 0.45% |
| 1 CRWP in-line | 0.4% |
| 2 CRWP in-line | 0.42% |
| 3 CRWP in-line | 0.56% |
| 1 CRWP staggered | 0.43% |
| 2 CRWP staggered | 0.42% |
| 3 CRWP staggered | 0.66% |

Reference

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Reviewer 3: In connection with climate change and an increase in the average annual temperature on Earth, there is a new danger of the negative impact of high temperatures on human life.

In this case, the improvement of air conditioning systems, including the search for the best thermal enhancement factor, cost-benefit ratio, etc., takes on a new sense, which is one of the main targets of this article. The topic is timely and of great practical significance to environmental protection, enhancing safety, and people's life comfort.

The manuscript is well-structured and includes all necessary parts.

Two key strengths of the paper are a good introduction section and an analysis and discussion of the results. Both research objectives and content are clear. The key scientific issues to be solved are moderate. The research experimental method is reasonable.

There are also several shortages worthy to be mentioned:

- 1. Seriously revise the formulas
 - a. If you use an italic font in formulas, use an italic font in their descriptions. For example, in formula (1), the Nusselt number (Nu) and friction factor (f); in formula (3), heat transfer coefficient (h). It may confuse the reader.

Thank you for the corrections. For formula and description fonts, improvements have been made where all formula and description fonts are italicized consistently.

b. The Nusselt numbers in formula (1) and formula (2) have different designations. It may confuse the reader.

Thanks for the correction. Consistent improvements have been made to writing the *Nu* symbol on paper.

c. What are Nusselt number and friction factor with subscript 0 in formula (1)?

Thanks for the corrections. The subscript 0 for Nusselt number and friction factor is meant for the baseline condition. This additional explanation has been added to the paper. The following is an explanation of formula 1.

$$TEF = \frac{\left(\frac{Nu}{Nu_0}\right)}{\left(\frac{f}{f_0}\right)^{\frac{1}{3}}} \tag{1}$$

Di mana: Nu_0 = Nusselt number pada kondisi baseline f_0 = friction factor pada kondisi baseline

d. In formula (5), a pressure drop is the lowercase letter Δp , but in formula (6), a pressure drop is uppercase ΔP . Are these different pressures?

Thank you for the corrections. I'm so sorry for the error in writing the pressure drop symbol which is inconsistent. Improvements in writing pressure drop have been made with uppercase P for the formula on the paper.

2. What is the error of pressure drop measurement with the Fluke 922 Airflow Micromanometer described in section "3.2 Effect of perforated vortex generators on pressure drop"? Did it cover the necessary range of pressures to be investigated? Could micromanometer error have affected the conclusions of the section? Because the pressure drop values of 4.58 Pa, 5 Pa and 5.4 Pa are close to each other.

Thanks for the question. Pressure measurement errors with the Fluke 922 Airflow Micromanometer are explained in the uncertainty analysis section with the calculation results as below.

From the test in the baseline case with a speed of 2.0 m/s, the results of the pressure drop are listed in Table 4, which show that the average *P* can be calculated as follows:

$$\overline{\Delta P} = \frac{\Delta P_1 + \Delta P_2 + \Delta P_3 + \dots + \Delta P_{30}}{30} = 3.51 \text{ Pa}$$
 (19)

The average standard deviation of the pressure drop can then be calculated using the equation

$$s = \sqrt{\frac{\sum_{i=1}^{N} (\Delta P_i - \overline{\Delta P})^2}{N(N-1)}} = 8.9 \times 10^{-5}$$
 (20)

Baseline case for the pressure drop value at a speed of 2.0 m/s is $3.51 \pm 8.9 \times 10^{-5} Pa$. Then, the error in the form of percentage can be calculated using the following equation:

$$\frac{8.9 \times 10^{-5}}{3.51} \times 100 = 0.71$$

Table 4 Baseline pressure drop data at a speed of 2.0 m/s

| | ΔP (Pa) | | | | |
|---------|---------|---------|---------|--|--|
| Data to | 2.0 m/s | Data to | 2.0 m/s | | |
| 1 | 0.013 | 16 | 0.012 | | |
| 2 | 0.013 | 17 | 0.013 | | |
| 3 | 0.013 | 18 | 0.012 | | |
| 4 | 0.013 | 19 | 0.012 | | |

| 5 | 0.012 | 20 | 0.013 |
|----|-------|----|-------|
| 6 | 0.013 | 21 | 0.013 |
| 7 | 0.013 | 22 | 0.012 |
| 8 | 0.012 | 23 | 0.013 |
| 9 | 0.013 | 24 | 0.012 |
| 10 | 0.013 | 25 | 0.013 |
| 11 | 0.013 | 26 | 0.013 |
| 12 | 0.013 | 27 | 0.013 |
| 13 | 0.012 | 28 | 0.013 |
| 14 | 0.012 | 29 | 0.012 |
| 15 | 0.013 | 30 | 0.012 |

The equal calculation approach changed into used for all data. Therefore, the overall error outputs for the pressure-drop vortex generator with placement variations (in-line and staggered), *Re* and amount of VG sets (one, two and three) are listed in Table 5.

Table 5. Overall Pressure Drop (ΔP)

| Vortex Generator | Overall Error P |
|-------------------|-----------------|
| Variations | (perforated) |
| 1 PRWP in-line | 2.94% |
| 2 PRWP in-line | 2.87% |
| 3 PRWP in-line | 1.98% |
| 1 PRWP staggered | 2.88% |
| 2 PRWP staggered | 2.34% |
| 3 PRWP staggered | 1.36% |
| 1 PCRWP in-line | 2.72% |
| 2 PCRWP in-line | 1.80% |
| 3 PCRWP in-line | 1.80% |
| 1 PCRWP staggered | 2.43% |
| 2 PCRWP staggered | 1.91% |
| 3 PCRWP staggered | 0.97% |

The results of the measurement error calculation are still below the maximum accuracy limit of the tool by 5% so that it does not affect the conclusion section.

3. The measurement error in the section "3.3 Effect of perforated VGs on thermal enhancement factor" and "Effects of perforated VGs on the cost-benefit ratio" is not clear. Can you show the error bar or describe it in the description?

Thanks for the question. The measurement error for the thermal increase factor and the cost benefit ratio has been added to the explanation in the paper in the data uncertainty section, as follows.

a. error bar TEF

The average TEF results from the experimental results can be calculated as follows.

$$\overline{TEF} = \frac{TEF_1 + TEF_2 + TEF_3 + \dots + TEF_{12}}{12} = 1.12$$
 (21)

Then, the average standard deviation of the TEF can be calculated with the equation

$$s = \sqrt{\frac{\sum_{i=1}^{N} (TEF_i - \overline{TEF})^2}{N(N-1)}} = 1.07$$
 (22)

Therefore, the TEF value was 1.12 \pm 1.07. Then, the error in the form of percentage can be calculated using the following equation:

$$\frac{1.07}{1.12} \times 100 = 0.94\%$$

The overall error results for the *TEF* vortex generator with placement variations (inline and staggered), *Re* and amount of VG sets (one, two and three) are are listed in Table 6.

Table 6. Overall error TEF

| Variasi Vortex Generator | Overall Error TEF |
|--------------------------|-------------------|
| | (Berlubang) |
| 1 RWP in-line | 0.47 % |
| 2 RWP in-line | 0.47% |
| 3 RWP in-line | 0.43% |
| 1 RWP staggered | 0.47% |
| 2 RWP staggered | 0.47% |
| 3 RWP staggered | 0.43% |
| 1 CRWP in-line | 0.45% |
| 2 CRWP in-line | 0.45% |
| 3 CRWP in-line | 0.42% |
| 1 CRWP staggered | 0.45% |
| 2 CRWP staggered | 0.45% |
| 3 CRWP staggered | 0.41% |

b error bar CBR

First, find the average CBR of the experimental results with the following formula.

$$\overline{CBR} = \frac{CBR_1 + CBR_2 + CBR_3 + \dots + CBR_{12}}{12} = 2.14 \tag{23}$$

The average standard deviation of the pressure drop *CBR* can then be calculated using the following equation:

$$s = \sqrt{\frac{\sum_{i=1}^{N} (CBR_i - \overline{CBR})^2}{N(N-1)}} = 1.60$$
 (24)

The *CBR* value is 2.14 ± 1.60 . Then the error in the form of percentage can be calculated using the following equation:

$$\frac{1.60}{2.14} \times 100 = 0.63\%$$

The overall error results for the *CBR* vortex generator with placement variations (in-line and staggered), *Re* and amount of VG sets (one, two and three) are listed in Table 7.

Table 7. Overall error CBR

| Variasi Vortex Generator | Overall Error CBR |
|--------------------------|-------------------|
| | (Berlubang) |
| 1 RWP in-line | 0.32% |
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| 3 RWP in-line | 0.45% |
| 1 RWP staggered | 0.32% |
| 2 RWP staggered | 0.31% |
| 3 RWP staggered | 0.45% |
| 1 CRWP in-line | 0.4% |
| 2 CRWP in-line | 0.42% |
| 3 CRWP in-line | 0.56% |
| 1 CRWP staggered | 0.43% |
| 2 CRWP staggered | 0.42% |
| 3 CRWP staggered | 0.66% |

4. In my humble opinion, the section "3.5 Flow visualisation" is better presented first in section "3. Results and Discussion".

Thanks for the suggestions. The discussion of section 3.5 on visualization has been moved to the earlier section to 3.1 in the paper.

We tried our best to improve the manuscript and made some changes in the revised paper, and here we did not list the specific changes but marked in red in revised paper. We appreciate for Editors and Reviewrs' warm work earnestly, and hope that the correction will meet with approval. Once again, thank you very much for your comments and suggestions.

Yours Sincerely

Oktarina Heriyani